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**THESIS**

DETERMINING THE EFFECT OF ENDWALL  
BOUNDARY LAYER SUCTION IN A LARGE  
SCALE SUBSONIC COMPRESSOR CASCADE

by

Matthew Allen Webber

March, 1993

Thesis Advisor:

Garth V. Hobson

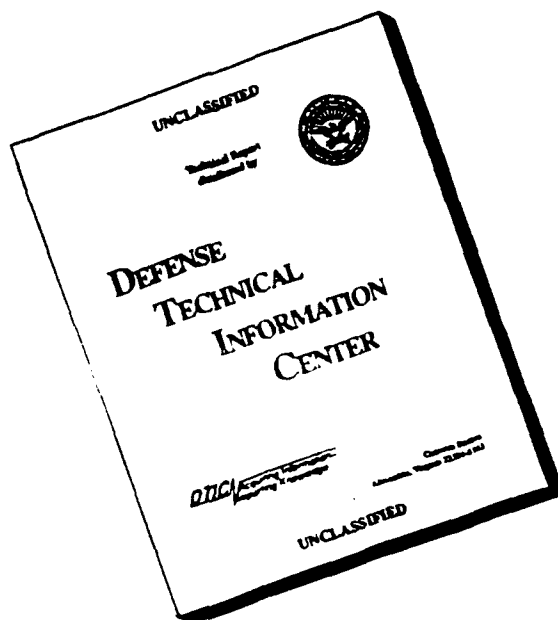
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### REPORT DOCUMENTATION PAGE

1a Report Security Classification: Unclassified			1b Restrictive Markings		
2a Security Classification Authority			3 Distribution/Availability of Report		
2b Declassification/Downgrading Schedule			Approved for public release; distribution is unlimited.		
4 Performing Organization Report Number(s)			5 Monitoring Organization Report Number(s)		
6a Name of Performing Organization Naval Postgraduate School		6b Office Symbol (if applicable) 31	7a Name of Monitoring Organization Naval Postgraduate School		
6c Address (city, state, and ZIP code) Monterey CA 93943-5000			7b Address (city, state, and ZIP code) Monterey CA 93943-5000		
8a Name of Funding/Sponsoring Organization		8b Office Symbol (if applicable)	9 Procurement Instrument Identification Number		
Address (city, state, and ZIP code)			10 Source of Funding Numbers		
			Program Element No	Project No	Task No
11 Title (include security classification) DETERMINING THE EFFECT OF ENDWALL BOUNDARY LAYER SUCTION IN A LARGE SCALE SUBSONIC COMPRESSOR CASCADE					
12 Personal Author(s) Matthew Allen Webber					
13a Type of Report Master's Thesis		13b Time Covered From To	14 Date of Report (year, month, day) 1993, March, 31		15 Page Count 140
16 Supplementary Notation The views expressed in this thesis are those of the author and do not reflect the official policy or position of the Department of Defense or the U.S. Government.					
17 Cosati Codes			18 Subject Terms (continue on reverse if necessary and identify by block number)		
Field	Group	Subgroup	Boundary Layer, Cascade, Controlled-Diffusion, Axial Velocity Ratio, Displacement Thickness		
19 Abstract (continue on reverse if necessary and identify by block number)					
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20 Distribution/Availability of Abstract _X_ unclassified/unlimited    same as report    DTIC users			21 Abstract Security Classification Unclassified		
22a Name of Responsible Individual Garth V. Hobson			22b Telephone (include Area Code) (408) 656-2888		22c Office Symbol Code 31/Sc

DD FORM 1473,84 MAR

83 APR edition may be used until exhausted

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BOUNDARY LAYER SUCTION IN A LARGE  
SCALE SUBSONIC COMPRESSOR CASCADE

by

Matthew Allen Webber  
Lieutenant, United States Navy  
B.S., University of Kansas

Submitted in partial fulfillment  
of the requirements for the degree of

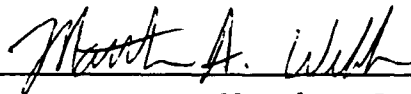
MASTER OF SCIENCE IN AERONAUTICAL ENGINEERING

from the

NAVAL POSTGRADUATE SCHOOL

March 1993

Author:



Matthew A. Webber

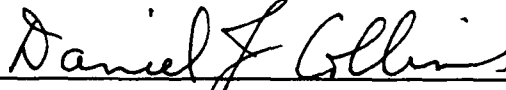
Approved by:



Garth V. Hobson, Thesis Advisor



Raymond P. Shreeve, Second Reader



Daniel J. Collins, Chairman  
Department of Aeronautics and Astronautics

# ABSTRACT

An arrangement of suction slots was installed in the Naval Postgraduate School's subsonic cascade wind tunnel. The aim was to improve flow two-dimensionality to enable flow separation of the installed controlled-diffusion blades at high incidence. The slots were located 17.25 inches upstream of the test section. Pressure and laser Doppler velocimetry measurements were made, for a Reynolds number of 711000 and an inlet flow angle of  $44.4^\circ$ , upstream and downstream of the test section to determine the effects of varying suction. The set of baseline inlet flow field measurements was to be used for comparison purposes in future tunnel modifications. The results showed that the tunnel endwall boundary layers were asymmetric for the baseline configuration. Uniform suction was not achieved in both the pitchwise and spanwise directions. However, the axial velocity ratio was reduced by 1.9% and the blade loading increased slightly with increased suction.

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## I. INTRODUCTION

### A. BACKGROUND

Cascade experimentation continues to supply designers of axial flow turbo-machines with useful data. Frequently the objective of the measurements is to obtain two-dimensional flow around a particular blade shape since blading design is still carried out along quasi-three-dimensional or two-dimensional stream tubes. Therefore, the two-dimensionality of the flow within the cascade is important.

Flow two-dimensionality in compressor cascade experiments has received considerable attention. The degree of two-dimensionality has generally been determined by measuring the axial velocity density ratio (AVDR), or axial velocity ratio (AVR) for incompressible flow, across the blade row. There is experimental evidence showing that the AVR and aspect ratio of a cascade are important parameters in affecting flow deviation, losses and pressure distribution, i.e., stall point, of test blades [Ref. 1].

If a cascade has parallel solid end walls, the displacement effect of the endwall boundary layers causes a convergence of the stream-tube through the cascade. This will have two effects on the midspan boundary layers: first, the convergence alters the blade loading and thus the freestream

pressure field applied to the surface boundary layers; secondly, the spanwise convergence of the blade surface flow causes a thickening of the midspan boundary layer [Ref. 2]. Therefore, for a cascade with an aspect ratio less than three, provisions for end wall boundary layer removal should be incorporated. This can take two forms, first, provisions should be made for a porous wall or slots upstream of the blade row and, or, secondly, a porous wall within the blade row. For either compressor or turbine cascades, careful removal of the upstream endwall boundary layer will improve the two-dimensionality of the flow [Ref.3].

#### **B. CASCADE MODIFICATION EXAMPLES**

The work of some researchers to modify existing cascades is well documented in References 2, 4, 5 and 6. Reference 2 presents measurements for a large scale turbine cascade in which an arrangement of end-plates was incorporated to improve the two-dimensionality of the midspan flow. The endwall boundary layers were removed through slots located in the inlet section. Since the slots were a considerable distance ahead of the test section, the boundary layer grew to approximately 15mm by the time it reached the test section. Therefore, endwall slots were installed within the test section to further remove the boundary layer, resulting in the effective control of the AVR.

The effect of endwall suction on the flow in two-dimensional wind tunnels is examined in [Ref. 4]. The model that was used involved a flat plate airfoil in a tunnel with a suction window shaped to permit an analytic solution. The solution showed that the lift coefficient depended explicitly on the porosity parameter of the suction window and implicitly on the suction pressure differential.

Boundary layer control for a seven blade compressor cascade is described in [Ref.5]. In this study, a suction slot was placed ahead of the test section to allow optical access for laser-Doppler velocimetry measurements (LDV). The system required strong suction and a complex baffling system within the suction slot. Six suction ducts were located at a half blade spacing on either side of the blades. Each individual duct had a separate baffle control and was adjusted by changing the baffle position while examining the outlet flow. Presumably the control of the endwall suction distribution controlled the size of the endwall boundary layer at its intersection with the blade leading edge, thus controlling the origin of the corner disturbances. The AVDR was determined by averaging the local axial velocity ratio over three blade passages centered at the minimum velocity ratio point of the center blade wake. The average AVDR was found to be 1.0 with a day-to-day variation of  $\pm 3\%$ .

The boundary layer control used in Reference 6 on a large scale compressor cascade having 7 blades consisted of three techniques. First, an endwall boundary layer suction scoop was placed ahead of the test section. This scoop was instrumented with static pressure taps to ensure the boundary layer was removed without contracting or diffusing the inlet flow. Secondly, an upper and lower bleed passage was used at the side walls between blade 1 and 7. Thirdly, corner suction was applied on each test blade to eliminate secondary flow caused by the corner disturbance.

#### **C. PRIOR CONTROLLED DIFFUSION (CD) BLADE TESTS**

Previous experiments at the Naval Postgraduate School Turbopropulsion Laboratory's Cascade Tunnel, using 20 CD blades designed by Sanger [Ref. 7], have not produced stall of the blade profile at high inlet flow angles. This is possibly due to the effects of three-dimensional flow in the endwall region of the cascade. Dreon, [Ref.8], performed wake measurements and blade surface pressure measurements at air inlet angles of  $40.3^\circ$  and  $43.4^\circ$ . The AVDR for these tunnel settings was 1.062 and 1.042 respectively. The measured blade surface pressures gave no evidence of flow separation.

Extensive LDV experiments by Elazar, [Ref. 9], were performed at inlet angles of  $40^\circ$ ,  $43.4^\circ$ , and  $46^\circ$  which also did not show any flow separation on the suction surface. Five-hole pneumatic probe measurements were performed by

Classick [Ref.10] for an inlet flow angle of  $48^{\circ}$ . These measurements gave an AVDR of 1.108 and no flow separation occurred at this high incidence. More recently, Hobson and Shreeve [Ref. 11], performed detailed measurements at an inlet flow angle of  $48^{\circ}$ . Their results showed a steady pressure increase on the suction surface with no separation. However, the pressure gradient leveled off at the trailing edge, suggesting, possibly, the approach of flow separation.

#### **D. PURPOSE**

The present aim was to improve the two-dimensionality of the flow in the cascade using inlet boundary layer suction in order to allow an investigation of the onset of stall in the present installed CD blades at higher incidence angles. The effect of endwall suction was determined with probe and LDV measurements, with slots installed 17.25 inches ahead of the test section.

## **II. TEST FACILITY, INSTRUMENTATION AND INITIAL SETUP**

### **A. CASCADE BASE LINE CONFIGURATION**

A schematic diagram of the Rectilinear Subsonic Cascade Wind Tunnel facility is shown in Figure 1. A detailed description of its design and operation is found in an earlier thesis [Ref. 12]. Flow inlet conditions were investigated in detail by McGuire [Ref. 13]. Since that time, the inlet guide vanes were replaced with a new row of 60 blades. The design is given in Appendix D. The tunnel contained 20 CD blades with a chord of 5.01 inches and span of 10 inches. The blade spacing was set at 3 inches for a solidity of 1.67 and the setting angle was  $14.1^\circ$ . Blades 7 and 8, located at 20 to 23 inches pitch, were used for all LDV, AVR and boundary layer displacement thickness measurements. The instrumented blade was located at mid-pitch, 30 inches from the east sidewall.

### **B. WIND TUNNEL MODIFICATION**

The ability to perform boundary layer control on the cascade endwalls was incorporated into the original design. Two manifolds with access to a blower were located on each endwall 17.25 inches ahead of the test section. The manifolds were equipped with blank-off plates for the base line tests. The present modifications to the cascade included the installation of suction slots and connections to a blower.



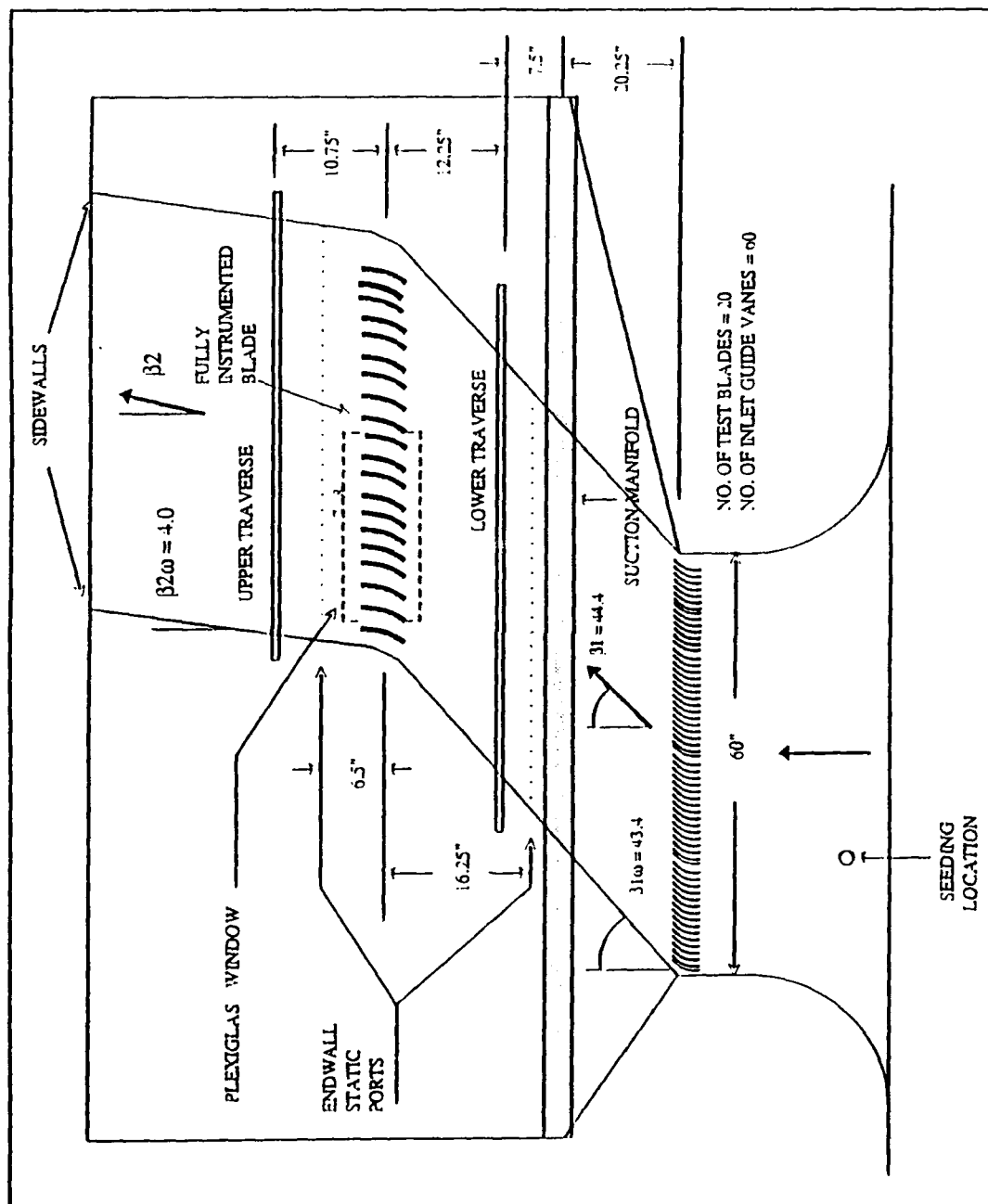
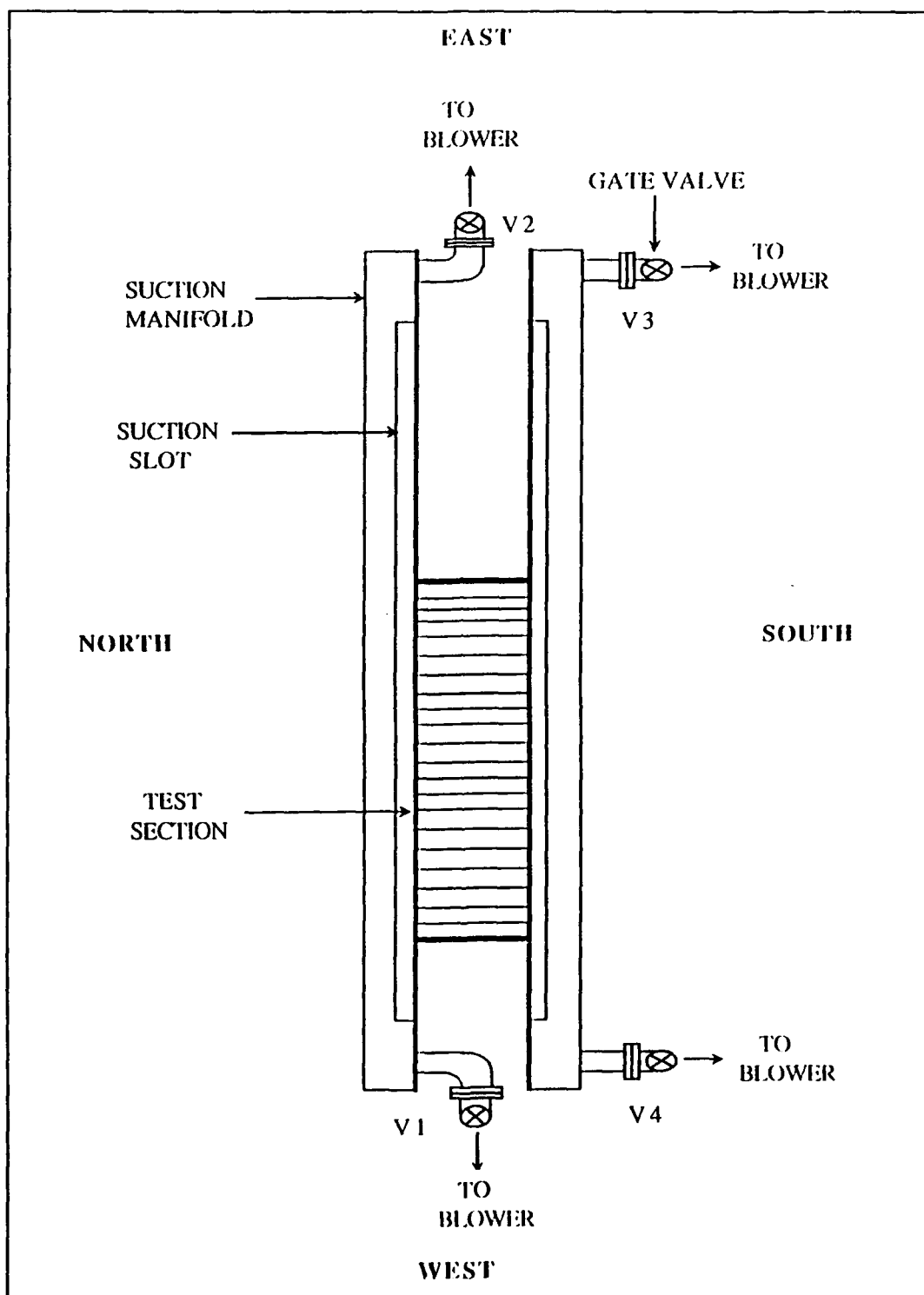


Figure 1. Rectilinear Subsonic Cascade Wind Tunnel

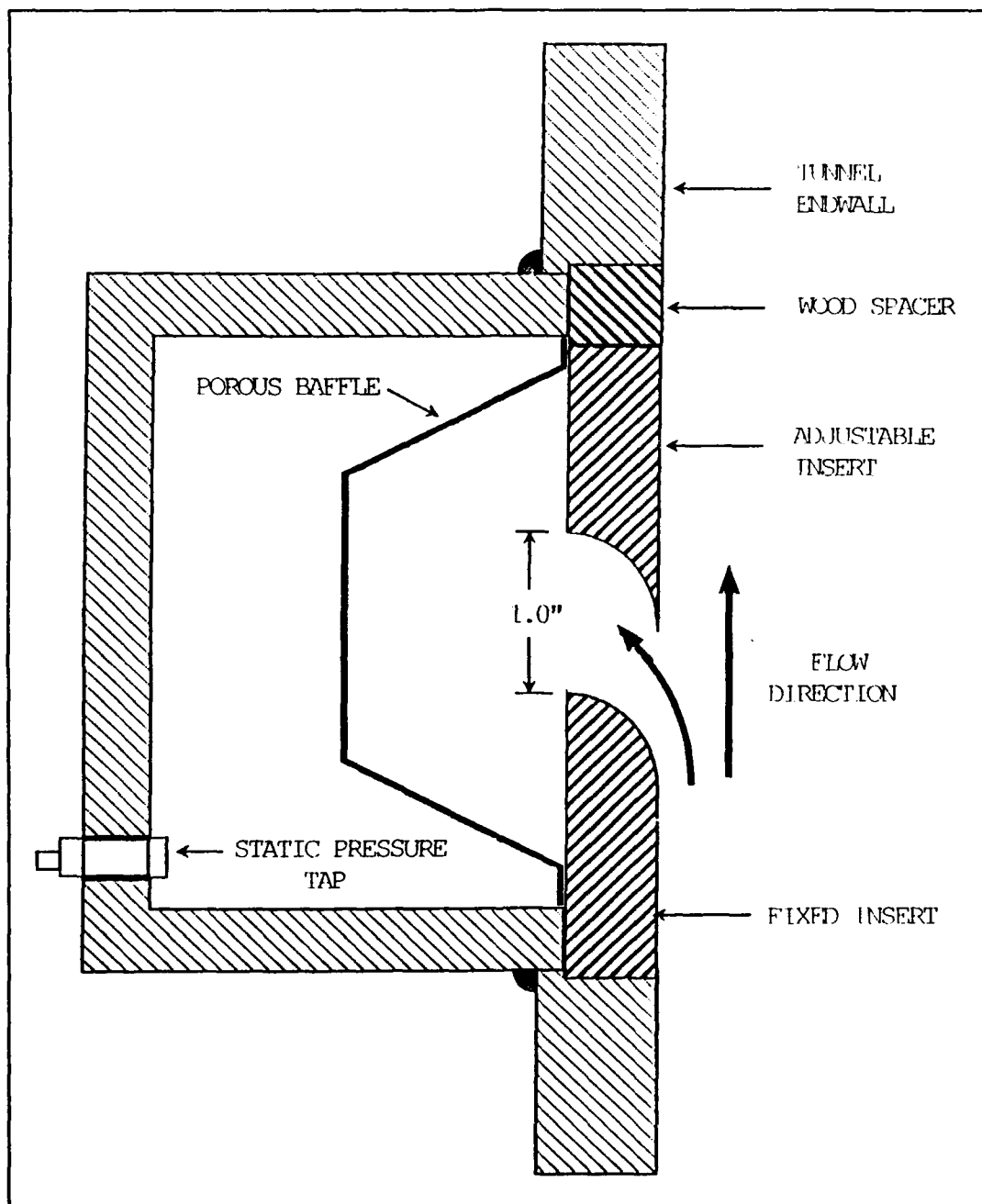
## 1. SUCTION SLOT

Flush mounted adjustable width slots were installed into the endwall manifolds. Figure 2 shows a top-down schematic of the tunnel suction configuration. A cross section of the slot, baffle and manifold arrangement is found in Figure 3. Each slot was comprised of two flat pieces of stock aluminum with a 5/8 inch radius machined on one edge. The pieces were attached together with machined cross-members that allowed for the variation of the slot height. A detailed drawing is found in Appendix D.

The endwall manifolds were each 120 inches and the test section was 60 inches. Since the inlet flow angle was set to  $44.4^\circ$ , the test section was not centered on the suction slots. Initial tests with the suction slot in the full open position showed that there was strong suction at the sidewall extremities but none at mid-pitch. Because of this, the slot was arbitrarily narrowed to one inch. Further testing showed that the suction improved in strength, however, the non-uniformity in the pitchwise direction remained. A porous baffle was attached to the backside of the slot to try and alleviate this problem. A detailed drawing of the baffle is found in Appendix D. In order to monitor the amount of suction applied across the manifold, eleven taps were drilled into the base of the manifold of which nine were attached to a water manometer. The nine pressure taps were centered on



**Figure 2.** Top View of the Suction Configuration



**Figure 3.** Suction Manifold Cross Section

the inlet to the test section. A dimensioned cross section is given in Appendix D. Figure 4 shows the end of the slot outside the tunnel which was taped over for the tests.

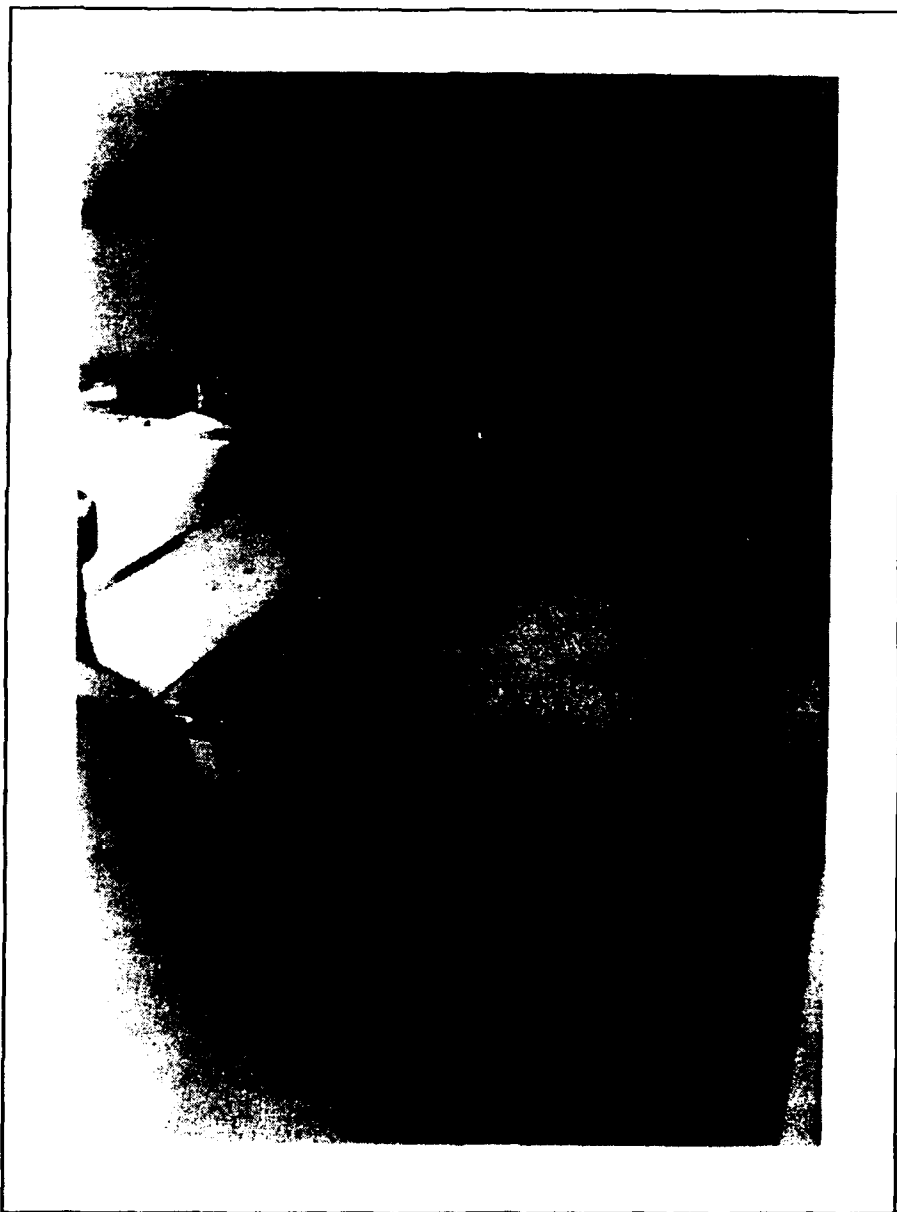
## **2. BLOWER CONFIGURATION**

A 30 horsepower Allis Chalmers electric motor driven centrifugal blower was placed behind the south wall of the wind tunnel. A photograph of the blower configuration is given in Figure 5. The blower inlet was connected to a cylindrical plenum having 6 hook-ups, three at a third of the length and three at the end. The blower plenum was connected to the tunnel by mounting four gate valves to each end of the tunnel manifolds and running flex tubing from the valves to the blower plenum. The gate valves allowed the amount of suction to be varied within the manifolds. The two unused attachments of the blower plenum were capped.

## **C. INSTRUMENTATION**

### **1. RAKE PROBE AND TRAVERSE**

A ten inch rake probe was designed to take detailed pressure distribution measurements across the span of the test section with a finer resolution at each end to better record the size of the endwall boundary layer. The probe assembly contained 20 individual probes, of which 17 were total pressure ports, numbered from the north endwall. The two central probes, (numbers 10 and 11), formed a yaw probe located at the midspan with a static pressure port,



**Figure 4.** Suction Slot Mounted to the South Endwall



**Figure 5.** Blower Configuration

(numbered 9), one inch off midspan. A detailed drawing of the probe is found in Appendix D. Figure 6 shows a plan-view schematic of the probe.

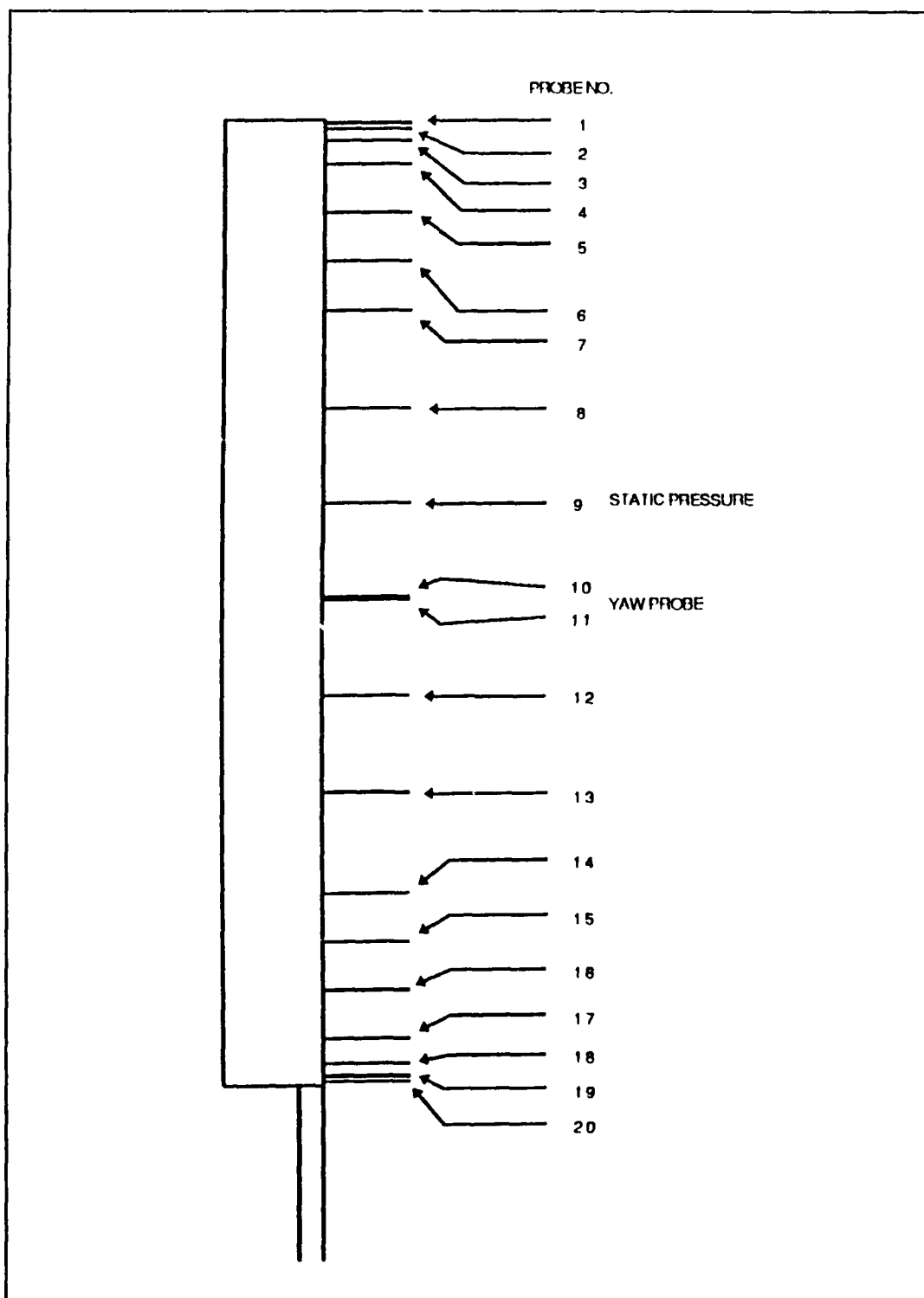
The rake probe was calibrated in the free jet located at the Turbopropulsion Laboratory. This was done by subjecting the probe to three different levels of dynamic pressure and varying the angle of incidence to obtain a calibration curve for the yaw probe. It was noticed that the static pressure port was subject to probe interference. Therefore, a correction factor was calculated to account for this difference. The calibration data and curves are given in Appendix C.

The rake probe was traversed both upstream and downstream of the test section. This was accomplished by mounting it to a traverse assembly which in turn was mounted to the side of the south wall of the tunnel, as seen in Figure 1. The traverse was equipped with a geared counter used to measure the distance. Figure 7 shows the rake probe mounted to the upper traverse. It also shows the relative position of the suction manifold to the test section, upstream, downstream and manifold pressure taps.

## **2. LDV SYSTEM**

The two component LDV system used for the base line tunnel configuration is described in detail by Murray [Ref.14].





**Figure 6.** Rake Probe Plan View



**Figure 7.** Rake Probe Mounted to Downstream Traverse

For the modified tunnel tests, a different LDV system was used to measure the flow under the influence of high suction. The new system used a similar laser light source, however, the rest of the optical system was primarily fiber optics. A complete description of the system is given by Dober. [Ref. 15] Figure 8 shows the test passage bounded by blade numbers 7 and 8 as seen through the optical window. Coordinates for each survey station are also given.

### **3. INSTRUMENTED CD BLADE**

Blade pressure measurements were taken with a fully instrumented blade, located at mid-pitch, that contained 39 static pressure taps, 19 on each surface and one at the leading edge. Figure 9 shows the tap locations.

### **4. STATIC PRESSURES**

Endwall static pressure measurements for both the north and south walls, upstream and downstream as shown in Figure 1, were measured with two banks of water manometers. The manometers were also used to monitor the pressures at the suction manifolds, gate valves, as shown in Figures 2 and 3, and plenum and Prandtl total pressures. ( A Prandtl probe was used to monitor inlet dynamic pressure near to blade 15 ). Each manometer consisted of 50 tubes and the arrangement of pressure taps for each is given in Figures 10 and 11.

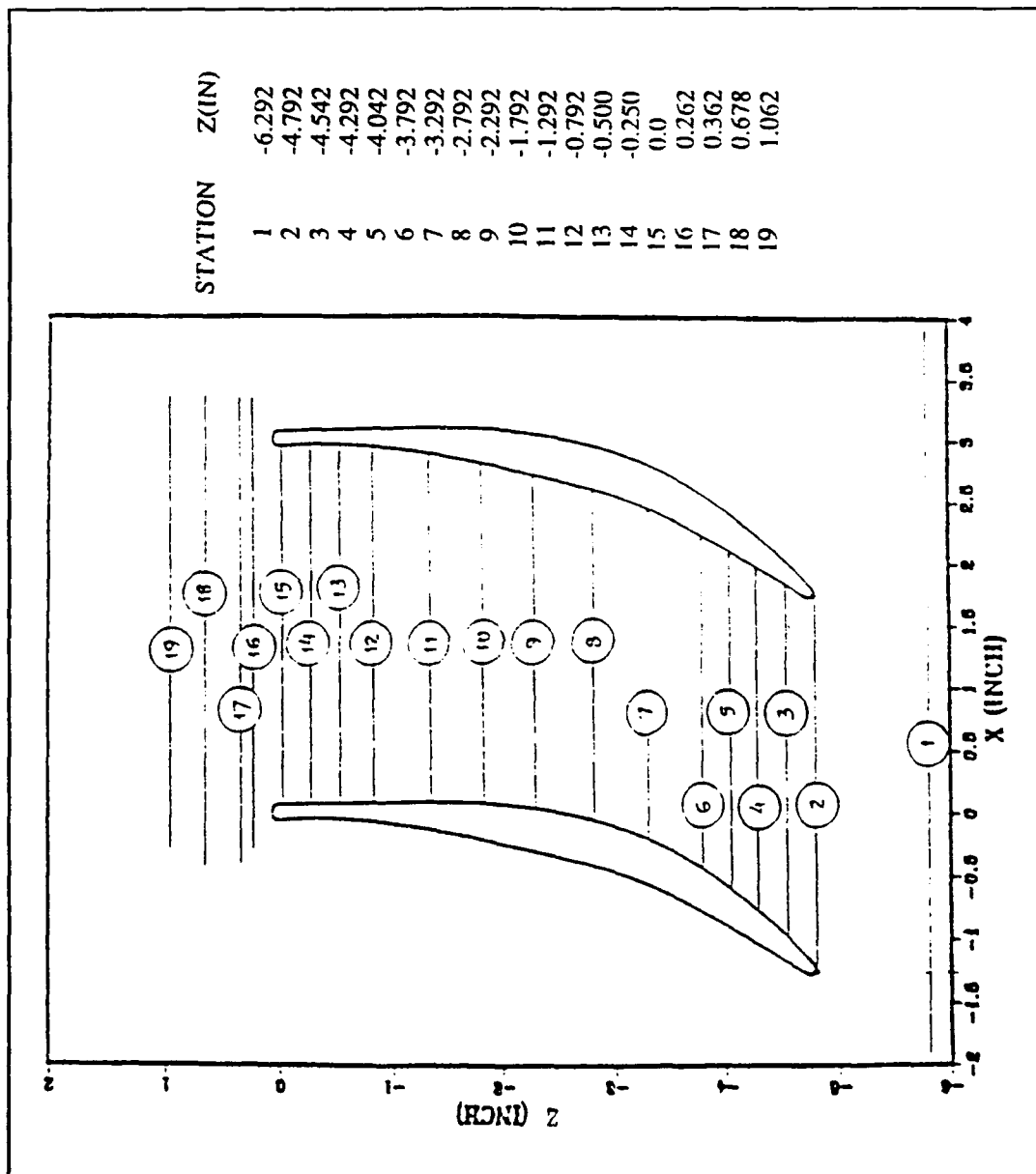
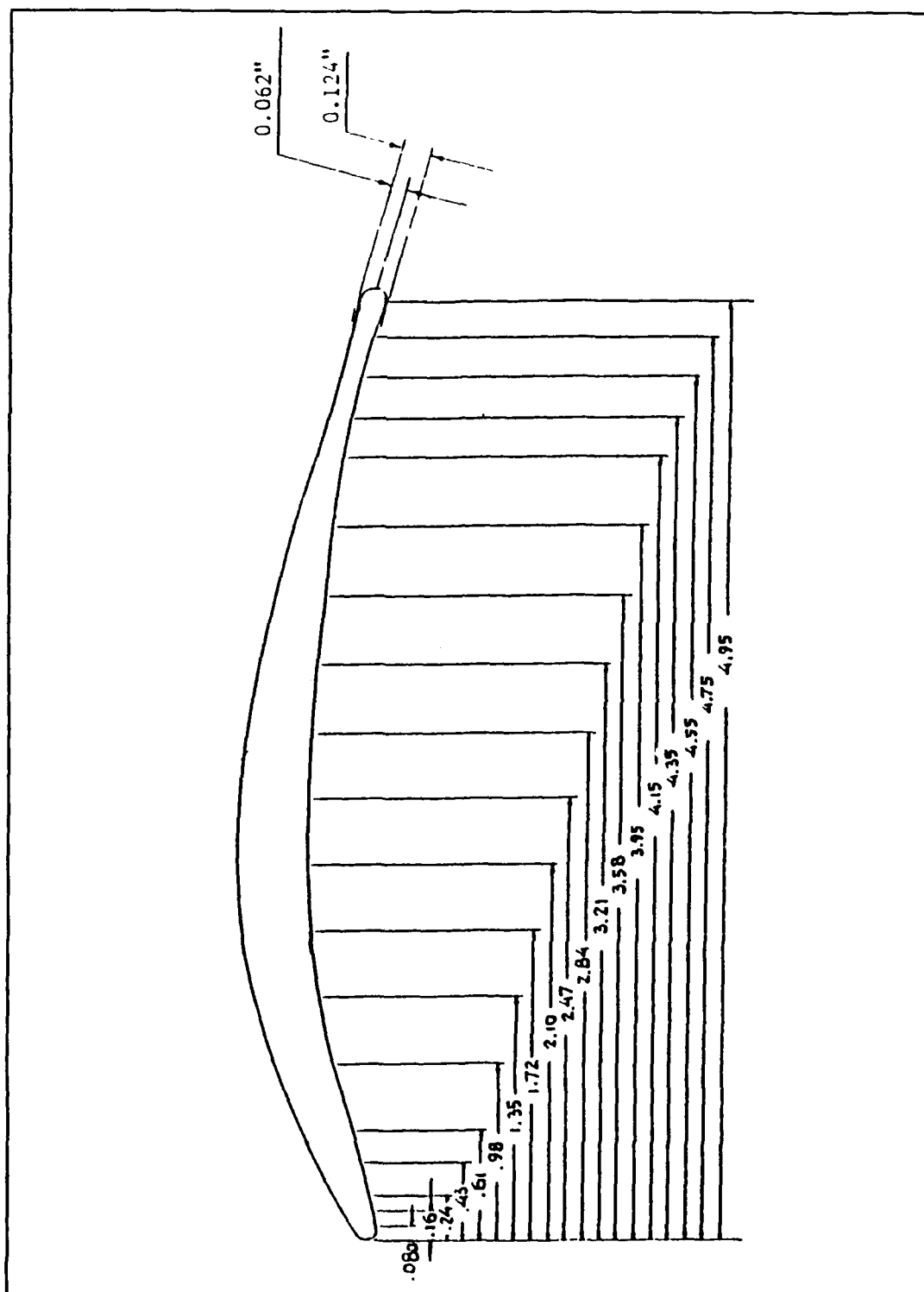


Figure 8. LDV Test Section Geometry



**Figure 9.** Instrumented Blade Pressure Tap Location

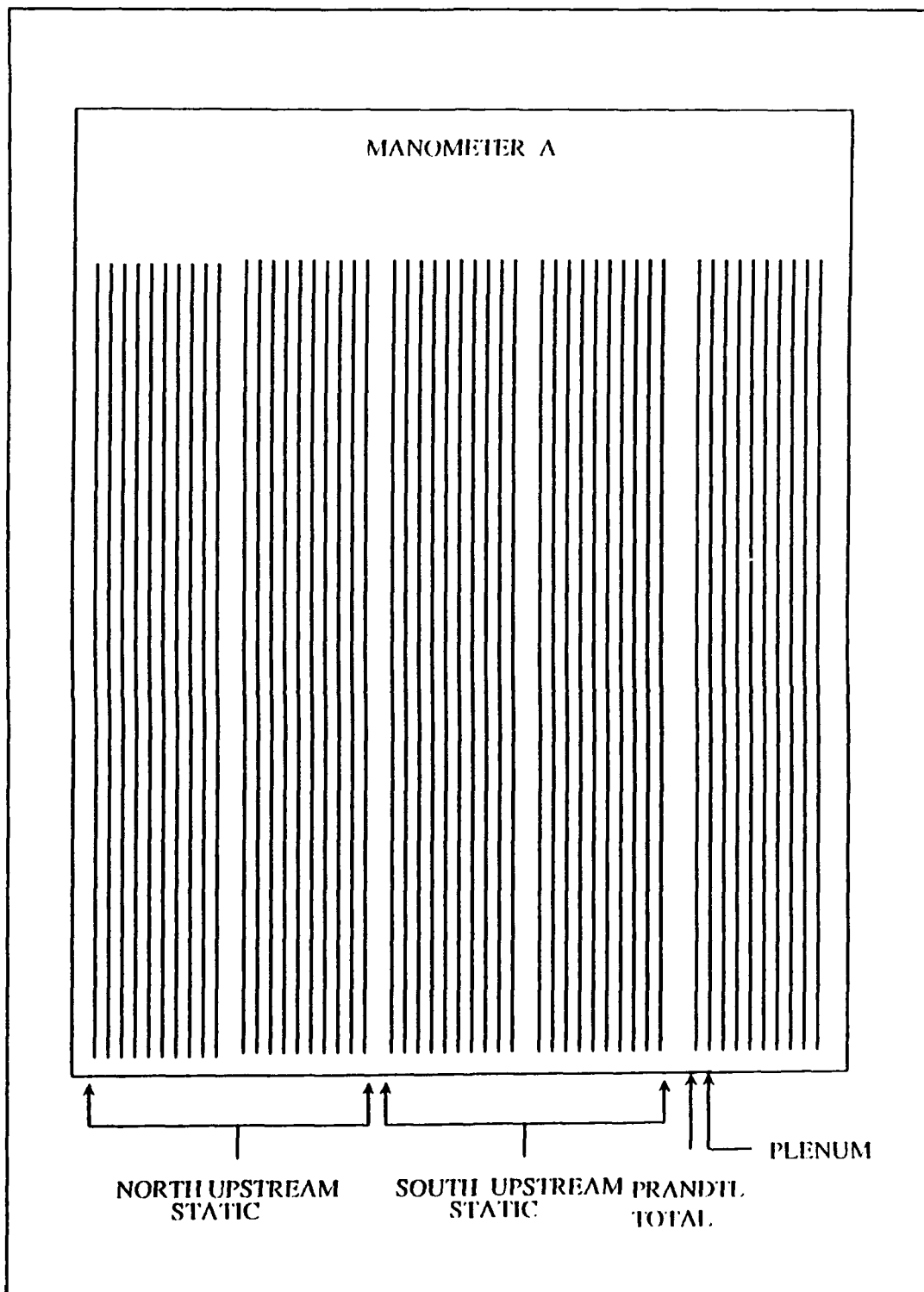
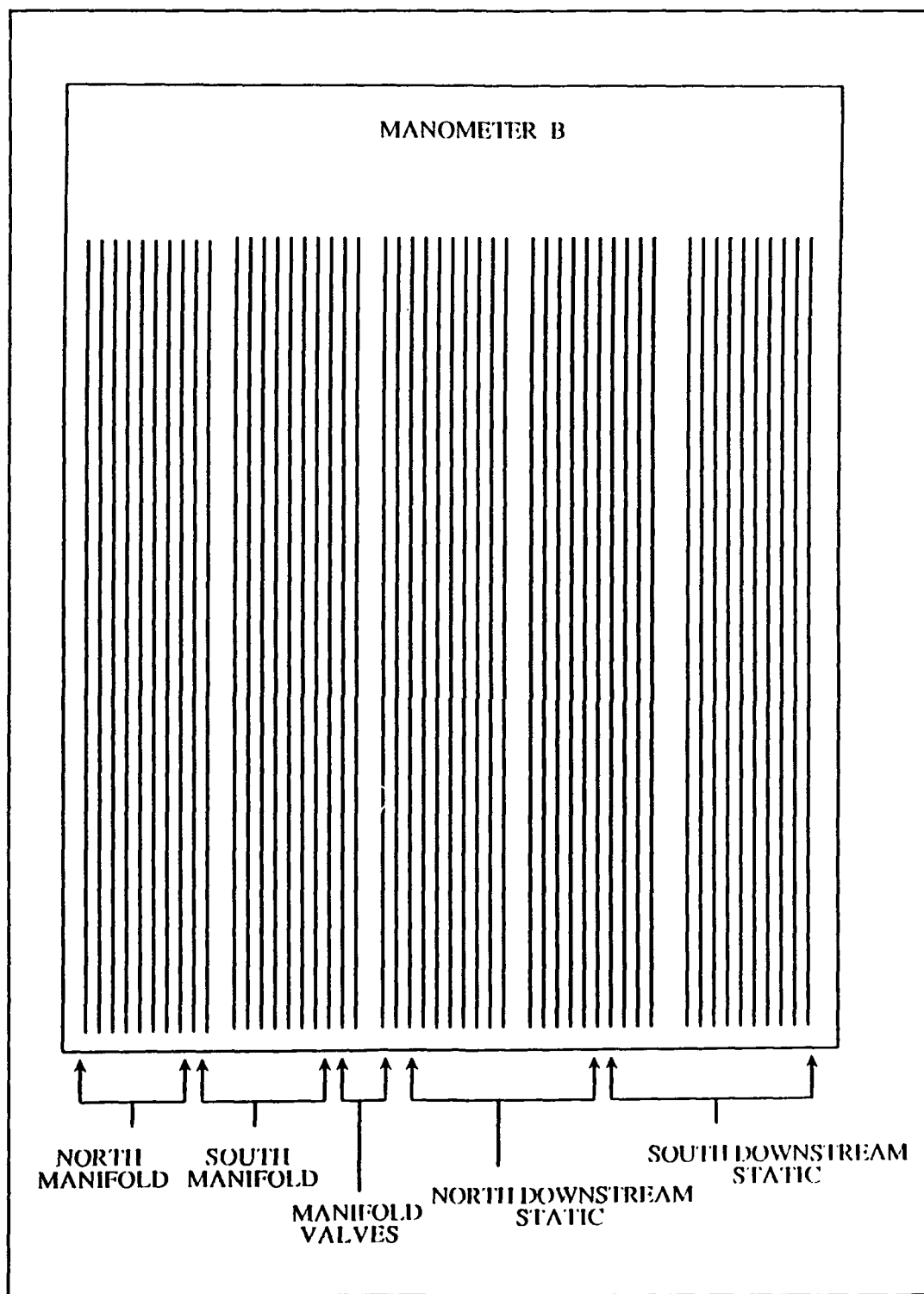


Figure 10. Manometer A. Pressure Tap Arrangement



**Figure 11.** Manometer B. Pressure Tap Arrangement

## **D. DATA ACQUISITION**

### **1. HARDWARE**

The data acquisition system consisted of two 48 channel scanivalves that were connected through an interface to a Hewlett Packard Data Acquisition System (HP-3052). The scanivalves were controlled by the HG-78k controller and is discussed in detail by Classick [Ref. 10]. A Hewlett Packard 900 series 300 computer controlled the system. Scanivalve 0 was connected to the instrumented blade along with the plenum pressure. Scanivalve 1 was connected to the rake probe as well as the plenum and Prandtl pressures.

### **2. SOFTWARE**

Two data acquisition programs were used to record the various tunnel pressures. The program ACQUIRE written by Classick [Ref. 10] was used to take the instrumented blade measurements. The program ACQUIRE4 was a fourth adaptation to the same program and its basic function was to obtain the spanwise rake pressure distribution at various pitchwise locations. In post-processing, the program produced a spanwise  $C_p$  plot and finally plotted a  $C_p$  surface distribution of the test section flow field. The program is fully described in Reference 15. The fourth modification consisted of adding equations to the program to calculate air flow angle, total velocity and axial velocity.



### III. EXPERIMENTAL PROCEDURE

#### A. BASE LINE CONFIGURATION

The procedure used to adjust the inlet flow angle of the cascade using the LDV system is well documented by Murray [Ref. 14]. Using this procedure, the inlet flow angle was measured to be  $44.4^\circ$  with the sidewalls set at  $43.4^\circ$  and inlet guide vane position at 50mm. LDV measurements were taken at stations 1, 2, 7, 15 and 18 to compare with earlier data [Ref. 9] and to produce a basis for later comparison with results with endwall suction. The plenum pressure was maintained at 12.0 inches of water with a deviation of  $\pm 0.1$  inches.

Inlet surveys were taken by mounting the rake probe to the upstream traverse position, as shown in Figure 1, and traversing the entire pitch-wise distance referenced from the east sidewall. Appendix A. contains the data taken at each location.

The AVR was obtained by surveying across one blade passage starting at 20 inches and terminating at 23 inches in  $\frac{1}{4}$  in. increments. This was done at both upstream and downstream locations.

#### B. SUCTION PROCEDURE

Three levels of suction were used to compare with the base line configuration. The levels were measured by the static

pressure taps located at the gate valves at each end of the suction manifolds. The blower was always operated at full speed with its downstream throttle valve in the full open position. When the blower was turned on, the plenum pressure dropped as much as one inch of water in the case of high suction. Therefore, the Prandtl total pressure was used as  $P_{ref}$  for all suction tests. This was kept at 10.5 inches of water, the same as the base line tests.

The procedure to set up suction was as follows: (a) start the tunnel and stabilize at 10.5 inches of total pressure on the Prandtl probe. (b) Fully open the gate valves and switch on the blower. (c) Close the valves to obtain equal and stable static pressure at the valves. (d) Adjust the plenum back to 10.5 inches on the Prandtl probe. All four valve static pressures measured -32.0 inches of water for high suction, -20.0 inches for medium suction and -10.0 inches for low suction. The maximum deviation allowed was .5 inches of water for any of the valves. Polaroid photographs were taken of the manometers for documentation purposes at each suction level. Example photographs are given in Appendix A.

The same measurements were taken for suction as for the base line. The instrumented blade measurements were taken only with suction. The LDV measurements were taken only at the inlet and outlet planes and referenced with respect to the Prandtl total pressure, ambient pressure and plenum pressure to calculate  $V_{REF}$  as described by Elazar [Ref. 9].

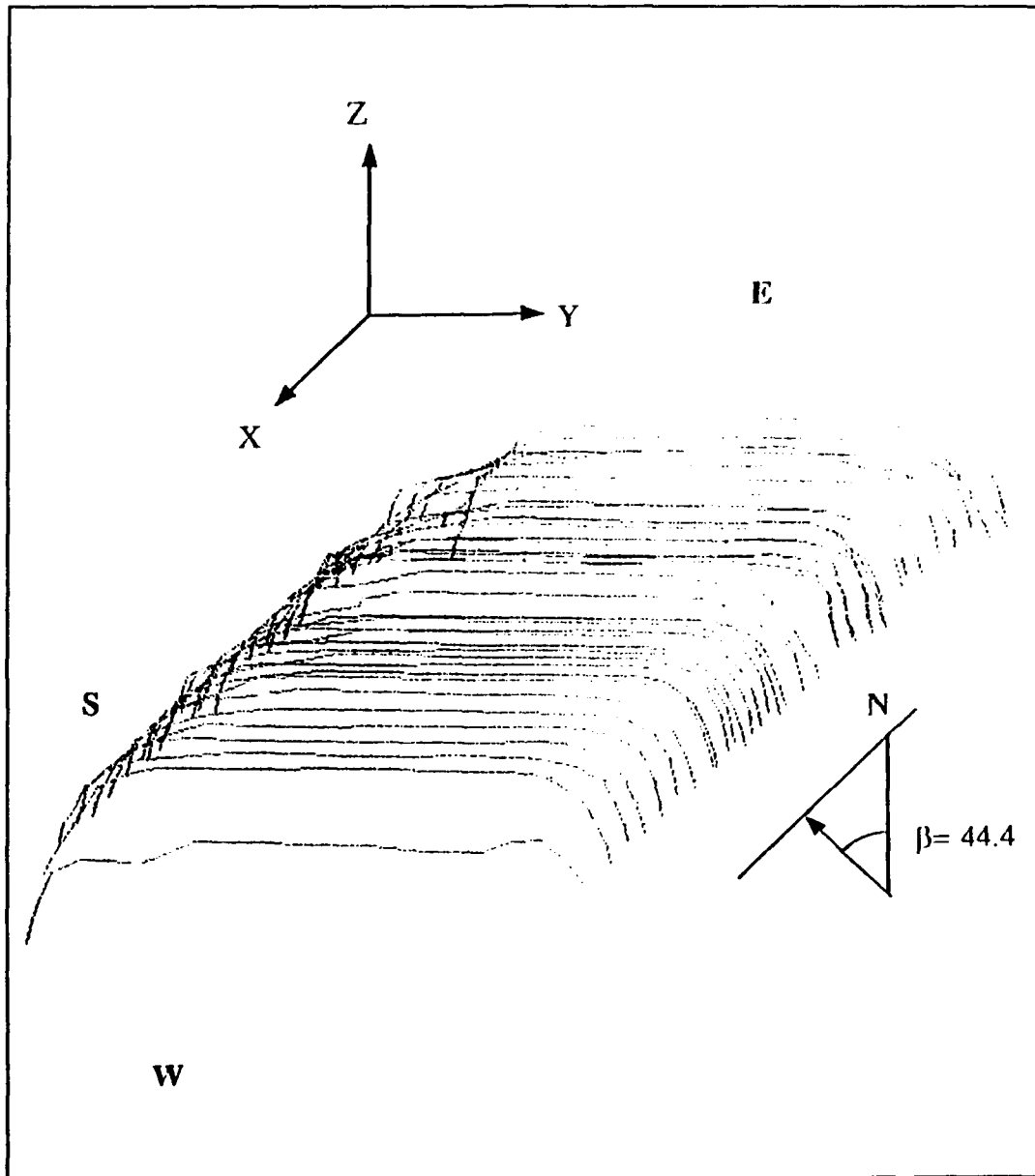
#### IV. RESULTS AND DISCUSSION

All data and equations for the graphs given in this chapter are found in Appendices A and B respectively.

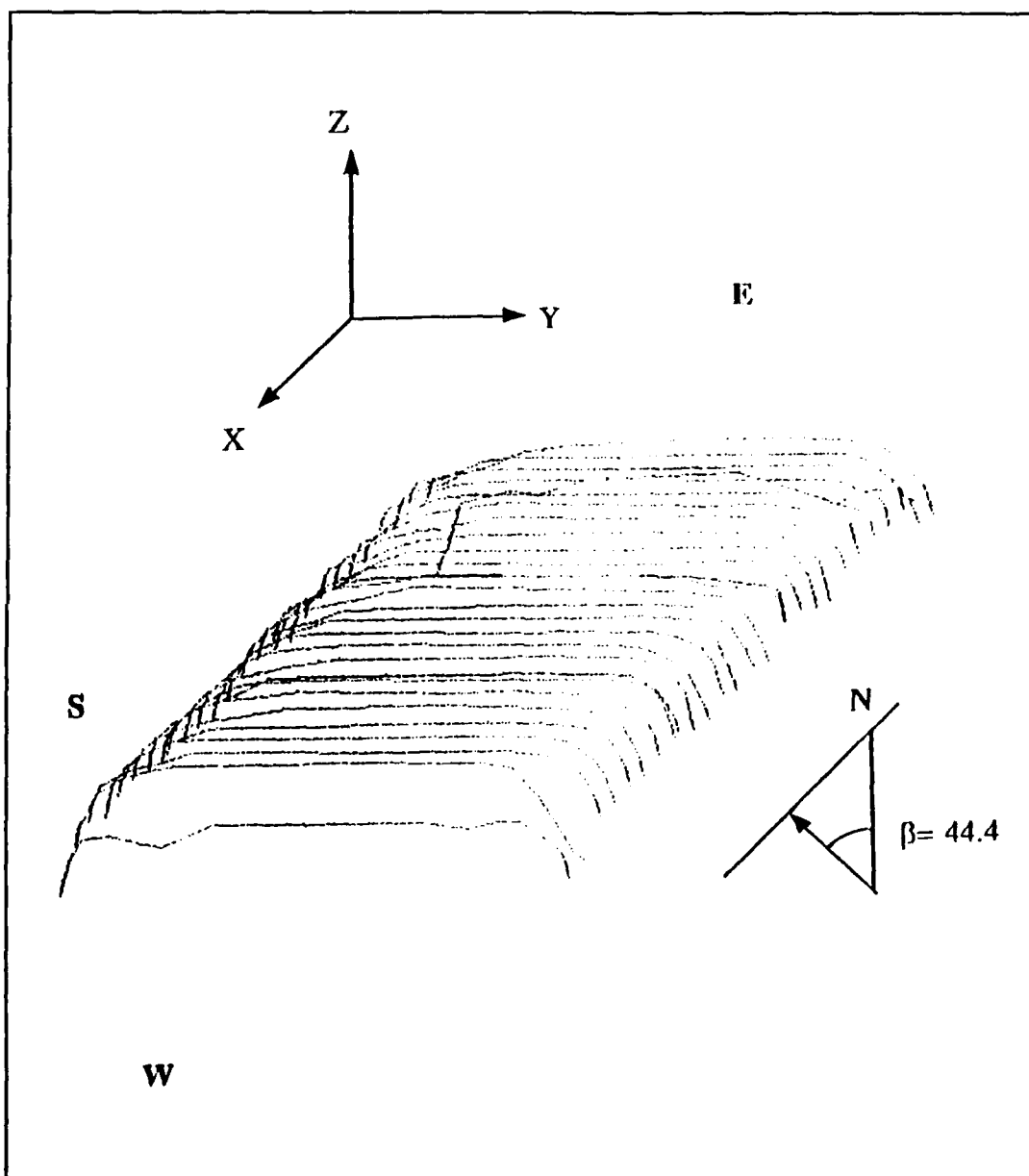
##### A. INLET SURVEYS

Inlet carpet plots are seen in Figures 12 through 15. These show that the inlet core flow uniformity is the same for the base line and at all levels of suction. Figure 12 is the full survey for the base line configuration. Figure 13 is for low suction, and shows the endwall boundary layers are being removed at the east and west ends of the test section with the minimum reduction at the mid-pitch location. Figures 14 and 15 are for medium and high suction respectively and show the same trend, giving the first indication of non-uniform suction in the pitchwise direction.

Plots for the spanwise pressure distribution are shown in Figures 16 through 20 for different pitch locations. These locations were chosen as a result of the carpet plots and the locations chosen for displacement thickness and AVR calculations. Figure 16, at 11.8 in. pitch, shows the increased boundary layer removal for increased suction at both end walls. However, the north wall boundary layer was not reduced to the same degree as the south wall. The base line showed the north wall boundary layer was thicker than that on



**Figure 12.** Inlet Flow Carpet Plot for Base Line Configuration



**Figure 13.** Inlet Flow Carpet Plot for Low Suction

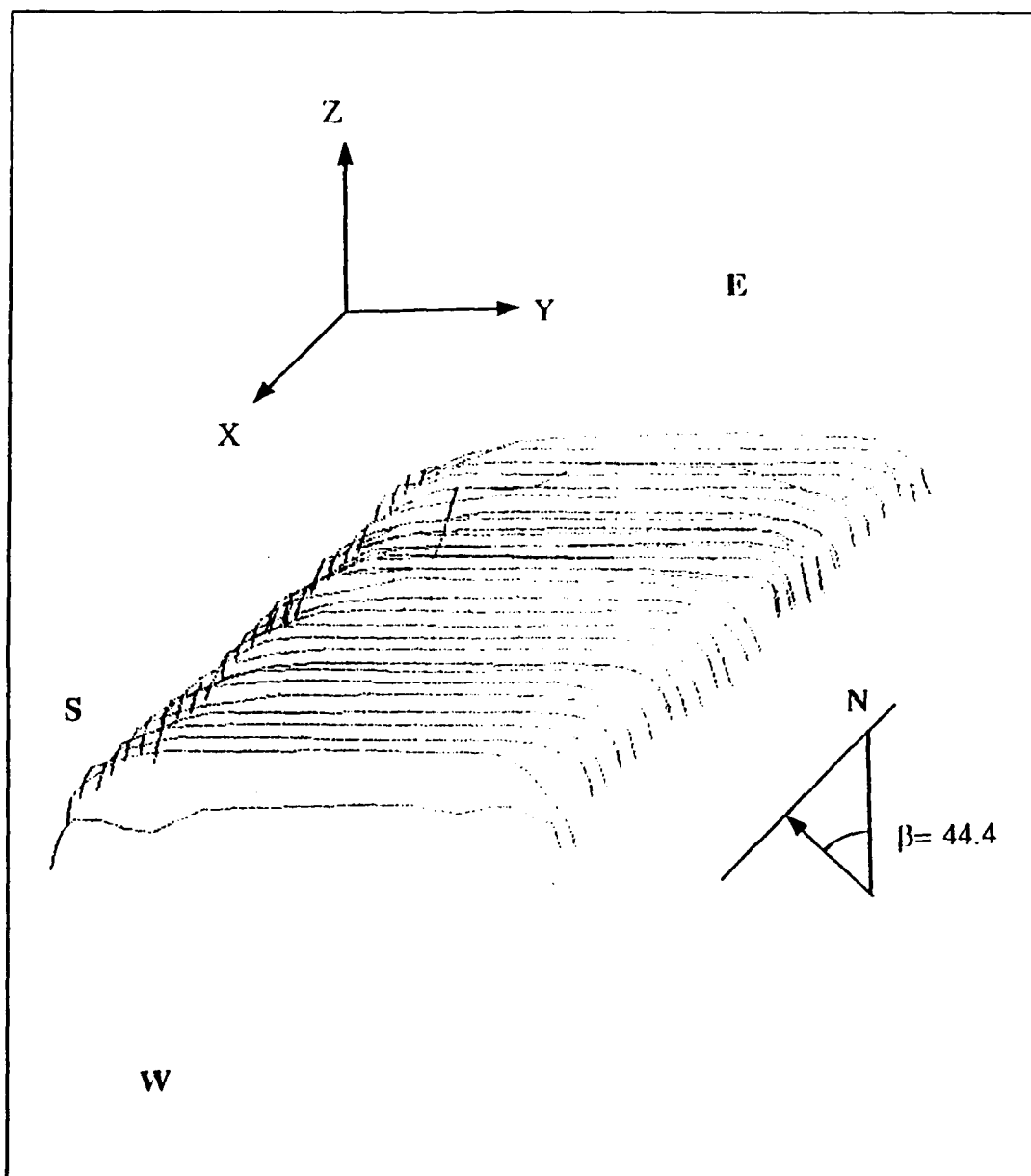
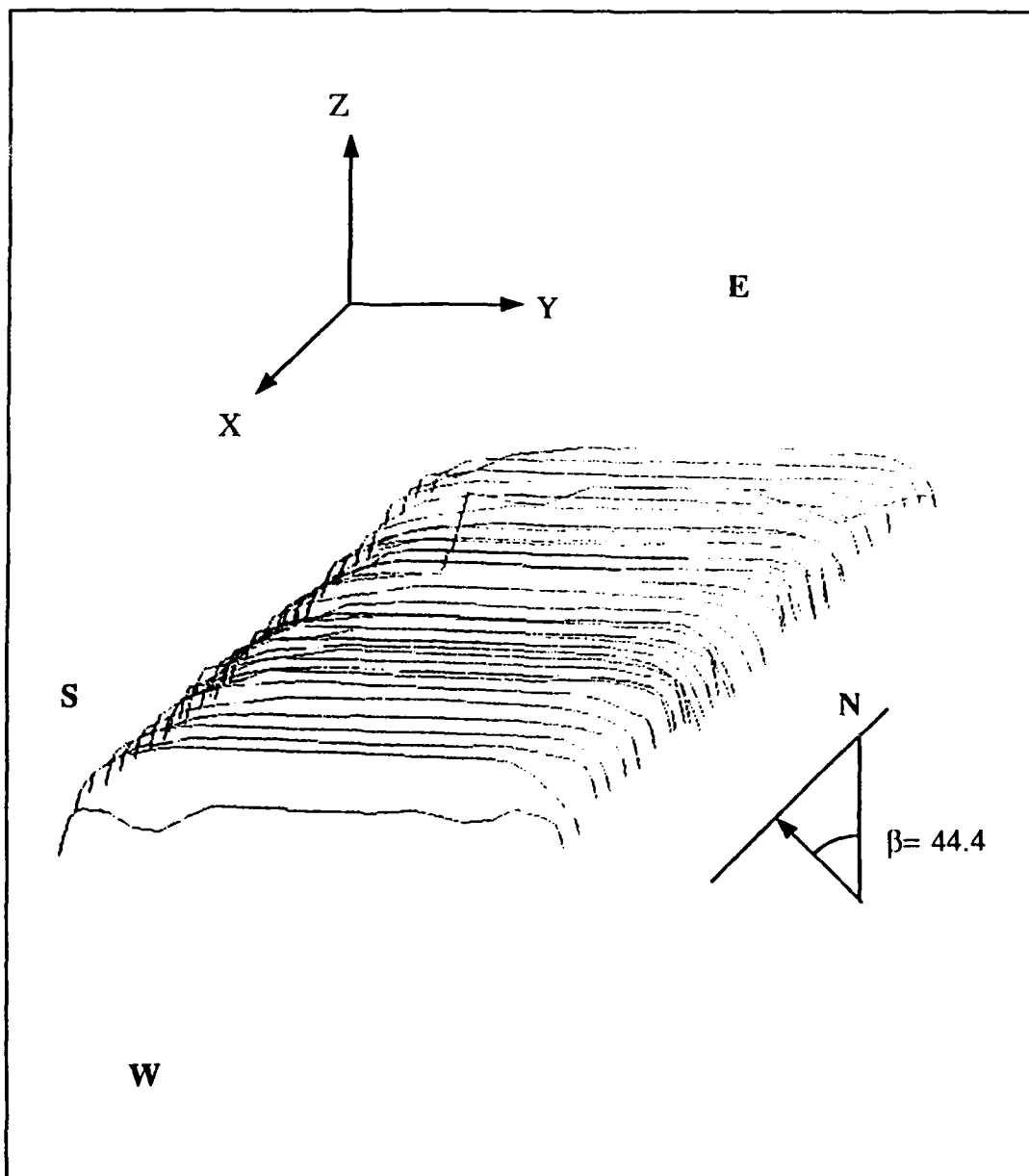
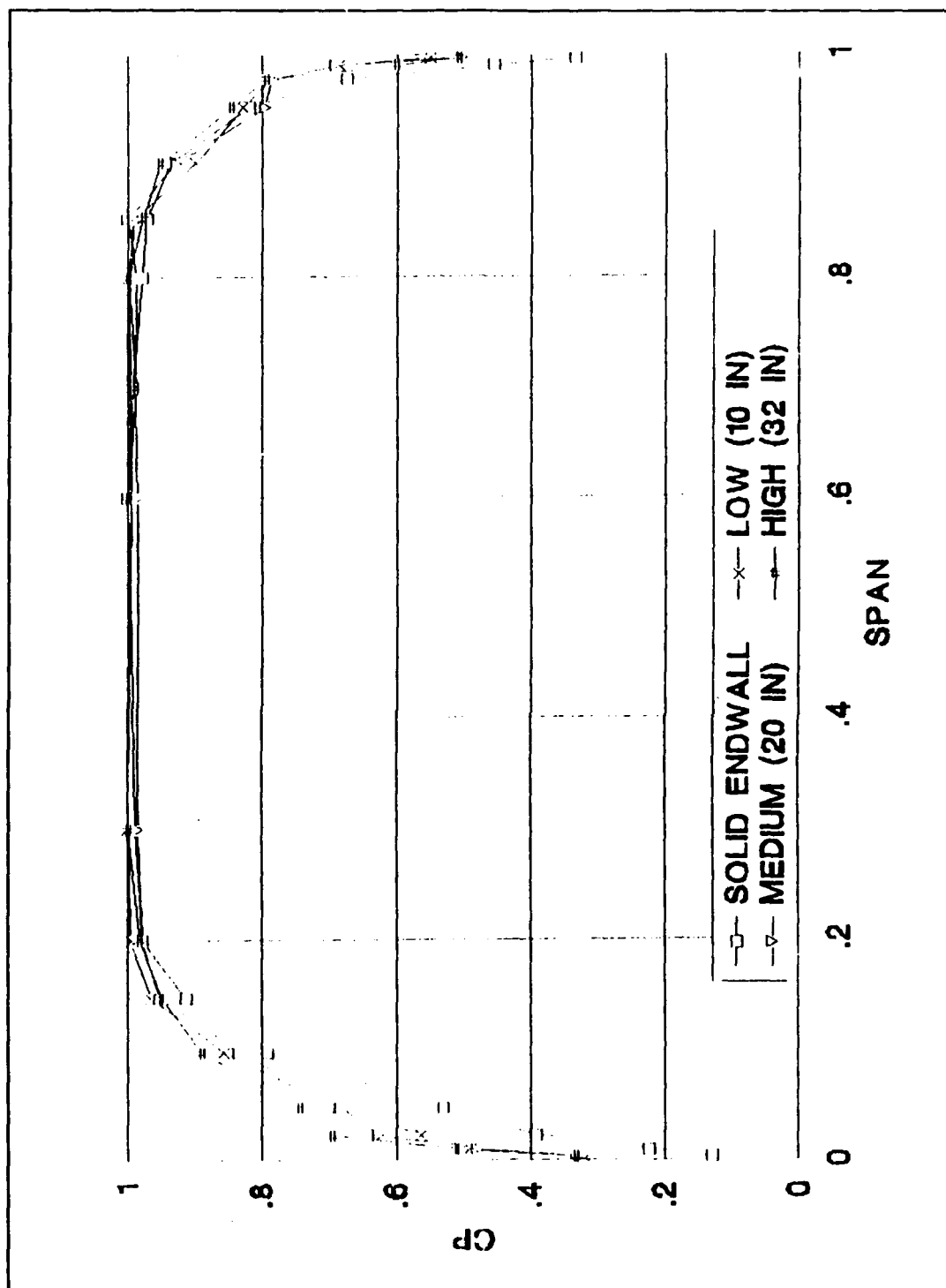


Figure 14. Inlet Flow Carpet Plot for Medium Suction



**Figure 15.** Inlet Flow Carpet Plot for High Suction



**Figure 16.** Superimposed Spanwise Pressure Distribution at 11.8 in. Pitch for Increased Suction



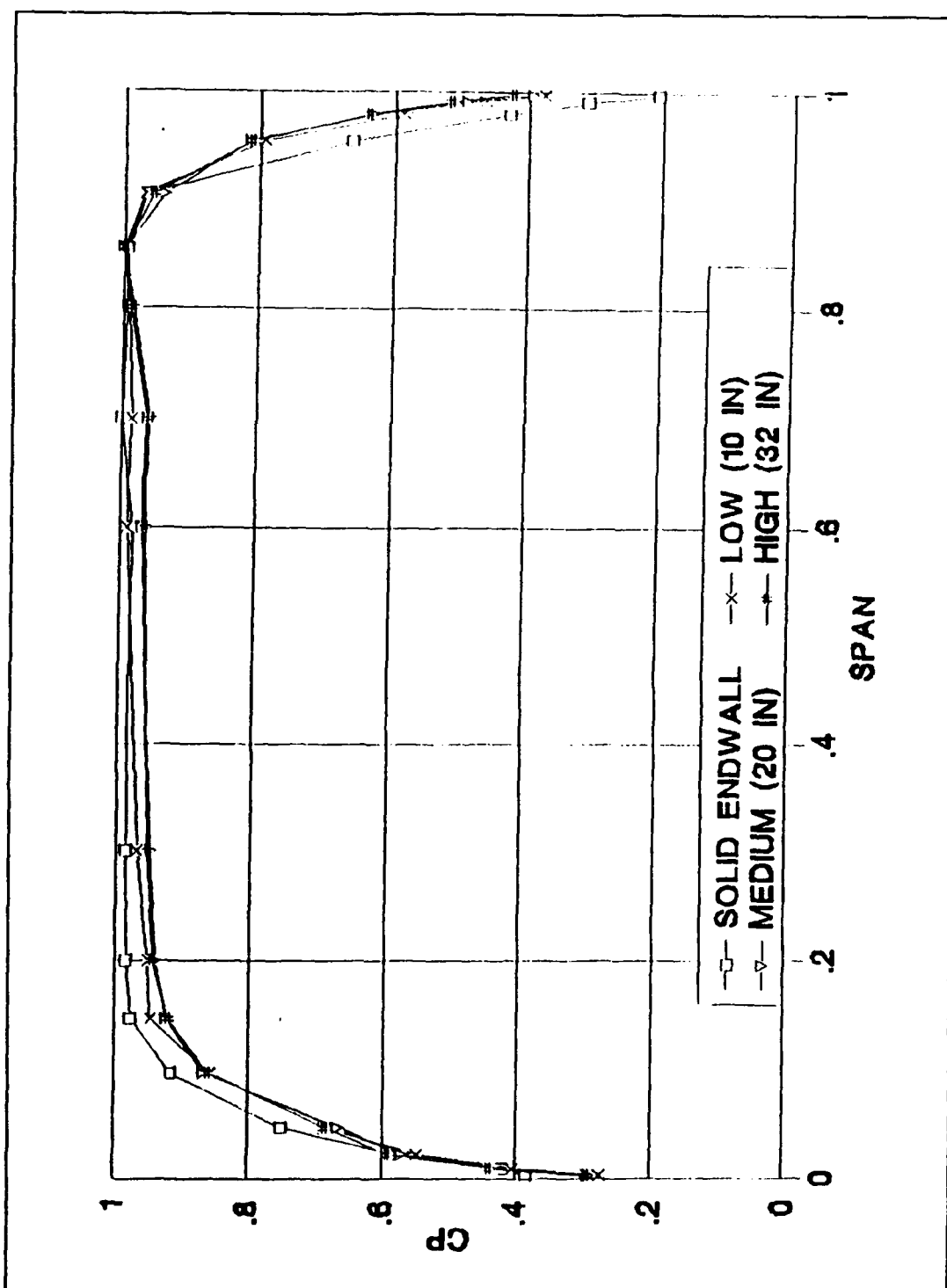


Figure 17. Superimposed Spanwise Pressure Distribution at 20 in. Pitch for Increased Suction

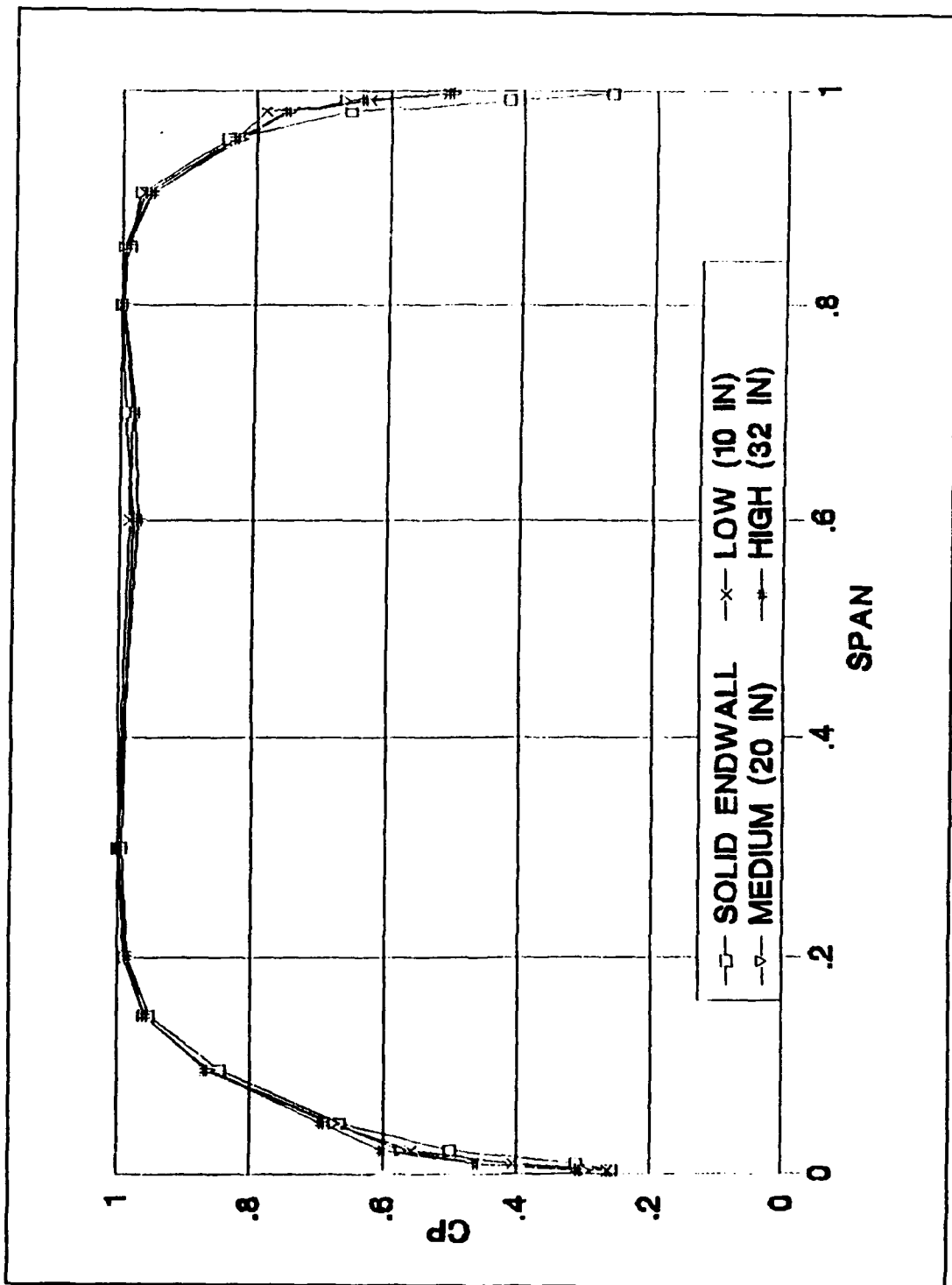


Figure 18. Superimposed Spanwise Pressure Distribution at 22 in. Pitch for Increased Suction

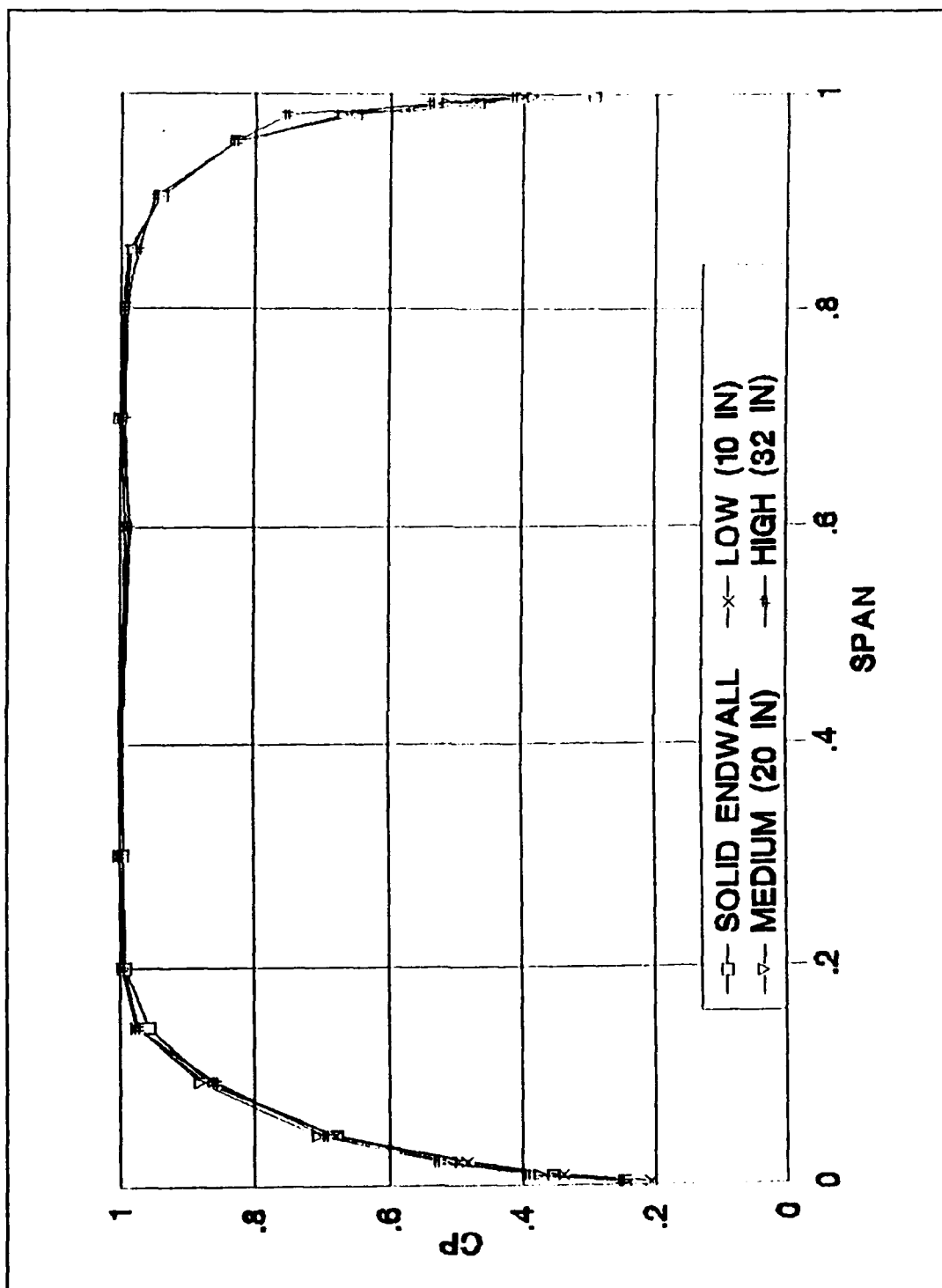
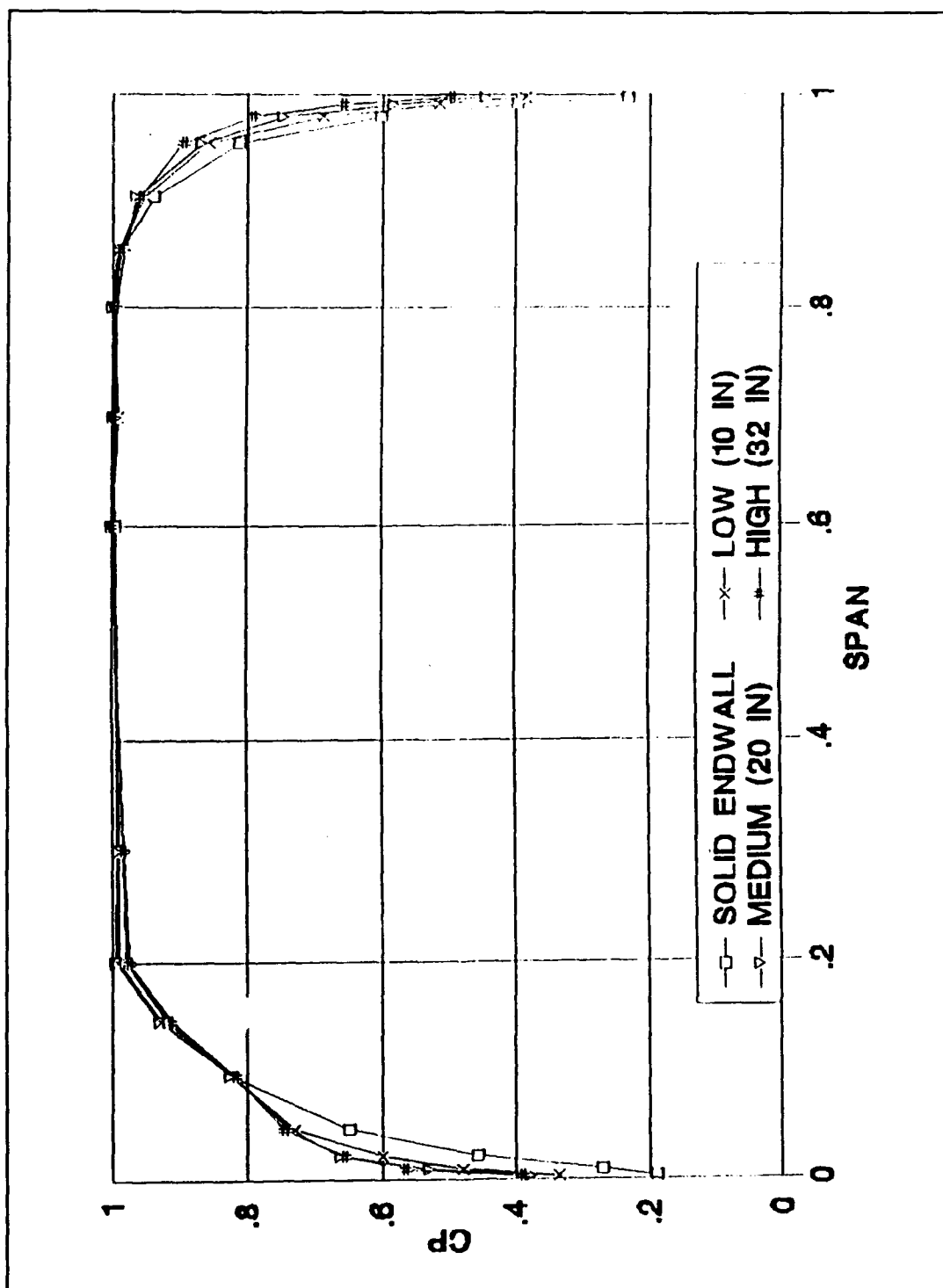


Figure 19. Superimposed Spanwise Pressure Distribution at 35.88 in. Pitch for Increased Suction



**Figure 20.** Superimposed Spanwise Pressure Distribution at 50.9 in. Pitch for Increased Suction

the south wall. This characteristic is seen in all the plots. In Figure 17, at 20 inches, the core flow is seen to have been shifted towards the south wall and the boundary layer to have thickened at the north wall from an initial displacement thickness of 0.133 inches to 0.146 inches. This happened only at the 20 inch pitch location. Furthermore, the removal of the boundary layer is seen to decrease moving towards mid-pitch. Figures 18 and 19 show reduced effects to the boundary layer which was also shown in the carpet plots. At the west end of the test section, Figure 20 shows the boundary layer removal again has a noticeable effect.

The displacement thickness was calculated for each level of suction and is shown in Figure 21. The value plotted is the average obtained over three inches from 20 to 23 inches in pitch. The north wall displacement thickness was reduced 0.0255 inches, (13%), from 0.1964 inches to 0.1709 inches. The south wall was decreased by .0591 inches, (39.5%), from 0.1497 inches to 0.0906 inches. The asymmetry of the inlet boundary layer displacement thickness is also seen as characteristic of the tunnel base line configuration. For the base line, there was a difference between the endwalls of 0.0467 in. and this increased to 0.0803 at high suction. Thus, there was a spanwise asymmetry of the boundary layer removal.

After viewing these data, it was evident that equalizing the gate valve pressures might not have been the best method

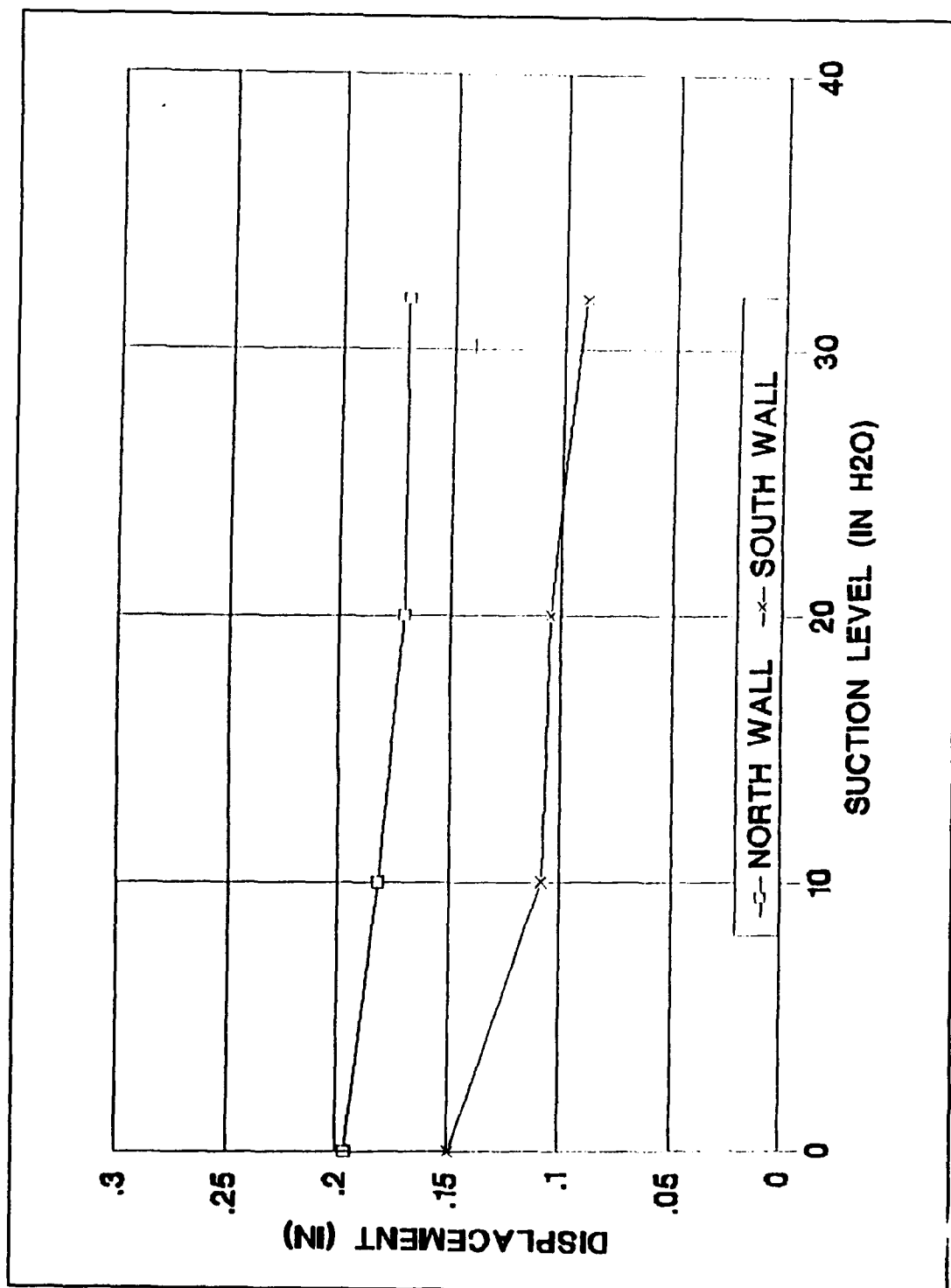
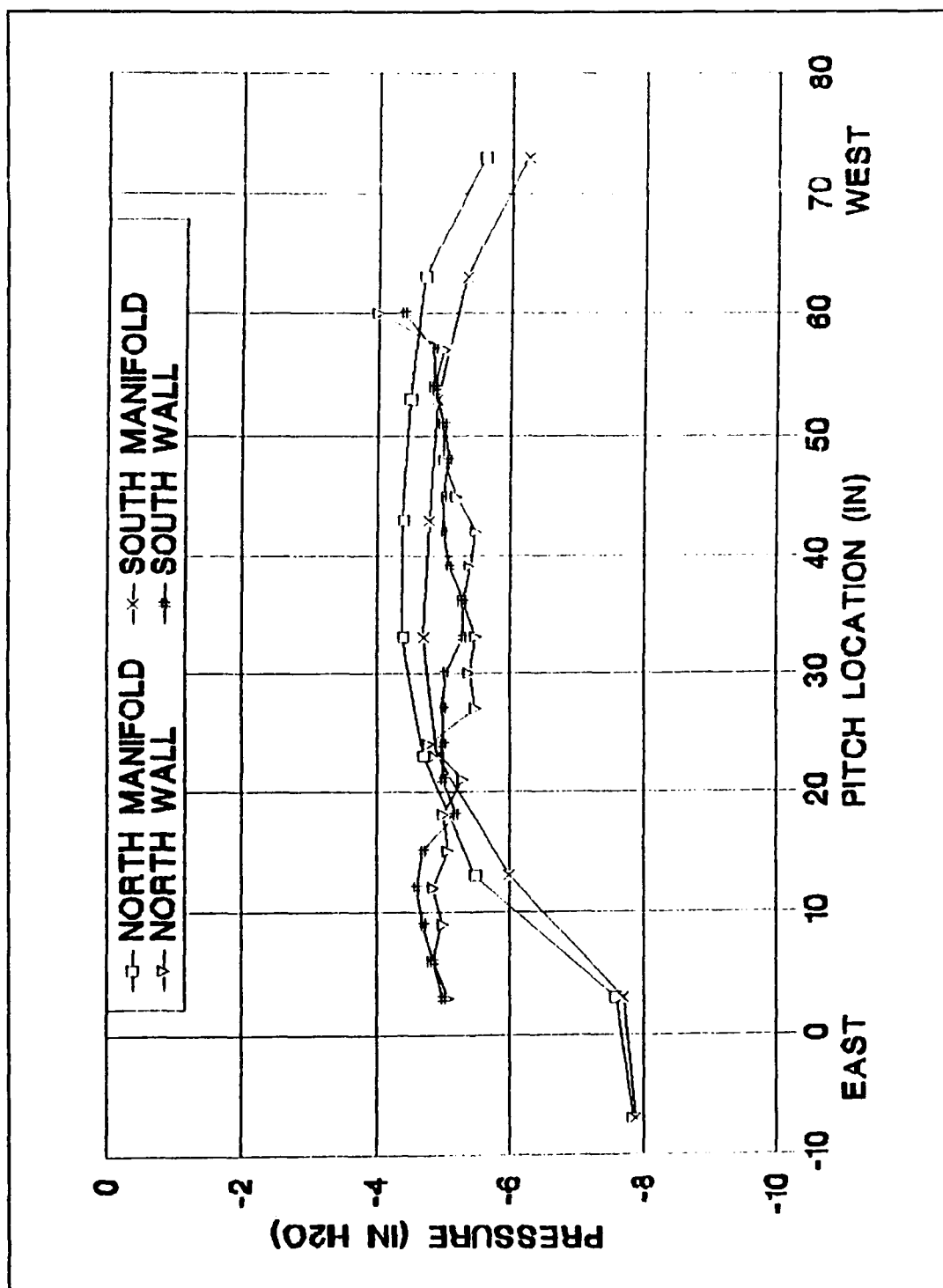


Figure 21. Average Displacement Thickness for the North and South Endwalls, 20 to 23 in. Pitch

of adjusting the suction. An additional test was conducted, but this time the manifold pressures were balanced with each other. This was accomplished by opening the north manifold valves fully and closing the south manifold valves to the medium suction level. This yielded a north endwall displacement thickness of 0.1767 inches and a south endwall displacement thickness of 0.1038 inches. The difference between these is 0.0729 inches, a 0.0074 inch reduction in the endwall boundary layer thickness difference.

Further illustration of the non-uniformity in the suction is seen in Figures 22 through 24. These show the suction manifold static pressures at their pitch locations for each level of suction. Figure 22 shows the manifold pressures were well below the endwall static pressures at the east end of the test section. This is the region where suction showed the most benefit, as seen in Figure 16. It also shows that the pressure differential between manifolds was at a minimum. In the mid-pitch region, the manifold pressure was above the endwall static pressures. This was the region where suction showed the least effect, as seen in Figure 19. The pressure differential between the manifolds is seen also to have increased. In the west end of the test section, the manifold pressures drop below the endwall static pressures but not to the same degree as the east end. This region showed effects of boundary layer removal as seen in Figure 20. The pressure differential between the manifolds also reached a maximum in



**Figure 22.** Pitchwise Manifold and Endwall Static Pressure Variations for Low Suction



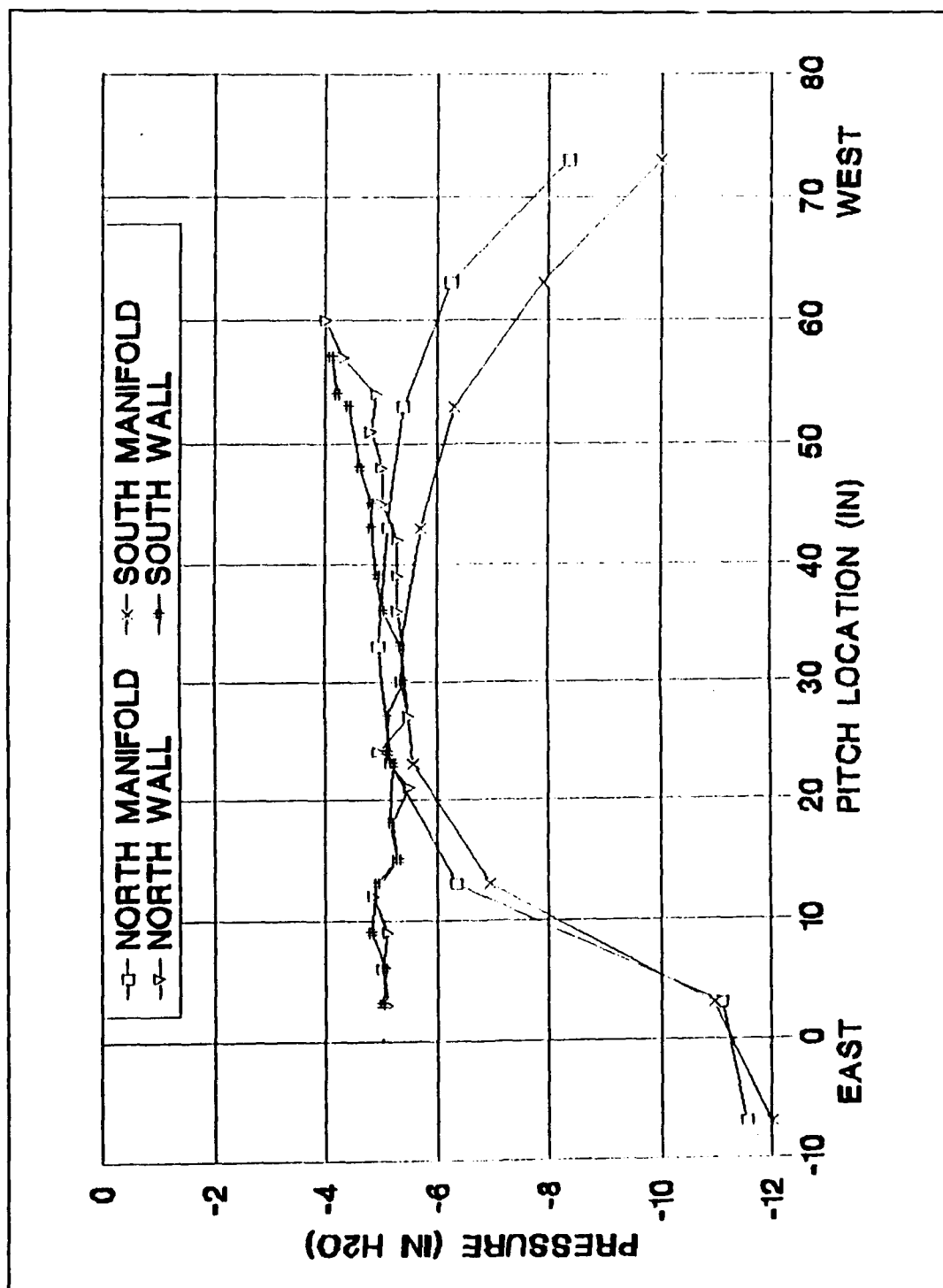


Figure 23. Pitchwise Manifold and Endwall Static Pressure Variation for Medium Suction

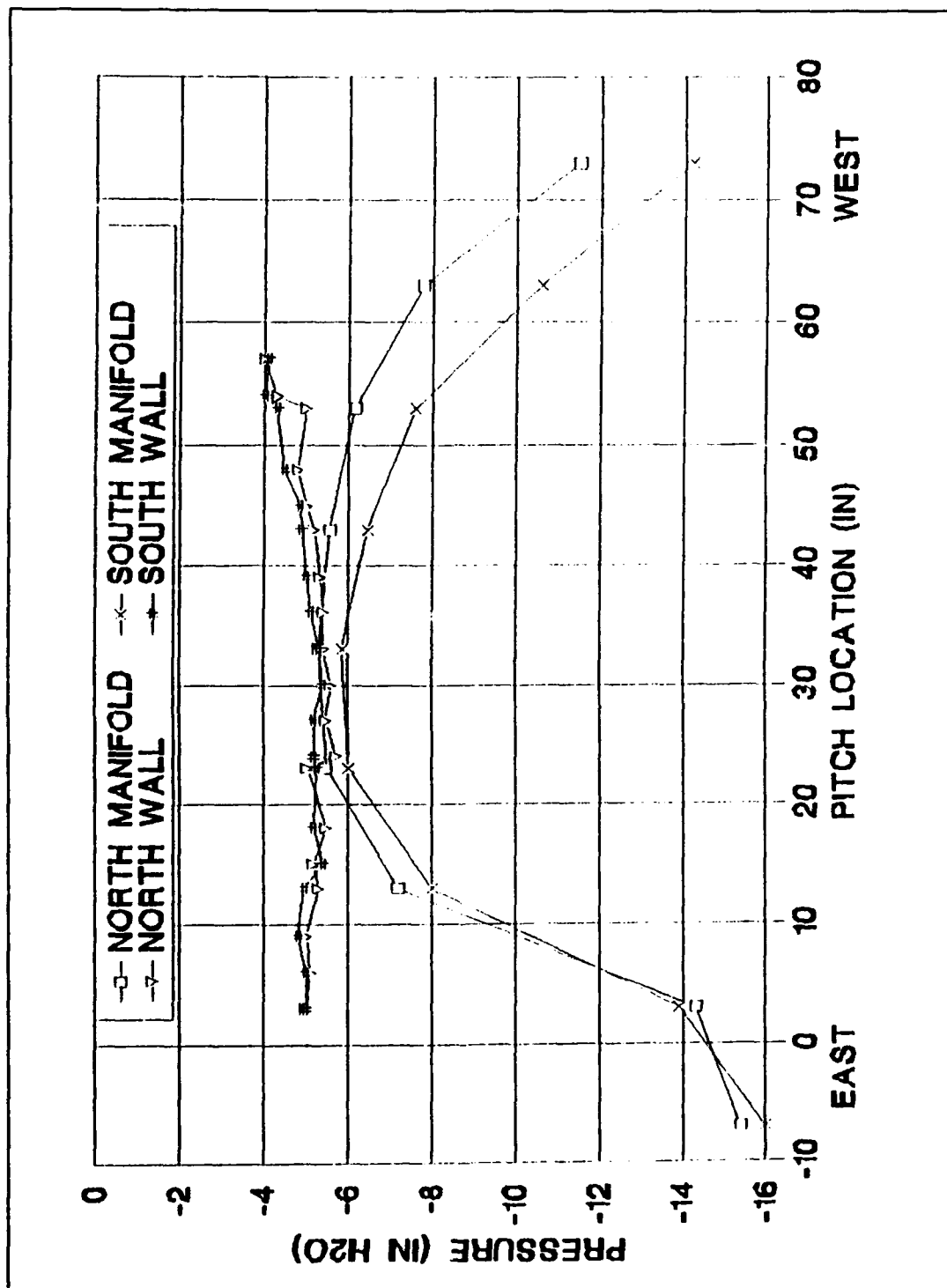


Figure 24. Pitchwise Manifold and Endwall Static Pressure Variation for High Suction

this region. The effect of increased suction, shown in Figures 23 and 24, shifted the manifold pressures down, until at high suction the mid-pitch minimum equalled the endwall static pressures. The endwall static pressures remained unchanged for all levels of suction. It is still unclear as to the relationship between the end wall and the manifold static pressures. It is noted that the boundary layer did not thicken at mid-pitch locations as it would if there were blowing in the region where the end wall static pressures were lower than the manifold, for low and medium suction levels.

#### **B. AVR**

The effect of suction on the AVR is shown in Figure 25. The were obtained over the three inch passage from 20 to 23 inches pitch. The base line AVR was computed to be equal to 1.071 and was reduced to 1.052 at the highest level of suction. The reduction of the AVR was expected, however, this value is not expected to be independent of pitch location due to non-uniformity in suction.

#### **C. INSTRUMENTED BLADE**

Figures 26 through 28 show the midspan pressure distribution on the instrumented blade with increasing suction. The pressures were non-dimensionalized with respect to the Prandtl dynamic pressure. It is seen that blade loading increased with increased suction. However, it was

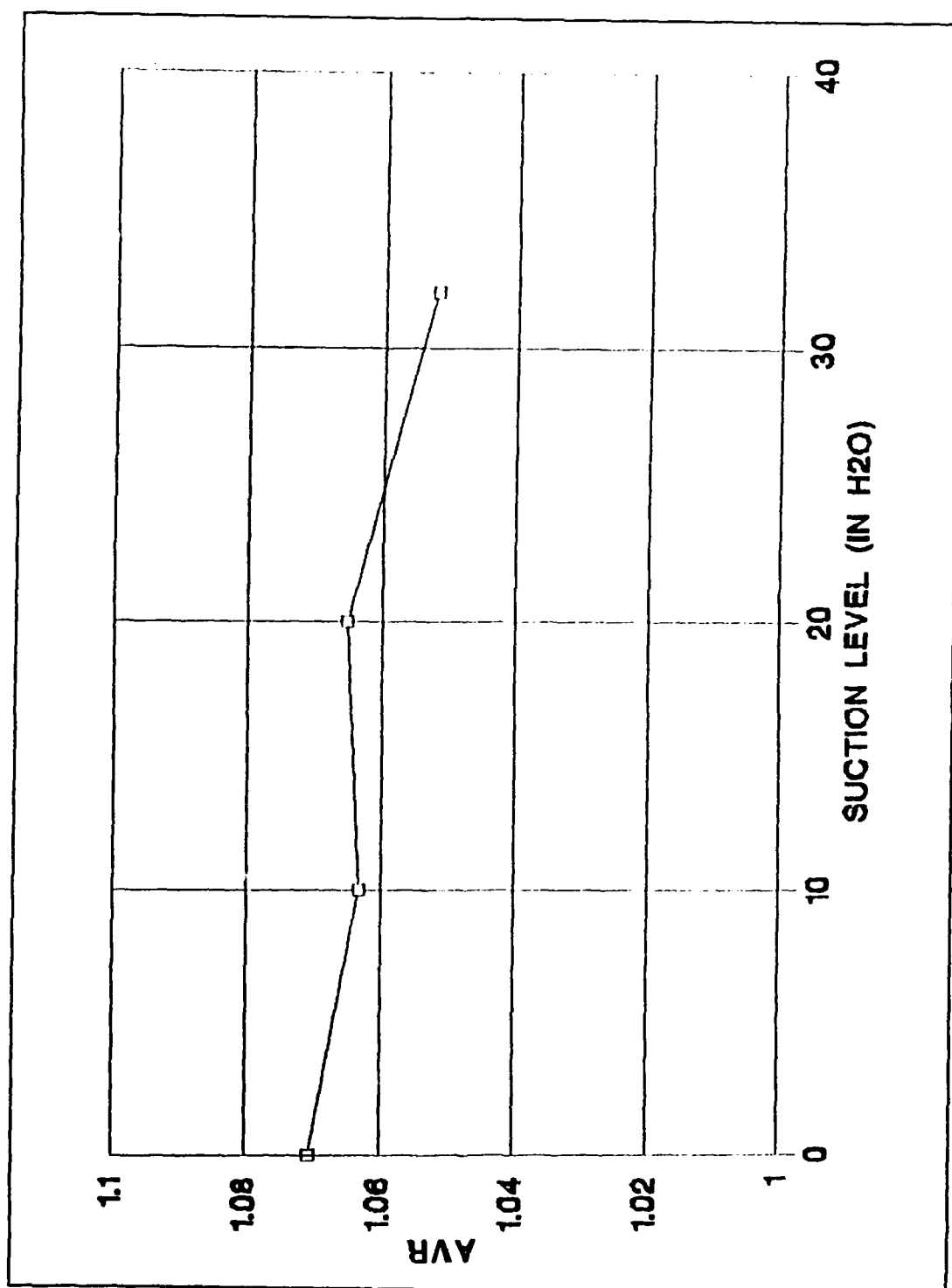
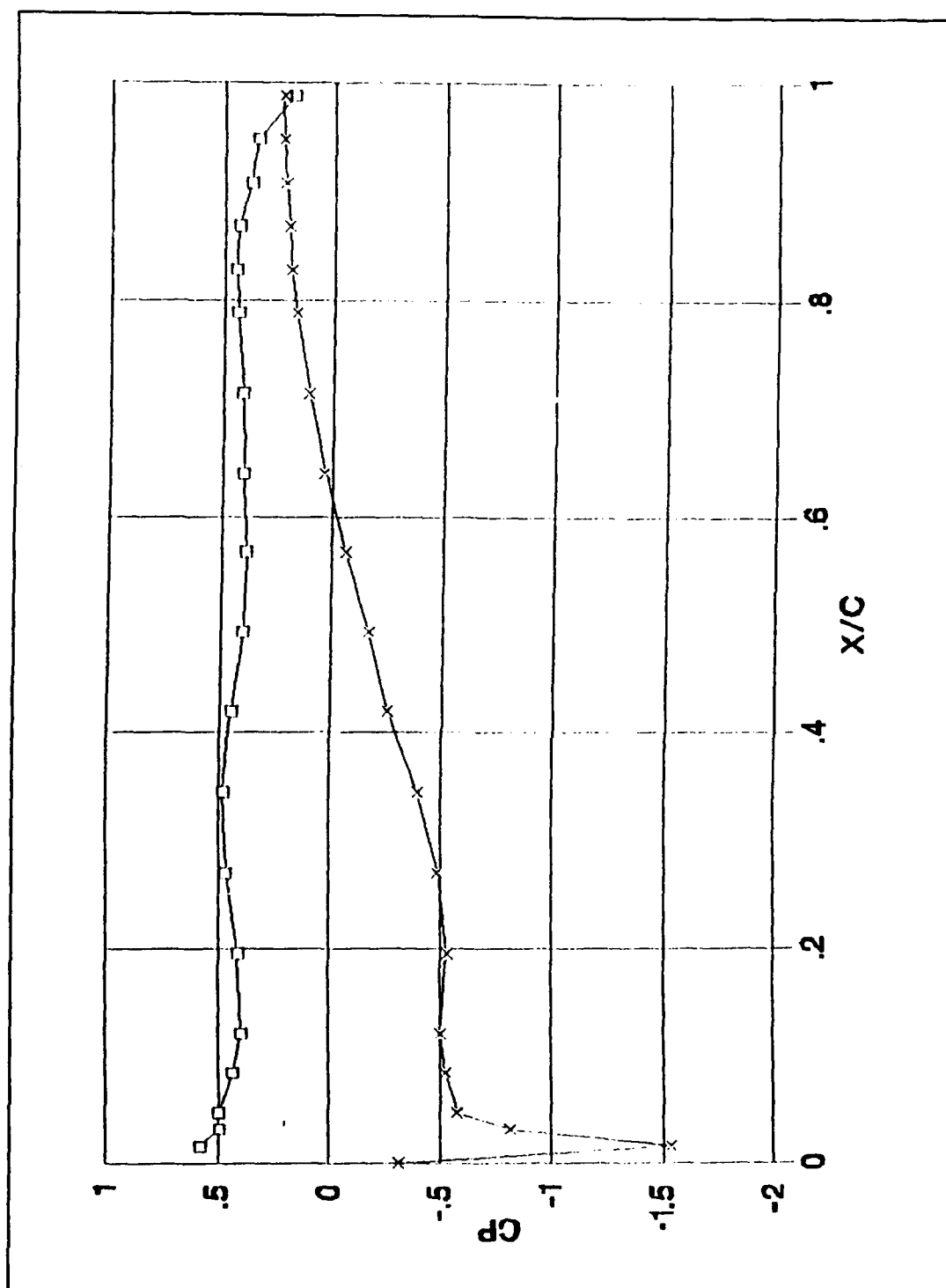


Figure 25. Suction Effects on the Axial Velocity Ratio for the Passage of 20 to 23 in. Pitch



**Figure 26.** Instrumented Blade Pressure Distribution for Low Suction

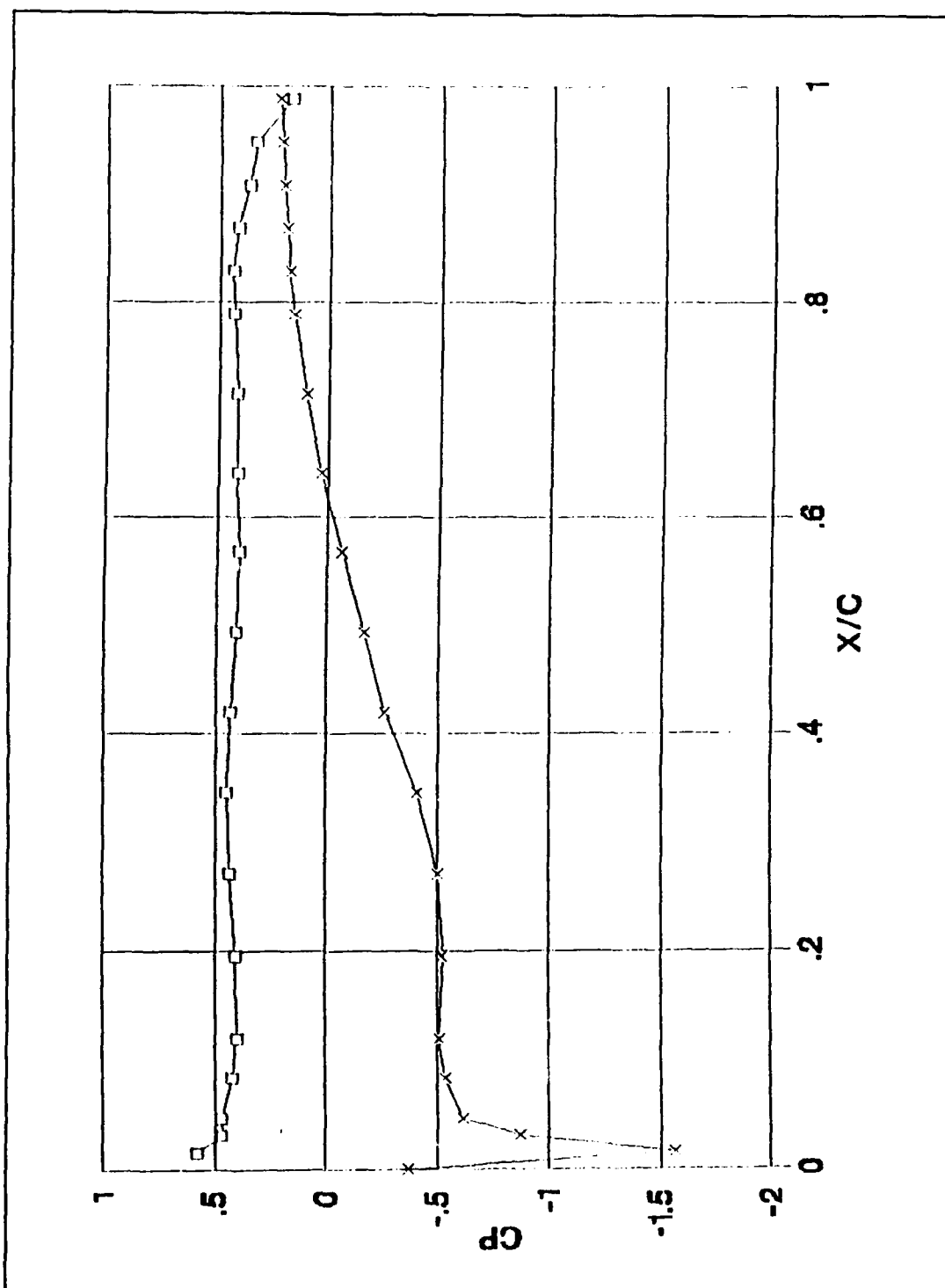
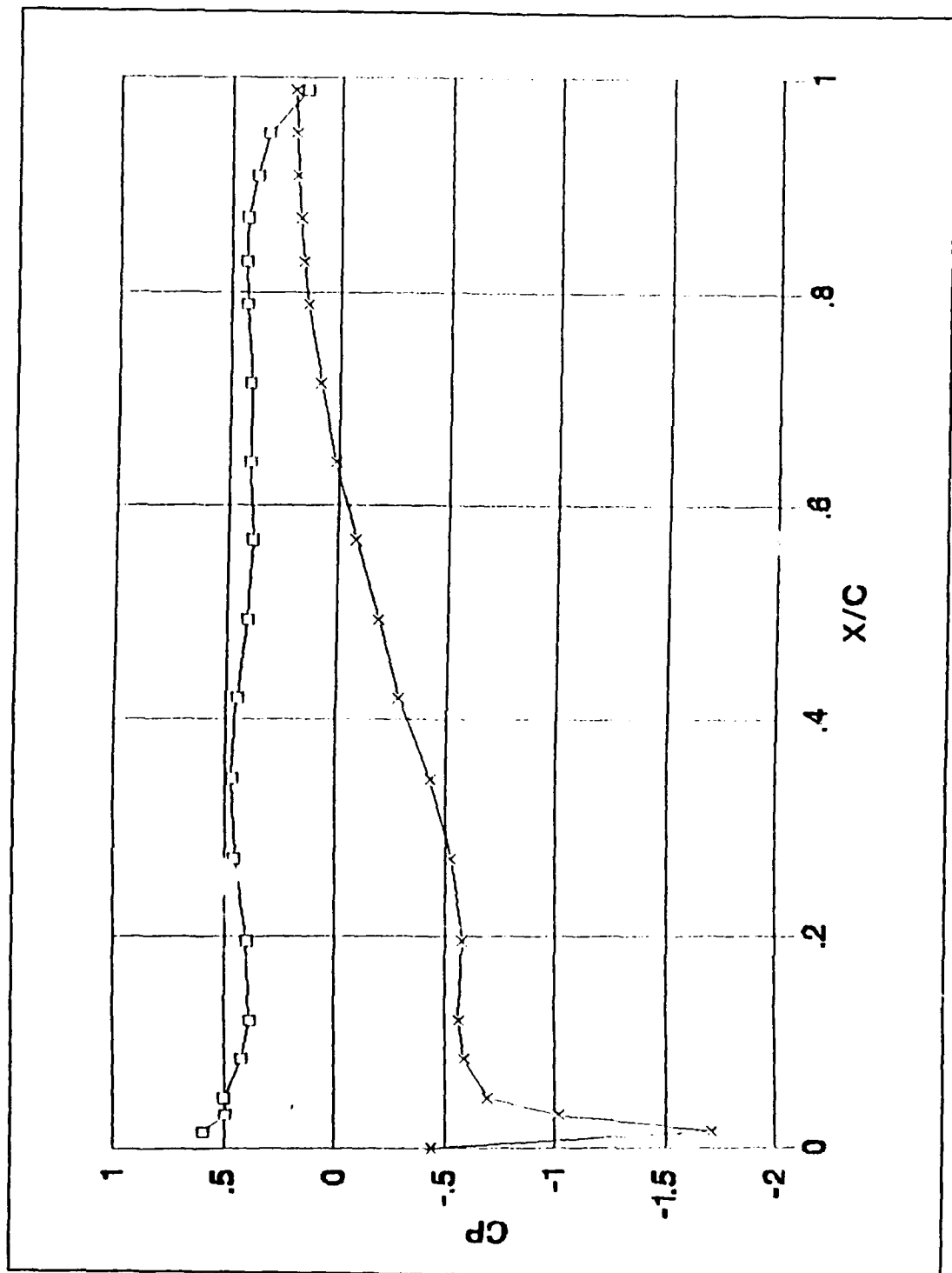


Figure 27. Instrumented Blade Pressure Distribution for Medium Suction



**Figure 28.** Instrumented Blade Pressure Distribution for High Suction

inconclusive whether boundary layer removal was responsible since the blade was located at mid-pitch. The yaw probe data for each suction level showed that the increased blade loading may be a result of increased inlet flow angle. In the region of the instrumented blade, the average flow angle measured for low suction was  $44.02^\circ$  and this increased to an average of  $44.59^\circ$  at high suction. It is important to note however, that the average inlet flow angle remained relatively constant across the pitch and all deviations were within the uncertainty in the yaw probe measurement.

#### **D. LDV SURVEYS**

Inlet and outlet flow measurements were made with the LDV system for comparison between base line and high suction. The inlet flow is seen in Figures 29 through 31. Figure 29 shows a slight reduction occurred in the inlet velocity and Figure 30 shows a half a degree reduction occurred in the inlet flow angle. This conflicts with the rake yaw probe measurements for the change in flow angle, therefore, the cause of the change in blade surface pressure distribution remains inconclusive. Outlet flow velocity profiles seen in Figure 31 show that the point of maximum deficit was shifted, thus the blade wake was displaced as a result of applying suction.



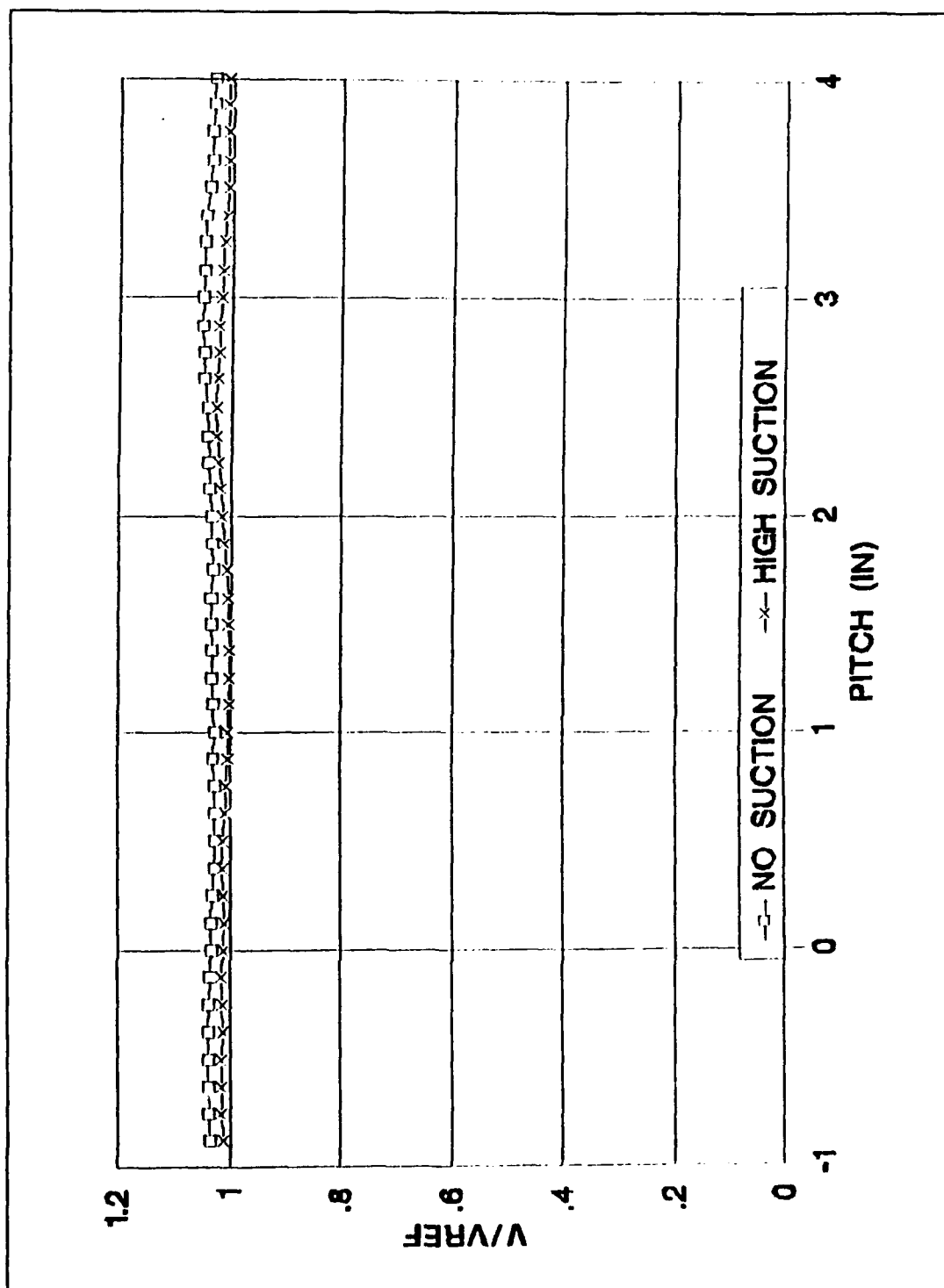
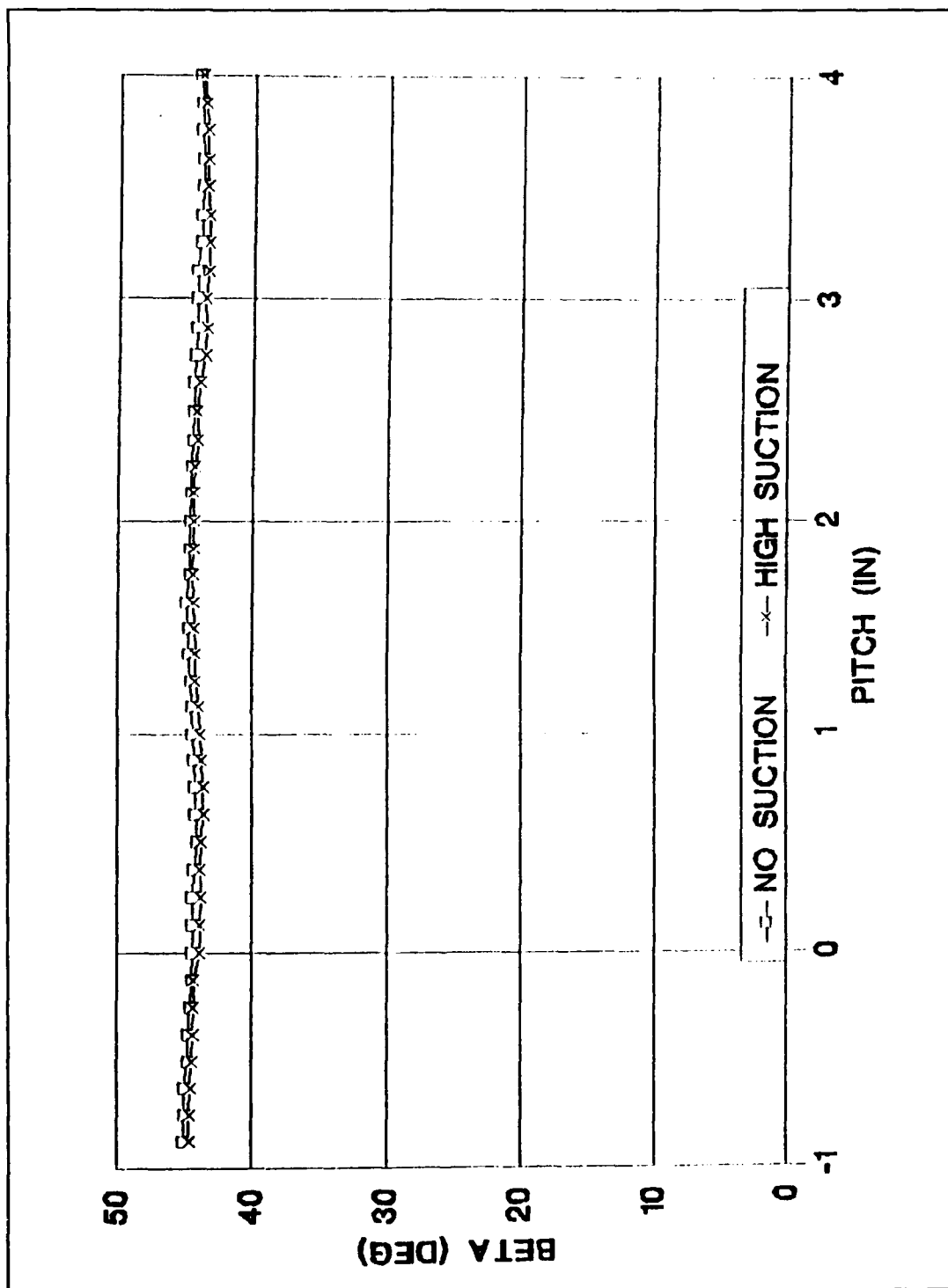
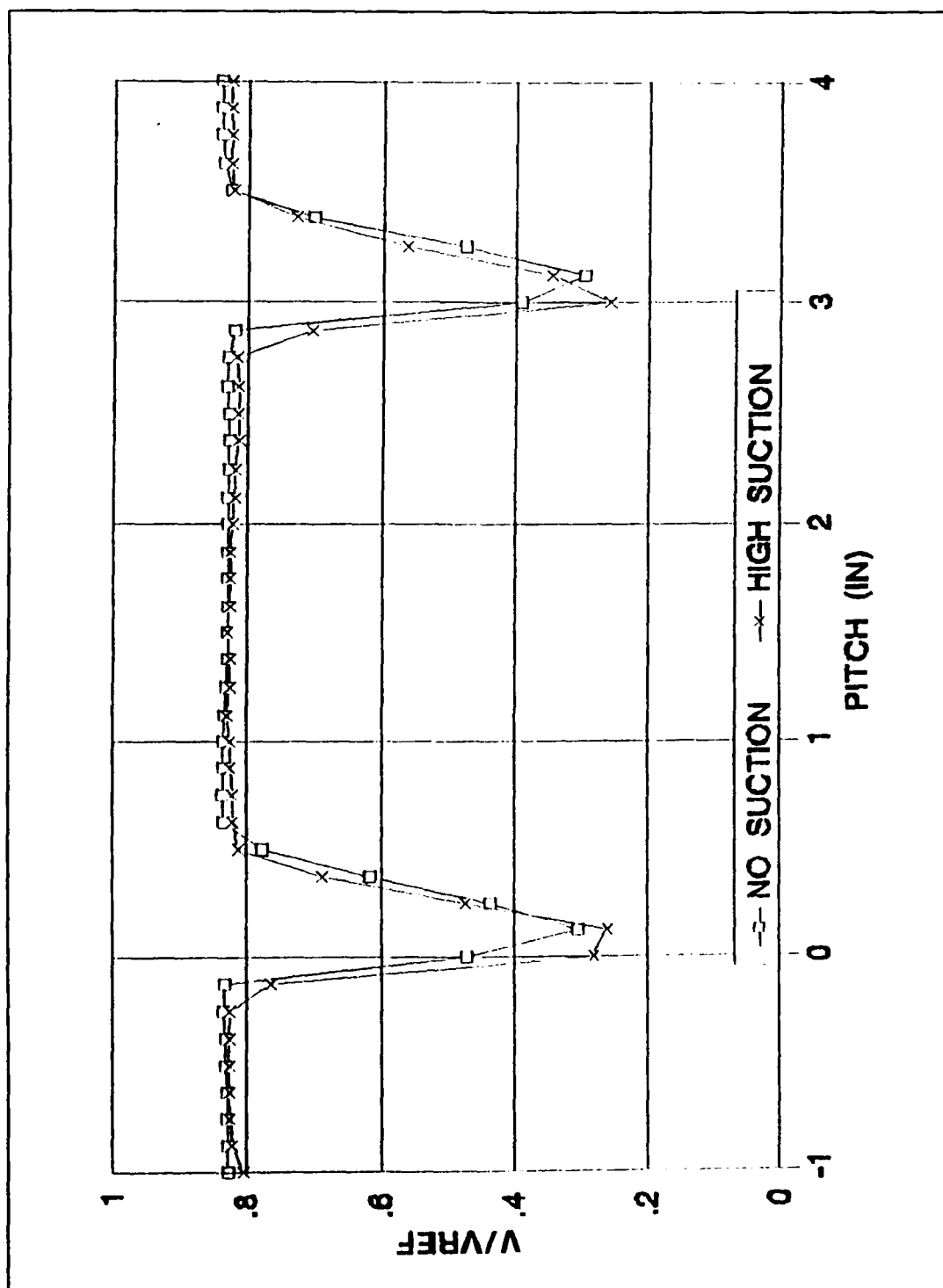


Figure 29. Station 1 Inlet Velocity Comparison for Base Line Configuration and High Suction



**Figure 30.** Station 1 Inlet Flow Angle Comparison for Base Line Configuration and High Suction



**Figure 31.** Station 18 Outlet Flow Velocity Comparison for Base line Configuration and High Suction

#### IV. CONCLUSIONS AND RECOMMENDATIONS

Boundary layer suction was performed 17.25 inches axially ahead of the test section in an effort to improve the two-dimensionality of the inlet flow. The conclusions resulting from these tests are as follows:

1. The base line configuration of the cascade now has a detailed set of data for the inlet flow boundary layers which can be used as the bench mark for future boundary layer removal tests.
2. The base line configuration has an inherent asymmetry in its endwall boundary layer thicknesses.
3. Carpet plots of rake probe surveys showed maximum suction at the east and west extremities of the test section and a minimum at mid-pitch.
4. The suction configuration and procedure used for these tests produced asymmetrical boundary layer removal in the spanwise direction.
5. There was a reduction in the displacement thickness on both endwalls for all levels of suction. However, the amount of displacement thickness reduction decreased with increased suction.
6. The AVR was reduced 1.9% across the passage of blades 7 and 8 from 1.071 to 1.052. For the flow to be considered acceptably two-dimensional, it would have had to be reduced to 1.03 or lower.
7. Endwall balanced manifold pressures did not yield symmetrical boundary layer suction and reduced south endwall suction did not increase the north endwall suction.

The present experiments were a first attempt at performing suction in the present cascade facility. Recommendations for the program to follow include:

1. Investigate the effect of varying the slot width.

2. Make provisions for the addition of another feed pipe to the suction plenum at the mid-pitch location of the manifold.
3. Section the manifold into separate chambers to enable better control of the pitchwise and spanwise suction levels.
4. Place an instrumented orifice at each gate valve to monitor the mass flow rate.
5. Perform tests on the blower to determine the optimum in its mass flow rate vs. pressure ratio characteristic.
6. Investigate the cause of the base line cascade's asymmetric endwall boundary layers.
7. Consider other methods for boundary layer removal, including the use of a porous plate with the present tunnel wall hardware, or, more extensively, rebuilding the tunnel walls to allow suction within the blade passages.

## **APPENDIX A. RECORDED DATA**

This Appendix contains the following:

1. All rake probe data.
2. Instrumented blade data.
3. Manometer data.
4. LDV data.

### **A1. RAKE PROBE DATA**

All rake probe data presented in this appendix was used for the computation of AVR, displacement thickness, spanwise pressure distribution and carpet plots. It is given in the order of the experimental procedure.

BASE LINE CONFIGURATION  
UP STREAM LOCATION

PITCH ) PROBE PORT	0 Cp	2.36	4.72	7.08	9.44	11.8	14.16
1	-.7808	-.0514	.1706	.2995	.4223	.1291	.1851
2	-.4715	-.0116	.2758	.3824	.5617	.2214	.3081
3	-.1837	.064	.4312	.5543	.7183	.3893	.4704
4	.4905	.1587	.5516	.803	.7738	.5304	.6244
5	.7262	.4899	.6617	.7303	.8214	.7908	.8364
6	.8556	.7777	.7818	.8357	.9203	.9142	.9396
7	.749	.8921	.9397	.9602	.9758	.979	.9651
8	.8977	.9214	.972	.9833	.9874	.9891	1
12	1	1	.9946	.9867	.9894	1	.9751
13	.7708	.978	1	.9974	.9998	.993	.9827
14	.4892	.947	.9962	1	1	.979	.9764
15	.6679	.9591	.9928	.994	.9953	.9724	.9697
16	.5538	.879	.9081	.9663	.9489	.9362	.9254
17	.2024	.6425	.5544	.8307	.7494	.8244	.7416
18	.0262	.428	.3012	.5899	.5265	.6744	.4767
19	-.0443	.2782	.1374	.3912	.3918	.4553	.3432
20	-.1924	.1686	.0615	.2619	.2396	.3315	.2087

PITCH ) PROBE PORT	16.52	18.88	21.24	23.6	26	17	19.36
1	.2276	.3383	.2714	.2231	.2152	.3107	.3847
2	.3558	.4611	.3889	.324	.3739	.4385	.5457
3	.5272	.6017	.5337	.5026	.4826	.6031	.6503
4	.6256	.6568	.6693	.6447	.6835	.675	.7334
5	.8036	.7651	.751	.7867	.8402	.7528	.7978
6	.8691	.8498	.8558	.8866	.9497	.8427	.9056
7	.8997	.9066	.959	.9928	.9978	.9174	.9614
8	1	1	.9842	.9959	1	.9921	.9853
12	.9851	.9592	.9818	.9669	.9863	1	.9808
13	.9824	.9692	.9813	.9702	.9941	.9966	.9839
14	.9758	.9766	.9998	.9862	.9965	.9916	1
15	.9605	.974	1	1	.9867	.9709	.9916
16	.8755	.8838	.9485	.9698	.9473	.8965	.9455
17	.7387	.6077	.8174	.8517	.8964	.6872	.6202
18	.619	.3472	.5588	.5346	.718	.5638	.3622
19	.4818	.1823	.3586	.3659	.454	.3837	.2423
20	.319	.077	.202	.2279	.3157	.2848	.17

CONTINUED							
BASE LINE CONFIGURATION RAKE DATA							
FULL TRAVERSEUP STREAM LOCATION							
PITCH )	21.72	24.08	26.44	28.8	31.16	33.52	35.88
PROBE							
PORT							
1	.1792	.1874	.291	.2957	.3093	.0758	.246
2	.3126	.3183	.3989	.4273	.4358	.2476	.3531
3	.4547	.4646	.5975	.5698	.5696	.4341	.5084
4	.6244	.6356	.6935	.7018	.6948	.661	.6792
5	.7599	.8184	.8757	.8431	.8042	.8081	.8686
6	.9014	.924	.9319	.8772	.8538	.8721	.9571
7	.9761	1	.9868	.9078	.9077	.9759	.9915
8	.9841	.9895	1	1	1	.9937	.9942
12	.9508	.9738	.9637	.9795	.9696	.9885	.9931
13	.9854	.9878	.9646	.9709	.9645	.9971	1
14	1	.9849	.9495	.9506	.9575	1	.9933
15	.9881	.9973	.915	.8964	.9354	.979	.9871
16	.9615	.9639	.8887	.8417	.8726	.9515	.9406
17	.8699	.8277	.8662	.6103	.711	.7604	.8304
18	.654	.6333	.7455	.398	.5655	.4952	.6519
19	.452	.4319	.5251	.2997	.365	.3127	.4637
20	.2912	.2813	.357	.167	.211	.2264	.2904
PITCH )	38.24	40.6	43	34.4	36.74	39.1	41.45
PROBE							
PORT							
1	.3627	.2666	.2161	.3336	.2473	.3959	.2175
2	.4811	.3686	.3638	.4632	.3553	.5435	.2983
3	.6522	.5736	.5257	.6099	.5098	.7311	.4698
4	.7146	.6624	.6614	.7012	.6897	.7885	.6954
5	.8567	.809	.7605	.8421	.8442	.8646	.8476
6	.9517	.8643	.8521	.9211	.8895	.9312	.9738
7	.9833	.9082	.95	.9378	.8964	.959	.9944
8	1	1	1	1	1	.9743	.991
12	.984	.9915	.9888	.982	.9791	1	1
13	.9836	.987	.9837	.9772	.9733	.9926	1
14	.9799	.9798	.9917	.9749	.9672	.9804	.999
15	.9539	.965	.9883	.9569	.9582	.9693	.9854
16	.8827	.91678	.946	.8948	.9113	.9088	.9366
17	.7556	.7944	.766	.7461	.7215	.7126	.8357
18	.5948	.5852	.5092	.4953	.5007	.4659	.5981
19	.4179	.421	.324	.3043	.3256	.292	.4183
20	.2341	.2824	.2077	.1896	.1985	.1829	.2675



CONTINUED

BASE LINE CONFIGURATION RAKE DATA

FULL TRAVERSE UP STREAM LOCATION

PITCH > PROBE PORT	43.81	46.18	48.54	50.9	53.3	55.61	57.98
1	.2597	.2525	.1948	.1603	.244	.2144	.0667
2	.4014	.348	.3143	.2694	.3696	.3573	.1482
3	.5492	.5165	.4697	.4571	.5444	.4395	.2162
4	.7077	.6835	.6996	.6493	.6996	.6891	.4277
5	.8646	.8301	.8334	.8232	.8892	.8485	.8188
6	.9641	.9073	.9059	.9299	.977	.9299	.9273
7	.9865	.9242	.95	.9955	.9988	.9547	.988
8	1	1	.9994	.9922	.9923	1	.9868
12	.9899	.9693	1	.9942	.9898	.9801	1
13	.9891	.9733	.9882	.9996	.9942	.9885	.9931
14	.9952	.9597	.9697	1	1	.9775	.9943
15	.9838	.9426	.9679	.9872	.9952	.9455	.9689
16	.9388	.8746	.9393	.9396	.9341	.8962	.873
17	.8286	.7017	.8172	.8125	.7464	.7617	.6863
18	.5833	.4981	.608	.6003	.5084	.5538	.4739
19	.4001	.3158	.4	.3965	.3512	.3667	.3377
20	.2567	.2045	.2782	.2321	.2119	.2207	.196

PITCH > PROBE PORT	57.98	60.38
1	.0667	-.72
2	.1482	-.4612
3	.2162	-.0524
4	.4277	.3904
5	.8188	.6025
6	.9273	.5396
7	.988	.5345
8	.9868	.9189
12	1	.8736
13	.9931	.806
14	.9943	.8088
15	.9689	.9614
16	.873	1
17	.6863	.6
18	.4739	.2896
19	.3377	.2078
20	.196	.126

BASE LINE CONFIGURATION RAKE DATA  
UP STREAM LOCATION 20 TO 23 INCHES

PITCH > PROBE PORT	20 Cp	20.25	20.5	20.75	21	21.25	21.5
1	.3835	.2685	.2171	.2379	.2858	.3487	.3559
2	.4183	.3076	.2875	.3237	.4205	.4512	.457
3	.5804	.4887	.4341	.4586	.5668	.5995	.5776
4	.7522	.6952	.6303	.6211	.6482	.677	.686
5	.9166	.903	.8441	.781	.7452	.7443	.7587
6	.9769	.96	.907	.8692	.8633	.8726	.8848
7	.9842	.9845	.9701	.9552	.9539	.9727	.9765
8	.9861	1	1	.9863	.9847	.9844	.991
12	.9801	.9786	.9581	.9475	.9777	.9953	.9728
13	1	.9836	.9655	.9668	.9836	.9889	.991
14	.9945	.9987	1	1	1	1	1
15	.9957	.9996	.9825	.981	.9946	.9917	.9869
16	.9688	.952	.9253	.9114	.9279	.9516	.9635
17	.6643	.701	.705	.6828	.7277	.7695	.882
18	.4289	.4772	.441	.3947	.4341	.5473	.6227
19	.3111	.2927	.261	.2024	.2146	.32	.385
20	.2046	.2041	.1486	.0962	.1266	.1702	.2787

PITCH > PROBE PORT	21.75	22	22.25	22.5	22.75	23
1	.3086	.2563	.2454	.2426	.3245	.3559
2	.384	.3085	.3186	.3504	.419	.4377
3	.5275	.4977	.4621	.4757	.5515	.5749
4	.6525	.6652	.6447	.6291	.6573	.6787
5	.8027	.8463	.8665	.8531	.8129	.7813
6	.925	.9527	.9142	.9241	.8967	.8594
7	.988	.9864	.9459	.9631	.9461	.9463
8	.9944	.9955	1	1	1	.9845
12	.9751	.9783	.9687	.9542	.9524	.9611
13	.9872	.9938	.9763	.9188	.9206	.9257
14	.9996	1	.987	.9735	.9919	1
15	1	.989	.9868	.9682	.9827	.9839
16	.9711	.9729	.9542	.9312	.9003	.8988
17	.8708	.8445	.8185	.7534	.7186	.7
18	.6497	.6603	.6358	.5635	.5152	.4645
19	.4346	.4208	.4061	.3873	.3323	.3065
20	.2659	.2664	.2706	.2257	.2086	.1721

BASE LINE CONFIGURATION RAKE DATA  
DOWN STREAM LOCATION 20 TO 23 INCHES

PITCH ) PROBE PORT	20 CP	20.25	20.5	20.75	21	21.25	21.5
1	.4517	.4278	.4504	.479	.518	.4991	.39
2	.5918	.5617	.5932	.6387	.6441	.6143	.4598
3	.7261	.7118	.716	.7373	.7327	.6747	.5099
4	.6498	.6825	.722	.7743	.7841	.7234	.5977
5	.5897	.5253	.6078	.6994	.8112	.8726	.8461
6	.6082	.5341	.5837	.683	.82	.9022	.9202
7	.5617	.563	.6353	.7658	.8407	.9243	.9552
8	.7085	.8569	.9488	.9961	1	1	1
12	1	1	1	1	.9735	.9765	.9659
13	.742	.8661	.9217	.9742	.9769	.9875	.974
14	.5602	.5278	.5714	.6562	.7611	.8842	.9605
15	.6936	.5722	.5562	.5645	.6508	.7861	.9315
16	.601	.5055	.5266	.5687	.6641	.7998	.9234
17	.586	.5801	.6374	.7105	.7944	.8359	.781
18	.7175	.7171	.7516	.7914	.8304	.835	.7311
19	.7466	.7551	.7929	.8311	.8537	.8283	.701
20	.6834	.7163	.7652	.8051	.8247	.7956	.7016

PITCH ) PROBE PORT	21.75	22	22.25	22.5	22.75	23
1	.289	.2882	.3951	.4808	.4503	.4339
2	.3539	.3663	.4627	.5544	.5808	.5256
3	.4164	.4658	.6005	.7268	.7132	.6691
4	.4716	.4817	.5431	.6113	.6447	.6121
5	.8072	.7398	.7023	.7217	.6357	.5066
6	.9296	.9225	.9294	.9551	.7912	.5885
7	.9466	.8933	.82	.7855	.6483	.5337
8	1	.9533	.838	.7871	.6576	.6183
12	.9633	.8478	.7158	.8076	1	1
13	.9916	.9221	.7842	.7755	.7734	.8448
14	.9988	.9555	.8707	.8081	.6335	.5284
15	.9922	1	1	1	.8214	.6016
16	.9685	.9694	.9285	.9248	.7635	.5615
17	.6821	.5733	.5979	.6491	.5889	.5351
18	.6152	.5322	.6194	.7075	.7407	.6854
19	.5902	.4865	.5893	.6923	.7535	.6951
20	.5738	.4376	.5184	.6479	.702	.6782

LOW SUCTION RAKE DATA  
FULL TRAVERSE UP STREAM LOCATION

PITCH ) PROBE PORT	0 Cp	2.36	4.72	7.08	9.44	11.8	14.16
1	-.9632	.3665	.5075	.4171	.5128	.3205	.3328
2	-.7206	.4172	.6626	.5621	.6475	.4896	.4729
3	-.4396	.61	.7382	.6987	.7441	.5624	.6122
4	.4427	.6431	.7334	.7689	.7862	.6863	.6775
5	.5687	.6794	.8199	.8337	.8803	.8577	.7897
6	.6648	.7663	.9472	.9656	.961	.966	.6975
7	.5778	.8817	.9831	.9629	.989	.9951	.94
8	.8907	.9447	.9815	.9909	1	1	1
12	1	1	.9936	.8851	.9926	.9943	.9725
13	.683	.9916	1	.9948	.9927	.9974	.968
14	.366	.9712	.9974	.9946	.9959	.998	.9708
15	.604	.9669	.9968	1	.998	.9919	.9471
16	.6541	.8844	.9089	.9771	.9581	.9006	.9119
17	.3967	.6311	.6522	.9312	.836	.8294	.873
18	.3588	.5114	.5029	.8366	.8007	.7385	.7695
19	.2817	.3573	.3669	.6696	.65	.6854	.5881
20	.1048	.2924	.255	.5266	.4968	.549	.4727
Ptot	5.268	10.226	9.97	10.064	10.206	10.164	10.692
Ppr	10.712	10.632	10.558	10.49	10.588	10.514	10.574
Ppl	12.25	12.174	12.146	12.022	12.146	12.042	12.116
Tpl	533.62	534.01	534.38	534.58	534.64	535.11	535.06
RAKE Q	7.834	13.56	13.118	13.118	13.296	13.328	13.63
BETA	44.3	43.11	44.1	43.66	43.62	43.77	43.76
PITCH ) PROBE PORT	16.52	18.88	21.24	23.6	26	26.44	28.8
1	.3344	.3863	.3364	.2382	.3081	.3447	.3381
2	.4697	.5532	.4584	.3741	.4319	.4684	.4631
3	.6156	.665	.5686	.5197	.557	.6456	.6089
4	.6752	.694	.664	.655	.7083	.7199	.6942
5	.8052	.7619	.751	.7928	.8551	.8473	.8
6	.8592	.8697	.8847	.9375	.9595	.9036	.8402
7	.8967	.9332	.9785	.9949	.994	.9204	.8889
8	1	.9788	.9774	1	1	1	1
12	.7958	1	.9843	.9825	.9957	.994	.9722
13	.9916	.9907	.9921	.9836	.9918	.9835	.9563
14	.9846	.9802	.9998	.9935	.9858	.9331	.9311
15	.9511	.9708	1	.9951	.9518	.8854	.9014
16	.88971	.9348	.9813	.9901	.9243	.864	.8741
17	.7771	.8096	.926	.9439	.8884	.7915	.6916
18	.6879	.6362	.7323	.7654	.837	.7883	.5119
19	.5904	.446	.5363	.6107	.6667	.6846	.3677
20	.4632	.2881	.368	.4862	.5179	.5259	.2689
Ptot	10.67	10.484	10.162	10.152	10.284	10.648	10.948
Ppr	10.552	10.514	10.47	10.478	10.508	10.518	10.448
Ppl	12.07	12.084	12.03	12.028	12.084	12.076	12.056
Tpl	535.14	535.03	535.15	535.7	535.72	535.51	535.67
RAKE Q	13.856	13.944	13.522	13.478	13.72	14.032	14.08
BETA	43.69	43.36	43.71	44.07	44.1	44.05	43.85

CONTINUED							
LOW SUCTION RAKE DATA							
FULL TRAVERSE UP STREAM LOCATION							
PITCH >	31.16	33.52	35.88	38.24	40.6	43	43.81
PROBE PORT							
1	.3382	.2334	.2066	.26	.2771	.2827	.2648
2	.4722	.3792	.3378	.4001	.3914	.4307	.4351
3	.6251	.4817	.4824	.5218	.5614	.5769	.5782
4	.721	.682	.6801	.671	.6881	.6825	.6995
5	.8087	.7968	.8718	.8469	.8187	.7506	.8367
6	.8739	.916	.9769	.9423	.8696	.847	.9677
7	.9598	.9854	.9983	.979	.953	.9707	.9991
8	.9847	.9983	1	1	.9978	.9957	1
12	.9992	.9913	.9939	.9829	.9891	.9951	.9881
13	.9959	.9918	.9992	.9885	1	1	.994
14	1	1	.995	.9718	.9928	.9968	.9898
15	.9884	.9996	.9878	.9536	.985	.9951	.9896
16	.9334	.9683	.9401	.8803	.9408	.9762	.9783
17	.7618	.7367	.8293	.7135	.8202	.8425	.9081
18	.5566	.4861	.6611	.50377	.6564	.665	.721
19	.445	.3398	.4719	.3693	.4399	.5177	.5479
20	.3728	.2772	.3977	.2715	.308	.4281	.429
Ptot	10.36	10.542	10.486	10.778	10.798	10.578	10.582
Ppr	10.404	10.548	10.478	10.44	10.464	10.36	10.362
Ppl	12.04	12.066	12.038	12.064	12.034	12.014	12.038
Tpl	535.73	535.68	535.67	535.74	536.02	536.03	536.21
RAKE Q	13.792	13.858	13.806	13.892	13.916	13.71	13.584
BETA	44.16	44.61	44.77	44.78	44.96	44.61	44.88
PITCH >	46.18	48.54	50.9	53.3	55.61	57.98	60.38
PROBE PORT							
1	.2967	.3473	.3344	.2809	.3811	.2693	-.0789
2	.4333	.5092	.4795	.4237	.5413	.4373	.1632
3	.5927	.6297	.5983	.5883	.691	.5192	.5
4	.7168	.7133	.729	.7475	.7605	.7461	.7969
5	.8501	.8144	.8236	.9069	.8578	.8771	.852
6	.9168	.8732	.9285	.9791	.911	.9503	.7443
7	.9405	.9109	.9907	.991	1	.991	.6579
8	1	.9871	.991	.9938	.9977	.9846	.9851
12	.9745	1	.9977	.9955	.9861	1	1
13	.9829	.9912	.992	1	.9932	.9923	.9997
14	.971	.97	1	.9929	.991	.9828	.9037
15	.9411	.9502	.9912	.9929	.9669	.9478	.9957
16	.8915	.9281	.9521	.9563	.9258	.7063	.8858
17	.7762	.7964	.857	.79	.8026	.3335	.6929
18	.6661	.5317	.6885	.6222	.6424	.1599	.5158
19	.5285	.4546	.5136	.4715	.4949	.1303	.3065
20	.3874	.3366	.3837	.3686	.3721	.0819	.1093
Ptot	10.966	10.884	10.486	10.63	10.648	10.916	6.46
Ppr	10.388	10.37	10.358	10.32	10.396	10.384	10.32
Ppl	12.05	12.052	12.064	12.048	12.07	12.094	12.076
Tpl	536.33	536.1	536.06	536.17	536.06	536.09	535.99
RAKE Q	13.718	13.968	13.508	13.54	13.384	13.642	9.824
BETA	44.49	44.34	44.71	45.07	45.2	45.52	47.08

LOW SUCTION SETTING RAKE DATA  
UP STREAM LOCATION 20 TO 23 INCHES

PITCH ) PROBE PORT	20 Cp	20.25	20.5	20.75	21	21.25	21.5
1	.2738	.2876	.3022	.3496	.378	.3437	.2989
2	.4036	.4325	.4481	.5005	.5215	.4459	.4052
3	.5485	.5709	.5823	.6275	.6425	.5822	.5597
4	.6843	.6618	.6571	.6734	.6822	.6704	.6686
5	.857	.854	.8202	.7951	.7821	.7759	.7942
6	.9465	.9265	.9006	.8923	.8914	.9007	.918
7	.9503	.9473	.9417	.9647	.9718	.968	.9695
8	.958	.9696	.9736	1	.9861	.9783	.9685
12	.9899	.973	.9551	.9783	.9978	.9897	.9754
13	.986	.9653	.9585	.9847	.9936	.9819	.9834
14	.9905	1	.9858	.9957	.994	1	1
15	1	.9979	1	.9884	1	.9885	.9978
16	.9691	.964	.9185	.9465	.9601	.9762	.9725
17	.7942	.817	.8058	.8638	.8924	.9317	.8877
18	.5874	.6177	.5932	.578	.6632	.7163	.8051
19	.4642	.4604	.4588	.4325	.4216	.5015	.591
20	.3759	.3937	.3676	.3546	.2983	.3525	.429
Ptot	10.534	10.668	10.702	10.326	10.07	10.066	10.098
Ppr	10.604	10.612	10.62	10.62	10.42	10.398	10.474
Ppl	12.096	12.15	12.08	12.054	11.98	11.936	11.882
Tpl	535.58	535.71	535.77	535.81	535.89	536.01	535.95
RAKE Q	13.942	13.844	13.71	13.614	13.538	13.438	13.348
BETA	43.32	43.55	43.56	43.35	43.14	43.28	43.26
PITCH ) PROBE PORT	21.75	22	22.25	22.5	22.75	23	
1	.2379	.2605	.3127	.3335	.3571	.2874	
2	.3697	.4041	.4498	.48	.4873	.4387	
3	.5188	.5565	.5826	.5948	.5862	.5752	
4	.6631	.6785	.6672	.6678	.6733	.6819	
5	.8465	.8565	.8424	.8232	.7933	.7847	
6	.9404	.9578	.9432	.9029	.893	.8768	
7	.9732	.9912	.9867	.9539	.9307	.9419	
8	.9814	1	1	.9841	.9824	.9875	
12	.984	.9851	.96	.9383	.9552	.9591	
13	.9855	.9825	.9694	.9545	.9663	.9496	
14	1	.9969	.9963	1	.9987	.9802	
15	.9969	.9974	.8858	.988	1	1	
16	.9691	.9674	.9367	.9262	.9296	.9273	
17	.8573	.8344	.7821	.7566	.7797	.8112	
18	.7892	.7852	.6977	.6304	.5822	.6273	
19	.6213	.6291	.5624	.4781	.4467	.4498	
20	.4972	.512	.4598	.358	.3238	.3397	
Ptot	10.23	10.196	10.392	10.542	10.45	10.404	
Ppr	10.44	10.548	10.494	10.596	10.47	10.5	
Ppl	12.03	12.002	12.006	12.002	11.984	12	
Tpl	536	535.91	535.96	536.08	536.01	535.96	
RAKE Q	13.57	13.558	13.474	13.398	13.45	13.434	
BETA	43.41	43.42	43.57	43.47	43.5	43.68	

LOW SUCTION SETTING RAKE DATA  
DOWN STREAM LOCATION 20 TO 23 INCHES

PITCH ) PROBE PORT	20 Cp	20.25	20.5	20.75	21	21.25	21.5
1	.4873	.4219	.4949	.5199	.5178	.4299	.3262
2	.6341	.5944	.6472	.6555	.5989	.5254	.3974
3	.7029	.6938	.7207	.7235	.6553	.5457	.4548
4	.6163	.6678	.7134	.7546	.7188	.5882	.5166
5	.5621	.5334	.6024	.7335	.818	.8324	.7934
6	.6251	.546	.5915	.7	.8275	.8887	.9073
7	.5108	.532	.6482	.7726	.9043	.9321	.9435
8	.6847	.8139	.9274	.9884	.967	.9981	.9906
12	1	1	1	.9939	.9878	.9994	.9873
13	.8605	.9447	.9939	1	1	1	.999
14	.5742	.597	.7083	.8327	.9464	.9968	1
15	.6524	.5824	.6193	.7235	.8933	.9732	.9992
16	.6307	.5537	.5937	.7197	.9027	.9551	.9778
17	.5899	.6131	.7169	.8291	.8929	.8909	.8512
18	.7273	.7375	.7998	.839	.8878	.8655	.7467
19	.736	.37612	.8049	.8362	.8705	.8525	.7485
20	.681	.7037	.7246	.7676	.8	.7848	.6989
Ptot	9.592	10.158	10.16	10.148	10.178	10.156	10.202
Ppr	10.582	10.486	10.482	10.576	10.574	10.514	10.508
Ppl	12.054	11.982	11.986	12.042	12.048	11.988	11.982
Tpl	534.88	535.14	535.23	535.34	535.35	535.58	535.67
RAKE Q	8.604	8.854	8.76	8.678	8.636	8.69	8.698
BETA	-2.75	-1.69	-1.94	-2.32	-3.58	-2.28	-2.84
PITCH ) PROBE PORT	21.75	22	22.25	22.5	22.75	23	
1	.2993	.3922	.4837	.5835	.5638	.5294	
2	.4117	.4841	.5984	.7281	.7018	.6556	
3	.4653	.5361	.6395	.7536	.7532	.688	
4	.4915	.4978	.5363	.6252	.6203	.6297	
5	.7471	.7414	.7616	.8327	.6894	.5856	
6	.8986	.9216	.9455	.9725	.7882	.6164	
7	.9428	.9165	.8786	.7865	.6274	.5248	
8	.9678	.9059	.8447	.7448	.6681	.7155	
12	.9739	.8731	.7582	.8779	1	1	
13	.9829	.9328	.8229	.8135	.8171	.8773	
14	.9943	.9802	.9315	.8522	.6583	.5737	
15	1	1	1	1	.8108	.663	
16	.9635	.9544	.9486	.9171	.7611	.5979	
17	.6819	.5862	.6204	.6567	.5924	.5766	
18	.628	.5426	.6395	.764	.7463	.7188	
19	.6862	.4912	.6029	.7476	.759	.7352	
20	.5477	.4342	.5455	.6794	.6947	.6632	
Ptot	10.178	9.622	8.91	8.01	8.68	9.744	
Ppr	10.49	10.478	10.504	10.586	10.514	10.486	
Ppl	11.978	11.988	11.998	12.028	11.984	11.996	
Tpl	535.51	535.51	535.69	535.59	535.66	535.593	
RAKE Q	8.766	7.862	5.962	6.236	7.842	8.724	
BETA	-2.86	-2.73	-2.3	-1.79	-2.15	-3.27	

MEDIUM SUCTION RAKE DATA  
FULL TRAVERSE UP STREAM LOCATION

PITCH ) PROBE PORT	0 Cp	2.36	4.72	7.08	9.44	11.8	14.16
1	-.8174	.4481	.4994	.475	.5371	.3217	.3349
2	-.6965	.5996	.6968	.6671	.6652	.1782	.4825
3	-.3477	.6499	.7409	.7474	.7806	.6245	.6111
4	.545	.6308	.7266	.7548	.7854	.6839	.6958
5	.5304	.727	.8772	.8746	.8787	.835	.8184
6	.5203	.83	.957	.9647	.9712	.9513	.8935
7	.5338	.9109	.9805	.9827	.9885	.9808	.931
8	1	.9608	.9813	.9889	.9845	.9856	1
12	.9771	1	.9897	.9851	.992	.9856	.9886
13	.659	.9893	.997	.9863	.9961	.9893	.9908
14	.2941	.9684	.9962	1	.998	.988	.9834
15	.4944	.9702	1	.9978	1	1	.9738
16	.7029	.9406	.9869	.9867	.9855	.9293	.8905
17	.5934	.7584	.7963	.8919	.9165	.7954	.8423
18	.6624	.6584	.7081	.8952	.8005	.7843	.8084
19	.6114	.4819	.6057	.8016	.6953	.6925	.6823
20	.4475	.4	.4389	.6302	.5339	.5013	.5139
Ptot	5.332	9.94	10.052	10.08	10.226	10.302	10.702
Ppr	10.474	10.348	10.584	10.584	10.544	10.594	10.588
Ppl	12.042	11.954	12.174	12.202	12.186	12.18	12.172
Tpl	527.59	527.58	527.63	527.69	527.79	527.71	527.71
RAKE Q	7.994	13.292	13.314	13.242	13.484	13.516	14.024
BETA	44.712	43.52	44.15	43.79	43.68	43.65	43.72
PITCH ) PROBE PORT	16.52	18.88	21.24	23.6	26	17	19.36
1	.3238	.3837	.3487	.2673	.3402	.3262	.395
2	.4941	.5532	.4289	.3671	.495	.4772	.5039
3	.6226	.676	.5579	.5343	.602	.5824	.6343
4	.6909	.711	.6136	.5565	.6792	.6821	.713
5	.7986	.7975	.7565	.8002	.8576	.8105	.8213
6	.8608	.88	.8695	.9364	.9584	.9163	.9262
7	.8959	.9434	.9212	.9946	.9957	.9495	.9513
8	1	.9838	.9361	1	1	.9866	.9601
12	.9905	.9992	.9654	.9775	.9986	.9941	.9724
13	.9949	.992	.9835	.9822	.9959	.9931	.9984
14	.9937	1	.9955	.9806	.9926	1	1
15	.9659	.9922	1	.982	.947	.9905	.9971
16	.9132	.9672	.9786	.9783	.9201	.9739	.9875
17	.8156	.8205	.9157	.9482	.9166	.8726	.7631
18	.7012	.7163	.794	.8212	.8628	.7446	.5873
19	.5948	.4792	.5221	.6259	.715	.581	.4961
20	.4485	.3842	.3874	.4891	.567	.4823	.3984
Ptot	10.488	10.242	10.3	10.312	10.336	10.134	10.216
Ppr	10.556	10.602	10.616	10.608	10.618	10.5	10.484
Ppl	12.138	12.168	12.15	12.194	12.174	12.11	12.098
Tpl	527.74	527.63	527.58	527.52	527.61	527.31	527.31
RAKE Q	13.956	13.89	13.628	13.754	13.942	13.67	13.562
BETA	43.46	42.94	43.44	43.76	44.03	43.29	43.5



CONTINUED  
MEDIUM SUCTION RAKE DATA  
FULL TRAVERSE UP STREAM LOCATION

PITCH > PROBE PORT	21.72	24.08	26.44	28.8	31.16	33.52	35.88
1	.2535	.2954	.3634	.3516	.3637	.3129	.2457
2	.373	.4381	.5119	.4929	.512	.4735	.3727
3	.5288	.5935	.6525	.6075	.6367	.5571	.5178
4	.636	.683	.729	.6943	.69	.6837	.7066
5	.8026	.8065	.8545	.8033	.7976	.7934	.8818
6	.9303	.9562	.9259	.8423	.8622	.9056	.975
7	.9735	.9815	.9442	.8829	.9393	.9805	.9943
8	.983	.9996	1	1	1	.9945	1
12	.981	.9817	.994	.9711	.9905	.9873	.9864
13	.9837	1	.9832	.9607	.9831	.9848	.9964
14	.9926	.9967	.9378	.9433	.9827	1	.9926
15	1	.9669	.9029	.9081	.9757	.9922	.9838
16	.977	.9496	.8854	.8556	.906	.964	.9435
17	.8656	.8452	.8106	.6815	.7361	.7803	.8291
18	.7839	.758	.83	.5196	.6004	.5269	.6705
19	.6389	.5985	.6843	.3683	.477	.3338	.5147
20	.4867	.4735	.5207	.287	.3567	.2527	.3837
Ptot	10.192	10.272	10.612	11.082	10.536	10.552	10.476
Ppr	10.45	10.396	10.49	10.558	10.592	10.524	10.468
Ppl	12.084	11.996	12.128	12.156	12.172	12.104	12.084
Tpl	527.39	527.3	527.69	527.61	527.59	527.5	527.51
RAKE Q	13.65	13.684	14.126	14.334	14	13.922	13.806
BETA	43.52	43.68	44	44.31	44.33	44.49	44.18
PITCH > PROBE PORT	38.24	40.6	43	46.18	48.54	50.9	53.3
1	.251	.2747	.3393	.3197	.378	.3785	.3759
2	.3796	.4282	.495	.4824	.5317	.5285	.5074
3	.5335	.5869	.6324	.6235	.6524	.6627	.6304
4	.6553	.7047	.6724	.7124	.721	.7363	.7475
5	.8344	.8352	.7742	.8533	.8041	.826	.916
6	.952	.8829	.8739	.9343	.8793	.918	.9736
7	.9964	.9475	.958	.9567	.9121	.9805	.9918
8	1	1	1	1	.9776	.9821	.996
12	.9889	.9954	.9966	.9753	1	1	.9956
13	.9872	.9985	.9917	.9757	.995	.9958	.9989
14	.9887	.9909	.9951	.9749	.983	.9942	1
15	.972	.9723	.9877	.9501	.96	.9824	.9929
16	.929	.9491	.9698	.8929	.9345	.9655	.9565
17	.7836	.8187	.8771	.7928	.8178	.8688	.8405
18	.5885	.6195	.7274	.7352	.6733	.7471	.7076
19	.4215	.4861	.5791	.5784	.5337	.5833	.5401
20	.3126	.3647	.4581	.4456	.4145	.4461	.4186
Ptot	10.59	10.776	10.61	10.772	10.804	10.366	10.472
Ppr	10.424	10.428	10.424	10.192	10.306	10.302	10.246
Ppl	12.078	12.018	11.964	11.942	11.94	11.926	11.932
Tpl	527.48	527.482	527.3	527.47	527.38	527.38	527.36
RAKE Q	13.824	14.016	13.782	13.55	13.854	13.378	13.334
BETA	44.55	44.35	44.68	43.99	44.48	44.68	45.15

CONTINUED			
MEDIUM SUCTION RAKE DATA			
FULL TRAVERSE UP STREAM LOCATION			
PITCH >	55.61	57.98	60.38
PROBE			
PORT			
1	.4204	.3137	.0535
2	.6107	.5007	.3025
3	.7157	.6058	.588
4	.7443	.7302	.768
5	.8796	.8984	.7691
6	.9559	.9662	.6286
7	1	.9936	.5847
8	.9943	.997	.9217
12	.9787	1	1
13	.9817	.9953	.988
14	.9836	.9922	.8799
15	.9629	.9682	.9769
16	.9111	.8104	.9845
17	.8285	.5026	.7776
18	.7628	.3657	.7338
19	.6041	.2332	.5289
20	.4558	.131	.3732
Ptot	10.614	10.704	6.844
Ppr	10.342	10.316	10.338
Ppl	11.916	11.874	11.862
Tpl	527.28	527.35	527.37
RAKE Q	13.126	13.42	10.166
BETA	45.37	45.44	47.58

MEDIUM SUCTION SETTING RAKE DATA  
UP STREAM LOCATION 20 TO 23 INCHES

PITCH ) PROBE PORT	20 Cp	20.25	20.5	20.75	21	21.25	21.5
1	.2928	.295	.3132	.3352	.354	.3137	.2855
2	.4335	.4469	.4653	.4811	.4863	.4442	.3992
3	.5696	.5743	.5984	.6075	.6186	.5733	.5404
4	.6654	.6742	.6586	.6664	.6751	.6651	.6682
5	.868	.8402	.8039	.7681	.7742	.7705	.8195
6	.9239	.9124	.899	.8754	.8889	.9102	.9284
7	.94	.9464	.9467	.9679	.966	.9669	.9644
8	.9484	.9647	.9782	.9802	.9734	.9733	.9676
12	.9678	.9524	.9614	.9826	.9874	.9793	.966
13	.9617	.9565	.9704	1	.9886	.9884	.9866
14	.9878	.9878	.9893	.9969	.9838	1	1
15	1	1	1	.9746	1	.9964	.9947
16	.6413	.9362	.9421	.6429	.9592	.9755	.9711
17	.8132	.8226	.8528	.8617	.8965	.9014	.8645
18	.6332	.6725	.6815	.6494	.6829	.7936	.8228
19	.4917	.4921	.5012	.4762	.4348	.5361	.6276
20	.3937	.3863	.422	.4273	.3113	.3747	.4724
Ptot	10.698	10.664	10.474	10.222	9.99	9.96	10.114
Ppr	10.678	10.658	10.564	10.532	10.542	10.404	10.604
Ppl	12.172	12.084	12.018	11.978	11.944	11.892	11.946
Tpl	536.38	536.5	536.44	536.48	536.51	536.52	536.48
RAKE Q	13.924	13.716	13.662	13.614	13.448	13.35	13.37
BETA	43.21	43.43	43.35	43.28	43.5	43.47	43.57
PITCH ) PROBE PORT	21.75	22	22.25	22.5	22.75	23	
1	.2744	.2943	.3128	.3324	.3368	.3237	
2	.4122	.458	.4619	.4742	.4835	.4187	
3	.5587	.5726	.5917	.5952	.5976	.5761	
4	.6563	.6737	.6683	.6562	.6725	.662	
5	.8465	.8609	.8357	.804	.787	.7735	
6	.9546	.9589	.923	.8972	.8676	.8718	
7	.9859	.9911	.9611	.9467	.9451	.9424	
8	.9882	1	.9823	.988	.974	.9676	
12	.9839	.973	.9451	.9327	.9383	.9432	
13	.9809	.9785	.9592	.9454	.9455	.9461	
14	1	.9996	1	1	1	1	
15	.9924	.9971	.9867	.9788	.9854	.9971	
16	.9687	.9581	.9096	.9208	.9181	.9392	
17	.8455	.8248	.7922	.7805	.799	.8494	
18	.8146	.7507	.6927	.604	.643	.675	
19	.6659	.6671	.5534	.4757	.4645	.5278	
20	.4974	.503	.431	.3473	.3828	.4029	
Ptot	10.044	10.31	10.492	10.662	10.528	10.358	
Ppr	10.412	10.576	10.598	10.682	10.626	10.586	
Ppl	11.924	12.178	12.09	12.1	12.094	12.046	
Tpl	536.58	536.75	536.84	536.8	536.72	536.77	
RAKE Q	13.462	13.632	13.514	13.542	13.472	13.342	
BETA	43.58	43.51	43.67	43.54	43.56	43.52	

MEDIUM SUCTION SETTING RAKE DATA  
DOWN STREAM LOCATION 20 TO 23 INCHES

PITCH ) PROBE PORT	20 Cp	20.25	20.5	20.75	21	21.25	21.5
1	.4835	.4694	.5009	.5211	.5308	.4398	.3253
2	.6248	.6129	.6533	.6294	.6496	.5124	.4013
3	.7044	.6901	.7219	.7183	.6613	.5549	.4593
4	.6223	.6585	.7233	.7589	.7327	.625	.5184
5	.5492	.5411	.6275	.7402	.8068	.8239	.7933
6	.6086	.555	.5974	.7074	.8162	.8763	.8976
7	.5045	.5347	.6366	.7694	.8591	.9316	.9396
8	.6787	.8369	.9525	.9822	1	1	.9841
12	1	1	1	.9986	.9908	.9883	.9811
13	.8755	.9462	.9982	1	.9953	.9923	.9911
14	.5695	.6196	.7508	.8382	.9461	.9988	1
15	.6481	.5913	.6358	.7319	.8843	.985	.9929
16	.6236	.5651	.6206	.7313	.8583	.9553	.9746
17	.5934	.6391	.7426	.8456	.9003	.9166	.8549
18	.7225	.752	.7668	.8474	.8806	.8784	.7921
19	.7329	.7623	.7757	.846	.8545	.8565	.7563
20	.6799	.7026	.7453	.7939	.7765	.8002	.7397
Ptot	9.822	10.08	10.154	10.102	10.132	10.118	10.158
Ppr	10.528	10.562	10.628	10.578	10.59	10.594	10.59
Ppl	12	11.976	12.044	12.038	12.04	12.086	12.016
Tpl	536.02	536.13	536.06	536.2	536.34	536.39	536.48
RAKE Q	8.67	8.834	8.0816	8.704	8.636	8.586	8.588
BETA	-3.56	-2.22	-1.29	-2.97	-3.14	-3.06	-2.58
PITCH ) PROBE PORT	21.75	22	22.25	22.5	22.75	23	
1	.3089	.39	.4948	.6233	.5712	.5336	
2	.4064	.4853	.6048	.6877	.6958	.606	
3	.477	.5372	.6392	.7658	.7451	.7063	
4	.4893	.492	.5413	.6402	.6141	.6134	
5	.7382	.7175	.7849	.8179	.6727	.54	
6	.9088	.9258	.9681	.9645	.7661	.6188	
7	.9385	.9074	.8699	.7822	.5941	.5262	
8	.9578	.9094	.7958	.7316	.6706	.6964	
12	.9541	.8714	.7585	.9073	1	1	
13	.9715	.9364	.8348	.8276	.8259	.8678	
14	.985	.9722	.9526	.8458	.6489	.5664	
15	1	1	1	1	.8033	.6555	
16	.95	.9338	.9381	.9369	.7499	.5988	
17	.664	.5977	.6114	.6729	.6042	.5844	
18	.6144	.5564	.6351	.7847	.7453	.7168	
19	.5656	.4933	.5949	.7689	.7471	.7372	
20	.5215	.4454	.5239	.7091	.6961	.6528	
Ptot	10.114	9.784	8.852	7.832	8.732	9.696	
Ppr	10.53	10.522	10.536	10.618	10.58	10.576	
Ppl	12.07	12.054	12.094	12.094	12.104	12	
Tpl	536.51	536.33	536.22	536.17	536.25	536.31	
RAKE Q	8.57	7.686	5.948	6.356	7.866	8.728	
BETA	-3.2	-2.83	-2.27	-1.06	-1.5	-2.29	

HIGH SUCTION RAKE DATA  
FULL TRAVERSE UP STREAM LOCATION

PITCH ) PROBE PORT	0 Cp	2.36	4.72	7.08	9.44	11.8	14.16
1	-1.1581	.4405	.5266	.4818	.5068	.3306	.3433
2	-.8766	.5699	.7215	.6599	.6976	.5096	.5175
3	-.5225	.6514	.7788	.7761	.7466	.6592	.6438
4	.71186	.6108	.774	.805	.8188	.743	.7161
5	.6177	.7165	.9262	.9297	.9012	.8888	.8027
6	.5328	.8443	.9633	.9844	.9546	.9502	.8563
7	.5333	.8947	.9706	.9925	.981	.9823	.9098
8	.9784	.9586	.9797	.9981	1	.999	.9824
12	1	.9886	.9925	.9971	.978	1	.999
13	.6998	1	.9963	.9942	.9744	.988	1
14	.2519	.9521	.9932	.9988	.9874	.998	.9853
15	.5338	.9669	1	1	.9866	.9778	.9657
16	.8007	.9574	.9934	.9858	.9394	.9524	.9139
17	.9306	.8254	.9492	.9492	.88	.8463	.8691
18	.9883	.7301	.8449	.9232	.8194	.7948	.826
19	.9292	.6118	.7195	.8177	.7502	.6	.6852
20	.6656	.507	.5992	.6816	.5568	.5051	.5591
Ptot	4.264	9.862	9.646	9.61	10	9.836	10.482
Ppr	10.408	10.294	10.306	10.216	10.168	10.2	10.48
Ppl	11.994	11.888	11.822	11.75	11.744	11.744	12.046
Tpl	532.04	532.13	532.21	532.33	532.39	532.62	533.28
RAKE Q	6.98	13.134	12.862	12.822	13.056	13.13	13.92
BETA	44.87	43.51	44.19	43.88	43.71	43.79	43.44
PITCH ) PROBE PORT	16.52	18.88	21.24	23.6	26	17	19.36
1	.3454	.3859	.3276	.279	.3623	.3178	.3447
2	.506	.5626	.4452	.4275	.5015	.4433	.517
3	.6197	.6764	.5851	.5822	.6057	.5887	.6457
4	.6705	.7359	.6781	.6673	.6859	.6776	.7315
5	.7964	.8217	.7391	.8231	.8507	.8184	.8669
6	.848	.8915	.9293	.9511	.9482	.9246	.9394
7	.8979	.9448	.9843	.9965	.9905	.96	.9533
8	.9674	.9688	.9958	.9967	1	.9718	.9619
12	.957	.9821	.9935	.9799	.9932	.9809	.9708
13	.9835	.983	.996	.9955	.9893	.9944	.9969
14	1	1	1	1	.9818	.9964	1
15	.9716	.9943	1	.9965	.9635	1	.9939
16	.9282	.97911	.9907	.9896	.9282	.9881	.98722
17	.8442	.8911	.9458	.9578	.8631	.8731	.7751
18	.7139	.748	.8488	.8294	.8851	.7739	.5788
19	.6076	.5111	.5687	.7059	.7544	.6641	.5246
20	.4694	.35	.3808	.5926	.5649	.5072	.4237
Ptot	10.422	10.142	9.896	10.125	10.302	10.056	10.19
Ppr	10.408	10.46	10.418	10.396	10.504	10.392	10.416
Ppl	12.124	12.162	12.04	12.064	12.12	12.012	12.09
Tpl	533.17	533004	533.51	533.63	533.6	534.25	534.13
RAKE Q	13.534	13.638	13.506	13.628	13.888	13.486	13.574
BETA	43.37	43.52	43.99	44.27	44.36	43.66	44.05

HIGH SUCTION RAKE DATA							
FULL TRAVERSE UP STREAM LOCATION							
PITCH ) PROBE PORT	21.72	24.08	26.44	28.8	31.16	33.52	35.88
1	.2525	.3007	.3592	.3681	.3754	.2851	.2525
2	.4083	.4582	.5357	.517	.5061	.3916	.3937
3	.5463	.5785	.6342	.6299	.6464	.5256	.5289
4	.648	.682	.7182	.7025	.67746	.6925	.696
5	.8287	.8306	.8369	.8082	.7902	.8029	.8579
6	.9461	.9233	.8985	.8456	.8247	.9211	.9761
7	.9763	.9831	.9475	.9049	.9409	.9776	.9975
8	.9873	.9908	1	1	1	.9876	1
12	.9786	.9716	.9938	.9925	.9846	.9833	.9872
13	.9816	.986	.9849	.9777	.9882	.9882	.9937
14	.9904	1	.9515	.9562	.9919	.9935	.9918
15	1	.9755	.8875	.9251	.9824	1	.9751
16	.953	.9254	.8496	.904	.9353	.9614	.9498
17	.845	.8809	.7727	.7457	.7784	.7803	.8289
18	.775	.7858	.7356	.6115	.6084	.5592	.7534
19	.6486	.5932	.6767	.4293	.4926	.3944	.5379
20	.4912	.5038	.5317	.3299	.4059	.2908	.414
Ptot	10.204	10.428	10.596	10.688	10.362	10.522	10.44
Ppr	10.426	10.446	10.416	10.402	10.392	10.402	10.392
Ppl	13.132	12.118	12.096	12.11	11.996	12.1	12.052
Tpl	533.92	533.84	533.8	533.75	533.95	533.8	533.82
RAKE Q	13.658	13.808	14.146	14.208	13.8	13.89	13.814
BETA	44.01	44.25	44.14	44.6	44.58	44.73	44.8
PITCH ) PROBE PORT	38.24	40.6	43	34.4	36.74	39.1	41.45
1	.2899	.2774	.326	.317	.3087	.4429	.3724
2	.4066	.4397	.4941	.4657	.4728	.6101	.56
3	.5539	.6044	.6313	.6124	.6167	.7282	.6733
4	.6665	.6905	.6875	.6553	.703	.7358	.6922
5	.8568	.8261	.7703	.8455	.8485	.8259	.8159
6	.9704	.8858	.8557	.9266	.8864	.9187	.9452
7	.9917	.9321	.9579	.9428	.892	.9526	.986
8	1	1	1	1	1	1	.9912
12	.9943	.9856	.9849	.9856	.9913	.9947	.9988
13	.9964	.9937	.9847	.9766	.9766	.9903	.9956
14	.9968	.9878	.9957	.874	.9534	.976	1
15	.976	.9736	.9853	.9625	.9358	.9564	.9937
16	.9253	.9456	.9664	.9195	.951	.8966	.9466
17	.7635	.8565	.8942	.7575	.535	.739	.8941
18	.5851	.7178	.7769	.5457	.3704	.5444	.7902
19	.4178	.5823	.646	.4067	.4104	.3888	.623
20	.3352	.4286	.5005	.3245	.3462	.2954	.4862
Ptot	10.604	10.844	10.606	10.706	10.774	10.5	10.406
Ppr	10.364	10.378	10.45	10.49	10.42	10.372	10.384
Ppl	12.04	12.042	12.086	11.92	11.974	11.984	11.974
Tpl	533.82	533.65	533.61	534.03	534	533.89	533.71
RAKE Q	13.978	14.022	13.705	14.092	14.088	13.786	13.686
BETA	44.8	44.57	45.04	43.72	43.94	44.23	44.22

CONTINUED HIGH SUCTION RAKE DATA FULL TRAVERSE UP STREAM LOCATION							
PITCH )	43.81	46.18	48.54	50.9	53.3	55.61	57.98
PROBE							
PORT							
1	.2954	.3209	.3688	.3897	.3843	.431	.3644
2	.4645	.4972	.543	.5638	.5676	.6132	.5239
3	.6163	.6417	.6495	.6567	.7064	.6775	.6423
4	.7061	.7267	.7122	.7442	.7999	.7591	.7682
5	.8309	.8591	.8284	.8192	.9081	.8887	.9027
6	.9615	.9282	.8853	.9138	.9757	.9289	.9668
7	.9973	.9656	.9169	.9761	.9916	.9937	1
8	.9973	1	.9852	.9846	.999	1	.9897
12	.9898	.9706	1	.9998	.9946	.9939	.9908
13	.9935	.9805	.9968	1	.9979	.9966	.9836
14	1	.9727	.9841	.9952	1	.999	.9788
15	1	.9597	.9605	.9879	.991	.9872	.973
16	.9827	.9039	.9135	.9591	.9718	.9405	.8491
17	.9244	.8128	.8372	.8944	.8562	.9001	.6288
18	.8452	.7745	.7275	.793	.7499	.8109	.513
19	.6749	.6579	.5276	.6588	.6157	.6929	.4484
20	.5272	.4835	.4324	.4865	.487	.5366	.3119
Ptot	10.398	10.756	10.688	10.376	10.444	10.492	10.83
Ppr	10.346	10.384	10.434	10.336	10.436	10.442	10.348
Ppl	11.986	11.926	11.934	11.954	12.004	11.94	11.916
Tpl	533.76	534	533.97	534.01	534.03	534.01	534.05
RAKE Q	13.468	13.498	13.778	13.408	13.262	13.26	13.442
BETA	44.18	43.9	43.88	43.91	44.79	44.85	45.36
PITCH )	60.38						
PROBE							
PORT							
1	.1753						
2	.4108						
3	.7072						
4	.8256						
5	.774						
6	.5844						
7	.5537						
8	.9141						
12	.8904						
13	.947						
14	.8053						
15	.854						
16	1						
17	.9141						
18	.8335						
19	.6892						
20	.5013						
Ptot	7.098						
Ppr	10.32						
Ppl	11.884						
Tpl	534.29						
RAKE Q	9.598						
BETA	47.03						

HIGH SUCTION SETTING RAKE DATA  
UP STREAM LOCATION 20 TO 23 INCHES

PITCH ) PROBE PORT	20 Cp	20.25	20.5	20.75	21	21.25	21.5
1	.2931	.2939	.3125	.3291	.3331	.3396	.2784
2	.4379	.4503	.4802	.4825	.4864	.3979	.3993
3	.5908	.6049	.6017	.6312	.626	.6008	.5755
4	.6858	.6572	.668	.6798	.6711	.6763	.6843
5	.6623	.8635	.812	.7879	.7744	.7844	.8119
6	.9219	.9151	.892	.9045	.9056	.9131	.9407
7	.9455	.9843	.9752	.9753	.9686	.9672	.9707
8	.9534	.9974	.9868	.9952	.9741	.9714	.9738
12	.9639	.9813	.9799	.997	.9957	.9807	.9715
13	.9589	.9766	.9789	.991	1	.9777	.978
14	.9862	.9899	1	1	.9974	.998	.9954
15	1	1	.9846	.9932	.9929	1	1
16	.958	.9805	.9612	.9601	.9804	.9871	.9782
17	.8124	.9066	.9122	.8931	.9131	.9248	.8819
18	.6392	.7554	.7449	.6876	.7019	.8385	.8371
19	.5166	.5684	.5601	.5199	.4581	.5799	.6629
20	.4236	.4579	.481	.485	.3222	.3953	.5069
Ptot	10.426	10.064	10.162	9.968	9.876	9.946	10.092
Ppr	10.406	10.404	10.428	10.52	10.436	10.384	10.422
Ppl	11.922	11.89	11.936	11.918	11.938	11.918	11.966
Tpl	539.51	538.55	538.69	538.71	538.88	538.59	538.46
RAKE Q	13.646	13.488	13.552	13.584	13.496	13.424	13.464
BETA	43.24	43.67	43.32	43.15	43.23	43.28	43.147

PITCH ) PROBE PORT	21.75	22	22.25	22.5	22.75	23
1	.2833	.307	.3233	.3207	.3605	.3335
2	.4196	.4595	.4761	.4715	.4915	.4643
3	.576	.6018	.6053	.615	.6279	.604
4	.6831	.6934	.692	.6697	.702	.6987
5	.8677	.8671	.8401	.7981	.8117	.7872
6	.9688	.9977	.9304	.8932	.9118	.8947
7	.9902	.9871	.9745	.9414	.9458	.9728
8	.9936	1	.9949	.9793	.9709	.9808
12	.9926	.9709	.9502	.9256	.9815	.9664
13	.9884	.9765	.9628	.9499	.9799	.9694
14	.9936	.9968	1	1	.9978	.9964
15	1	.9913	.9856	.9935	1	1
16	.9814	.9575	.9481	.9196	.9751	.9602
17	.8957	.832	.816	.7926	.8768	.8981
18	.805	.7557	.6949	.6565	.7187	.7451
19	.6448	.6386	.5883	.5021	.5397	.592
20	.5063	.5092	.43	.3754	.4047	.4547
Ptot	9.99	10.11	10.28	10.452	9.954	10.01
Ppr	10.446	10.392	10.418	10.398	10.37	10.47
Ppl	11.954	11.916	11.906	11.952	11.912	11.98
Tpl	538.67	538.59	538.85	538.86	538.53	538.25
RAKE Q	13.564	13.46	13.4	13.316	13.408	13.344
BETA	43.2	43.41	43.5	43.55	43.43	43.2



HIGH SUCTION SETTING RAKE DATA  
DOWN STREAM LOCATION 20 TO 23 INCHES

PITCH > PROBE PORT	20 Cp	20.25	20.5	20.75	21	21.25	21.5
1	.5054	.4918	.5063	.5385	.5271	.4387	.321
2	.6248	.6199	.6493	.6639	.6357	.4886	.3963
3	.7111	.7047	.7239	.7258	.6774	.5132	.4492
4	.6206	.6679	.7264	.7622	.7303	.6383	.5315
5	.5485	.5428	.6255	.7284	.7163	.8226	.7885
6	.6114	.5531	.5948	.7015	.8092	.8643	.8971
7	.5044	.5319	.6275	.7713	.8682	.9234	.9298
8	.6673	.8037	.937	.9795	.9893	.9918	.9734
12	1	1	.996	.9968	.9968	.9956	.9786
13	.8806	.9522	1	1	1	.9908	.9841
14	.5717	.6082	.7133	.8348	.9567	1	1
15	.6588	.5952	.6407	.7596	.8931	.9806	.9943
16	.631	.556	.616	.7489	.8768	.9687	.9824
17	.5915	.6357	.7468	.836	.8945	.9345	.8551
18	.7293	.7096	.8018	.8456	.8903	.9052	.7704
19	.7448	.7163	.8018	.8362	.8682	.8834	.7556
20	.6694	.6863	.7386	.7584	.8138	.7884	.7175
Ptot	9.414	9.882	9.97	9.95	9.938	9.98	10.088
Ppr	10.534	10.396	10.49	10.54	10.446	10.438	10.534
Ppl	11.964	11.874	11.942	11.962	11.93	11.93	11.984
Tpl	536.32	536.56	536.66	536.74	536.8	536.77	536.7
RAKE Q	8.486	8.672	8.568	8.542	8.528	8.548	8.524
BETA	-2.829	-2.876	-1.548	-2.21	-1.91	-1.73	-2.7
PITCH > PROBE PORT	21.75	22	22.25	22.5	22.75	23	
1	.3196	.3834	.505	.6083	.5998	.5423	
2	.3996	.475	.5976	.727	.7256	.6585	
3	.4772	.528	.631	.7582	.7257	.7128	
4	.4867	.4851	.5284	.659	.6293	.6111	
5	.7553	.7058	.7784	.8249	.702	.5553	
6	.9083	.9241	.9584	.9938	.8035	.6134	
7	.947	.9066	.8642	.7865	.6215	.527	
8	.9645	.9033	.8056	.7373	.6822	.6913	
12	.9706	.8633	.7566	.8674	1	1	
13	.9853	.9286	.8068	.8133	.824	.8868	
14	.9938	.9697	.925	.8594	.6653	.5758	
15	1	1	1	1	.8337	.655	
16	.9692	.939	.9371	.9804	.7823	.6008	
17	.6919	.6	.6133	.6673	.6182	.5911	
18	.6237	.5402	.6267	.7649	.7168	.7277	
19	.5078	.5098	.5779	.7507	.722	.7362	
20	.553	.4434	.5225	.6832	.6944	.6609	
Ptot	9.924	9.64	8.808	7.766	8.39	9.674	
Ppr	10.5	10.444	10.458	10.474	10.554	10.598	
Ppl	11.978	11.936	11.948	11.97	12.032	12.1	
Tpl	536.71	536.95	536.92	536.94	536.96	536.77	
RAKE Q	8.53	7.478	5.916	6.008	7.59	8.72	
BETA	-3.03	-3.06	-2.1	-1.82	-2.48	-3.52	

SPANWISE BALANCED MANIFOLD PRESSURES  
UP STREAM LOCATION 20 TO 23 INCHES

PITCH ) PROBE PORT	20 Cp	20.25	20.5	20.75	21	21.25	21.5
1	.3606	.3446	.3116	.3265	.3306	.1727	.348
2	.5195	.4627	.4772	.4874	.4967	.4196	.5046
3	.6273	.5884	.6095	.6094	.6283	.5859	.6211
4	.7302	.7067	.7029	.6968	.6929	.683	.6943
5	.858	.8368	.8237	.8358	.8076	.8263	.8562
6	.9114	.8967	.89	.8918	.897	.9095	.9305
7	.9383	.9315	.9363	.9418	.9455	.9452	.9455
8	.9616	.9806	.9967	.998	.9779	.9646	.9551
12	.9783	.9648	.9774	.9968	.9877	.9774	.9661
13	.9754	.9544	.9863	.9876	.9936	.983	.9786
14	1	1	1	.9876	.9922	1	.998
15	.9965	.9949	.9971	1	1	.996	1
16	.9489	.9431	.9357	.9514	.9658	.9766	.9635
17	.7753	.8283	.8531	.8987	.8899	.9037	.8805
18	.551	.6405	.6824	.6309	.6375	.7838	.7879
19	.4724	.4797	.5041	.4771	.4019	.5891	.5703
20	.3619	.3644	.3811	.3772	.298	.4218	.3747
Ptot	10.406	10.516	10.36	10.132	10.062	10.1	10.194
Ppr	10.56	10.62	10.56	10.534	10.576	10.59	10.57
Ppl	12.11	12.084	12.096	12.056	12.052	12.066	12.034
Tpl	540.39	541.46	541.58	541.84	542.16	542.28	542.95
RAKE α	13.764	13.766	13.744	13.76	13.598	13.544	13.502
BETA	43.89	44.052	44.17	43.85	44.08	44.12	44.16
PITCH ) PROBE PORT	21.75	22	22.25	22.5	22.75	23	
1	.3193	.2554	.2555	.2736	.3106	.3166	
2	.5052	.4053	.438	.398	.4644	.4554	
3	.6161	.5832	.546	.5381	.545	.5647	
4	.6611	.6809	.6618	.6158	.6005	.6549	
5	.8372	.8668	.8273	.7431	.736	.7332	
6	.9514	.9483	.9077	.8461	.8368	.8439	
7	.965	.9718	.9712	.951	.9355	.9417	
8	.9785	.9935	1	.9877	.9975	.9829	
12	.9834	.9689	.949	.935	.9482	.9476	
13	.9856	.9777	.9631	.9454	.9428	.9637	
14	.999	1	.9931	1	1	1	
15	1	.9926	.9904	.9843	.9925	.9933	
16	.967	.952	.9464	.9004	.9255	.9404	
17	.8473	.8227	.7785	.7623	.7912	.803	
18	.8081	.7452	.6828	.6173	.6116	.6016	
19	.6441	.5996	.5487	.4407	.4463	.4212	
20	.4537	.449	.3984	.3001	.2945	.2964	
Ptot	10.122	10.21	10.402	10.578	10.392	10.466	
Ppr	10.558	10.592	10.484	10.514	10.496	10.562	
Ppl	12	12.06	12.056	12.008	12.052	12.174	
Tpl	543.12	543.35	543.68	543.84	543.98	544.49	
RAKE α	13.614	13.55	13.514	13.544	13.508	13.53	
BETA	44.23	44.14	44.029	43.97	44.03	44.27	

## A2. INSTRUMENTED BLADE DATA

The instrumented blade data is listed in the following two pages in the order of increased suction.

INSTRUMENTED BLADE PRESSURE DISTRIBUTION  
LOW SUCTION  
Q = 16.066 IN H<sub>2</sub>O

X	X/4	PRESSURE SIDE	SUCTION SIDE
0	0		10.67
.08	.0159680639	3.686	-30.36
.16	.0319361277	2.272	-18.76
.24	.0479041916	2.338	-14.7
.43	.0858283433	1.24	-14.04
.61	.1217564870	.73	-13.68
.98	.1956087824	.228	-14.2
1.35	.2694610778	1.84	-13.44
1.72	.3433133733	2.012	-11.28
2.1	.4191616766	1.544	-2.748
2.47	.4930132721	.778	-8.278
2.84	.5668662675	.526	-6.57
3.21	.6407185629	.854	-4.276
3.58	.7145708583	.878	-3.81
3.95	.7884231537	1.338	-2.86
4.15	.8283433134	1.488	-2.426
4.35	.8682634731	1.348	-2.284
4.55	.9081836327	.46	-2.084
4.75	.9481037924	-1.006	-1.912
4.95	.9880239521	-2.724	-1.868

INSTRUMENTED BLADE PRESSURE DISTRIBUTION  
MEDIUM SUCTION  
Q = 16.512 IN H<sub>2</sub>O

X	X/4	PRESSURE SIDE	SUCTION SIDE
0	0		-11.22
.08	.0159680639	3.718	-31.84
.16	.0319361277	1.826	-20.29
.24	.0479041916	1.838	-16
.43	.0858283433	1.068	-14.62
.61	.1217564870	.678	-14.31
.98	.1956087824	.832	-14.55
1.35	.2694610778	1.312	-14.16
1.72	.3433133733	1.518	-12.57
2.1	.4191616766	1.266	-10.2
2.47	.4930132721	.854	-9.604
2.84	.5668662675	.624	-6.884
3.21	.6407185629	.226	-5.262
3.58	.7145708583	.27	-4.144
3.95	.7884231537	1.184	-3.246
4.15	.8283433134	1.332	-2.842
4.35	.8682634731	1.028	-2.62
4.55	.9081836327	.184	-2.326
4.75	.9481037924	.312	-2.25
4.95	.9880239521	-3.07	-2.144

INSTRUMENTED BLADE PRESSURE DISTRIBUTION  
HIGH SUCTION  
Q = 15.762 IN H2O

X		PRESSURE SIDE	SUCTION SIDE
0	0		-12.47
.08	.0152680639	3.83	-32.47
.16	.0319361277	2.12	-21.58
.24	.0479041916	2.284	-16.46
.43	.0858283433	1.112	-14.73
.61	.1217564870	.548	-14.38
.98	.1956087824	.742	-14.6
1.35	.2694610778	1.578	-13.82
1.72	.3433133733	1.866	-12.27
2.1	.4191616766	1.45	-9.988
2.47	.4930139721	.832	-8.434
2.84	.5668662675	.592	-6.812
3.21	.6407185629	.842	-5.202
3.58	.7145708583	.864	-4.022
3.95	.7884231537	1.132	-3.126
4.15	.8283433134	1.28	-2.834
4.35	.8682634731	1.188	-2.61
4.55	.9081836327	.542	-2.398
4.75	.9481037924	-.308	-2.248
4.95	.9880239521	-3.036	-2.156

### **A3. MANOMETER DATA**

The manometer data was taken by marking the tubes during the test conducted and recorded at completion. Polaroid pictures were taken during the tests and are also presented in this section.

MANIFOLD AND ENDWALL STATIC PRESSURES  
LOW SUCTION  
PRESSURES MEASURED IN INCHES OF WATER

PITCH (IN)	NORTH MANIFOLD	SOUTH MANIFOLD	NORTH ENDWALL	SOUTH ENDWALL
-7	-7.85	7.2		
3	-7.6	-7.7	-5.1	-5
6			-4.85	-4.85
9			-5	-4.7
12			-4.85	-4.6
13	-5.5	-6		
15			-5.1	-4.7
18			-5	-5.2
21			-5.3	-5
23	-4.7	-4.9		
24			-4.8	-5
27			-5.5	-5
30			-5.4	-5.05
33	-4.4	-4.7	-5.5	-5.3
36			-5.3	-5.3
39			-5.4	-5.1
42			-5.5	-5
43	-4.4	-4.8		
45			-5.2	-5
48			-5	-5.1
51			-5	-5
53	-4.5	-4.9		
54			-4.85	-4.85
57			-5	-4.85
60			-4	-4.4
63	-4.7	-5.35		
73	-5.6	-6.25		

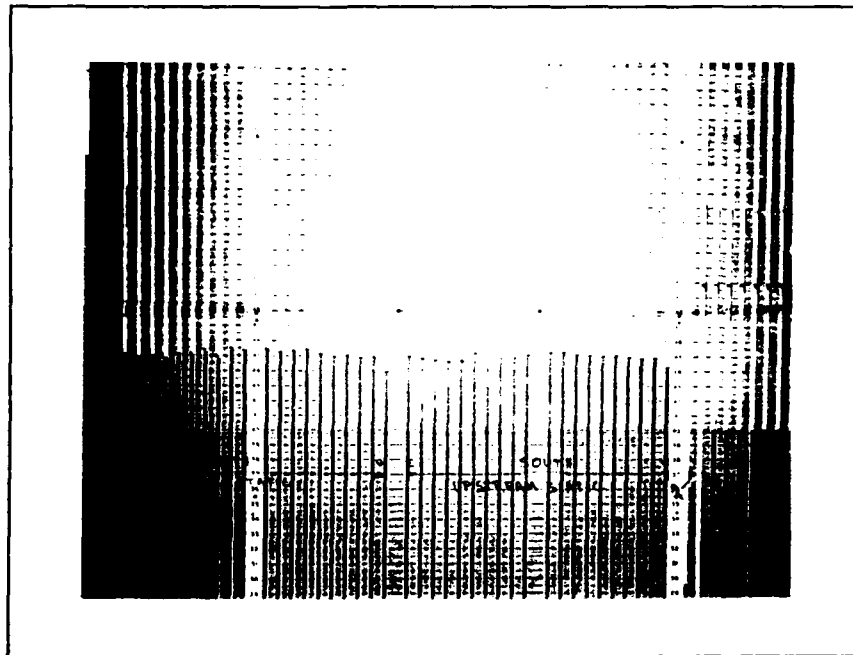


Figure A1. Manometer A for Low Suction Tests

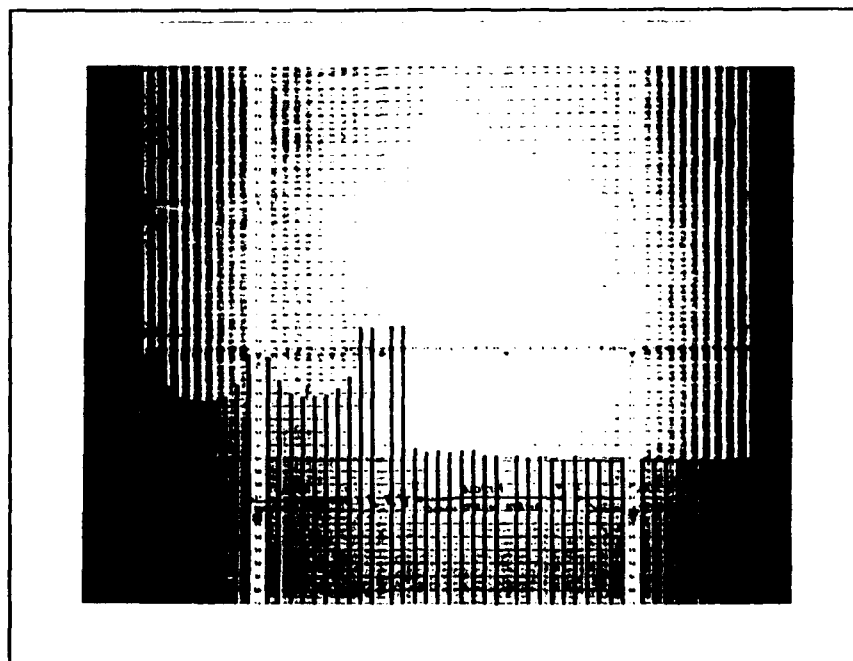


Figure A2. Manometer B for Low Suction Tests



MANIFOLD AND ENDWALL STATIC PRESSURES  
MEDIUM SUCTION  
PRESSURES MEASURED IN INCHES OF WATER

PITCH (IN)	NORTH MANIFOLD	SOUTH MANIFOLD	NORTH ENDWALL	SOUTH ENDWALL
-7	-11.55	-12		
3	-11.1	-10.95	-5.1	-5
6			-5	-5
9			-5.1	-5.05
12			-4.85	-4.8
13	-6.35	-6.95		
15			-5.3	-4.9
18			-5.2	-5.3
21			-5.5	-5.15
23	-5.15	-5.55		
24			-4.95	-5.2
27			-5.5	-5.1
30			-5.35	-5.1
33	-4.95	-5.35	-5.4	-5.4
36			-5.3	-5.35
39			-5.3	-5
42			-5.3	-4.9
43	-5.1	-5.7		
45			-5	-4.8
48			-5	-4.8
51			-4.8	-4.6
53	-5.4	-6.3		
54			-4.9	-4.4
57			-4.3	-4.2
60			-4	-4.1
63	-6.25	-7.9		
73	-8.35	-10		

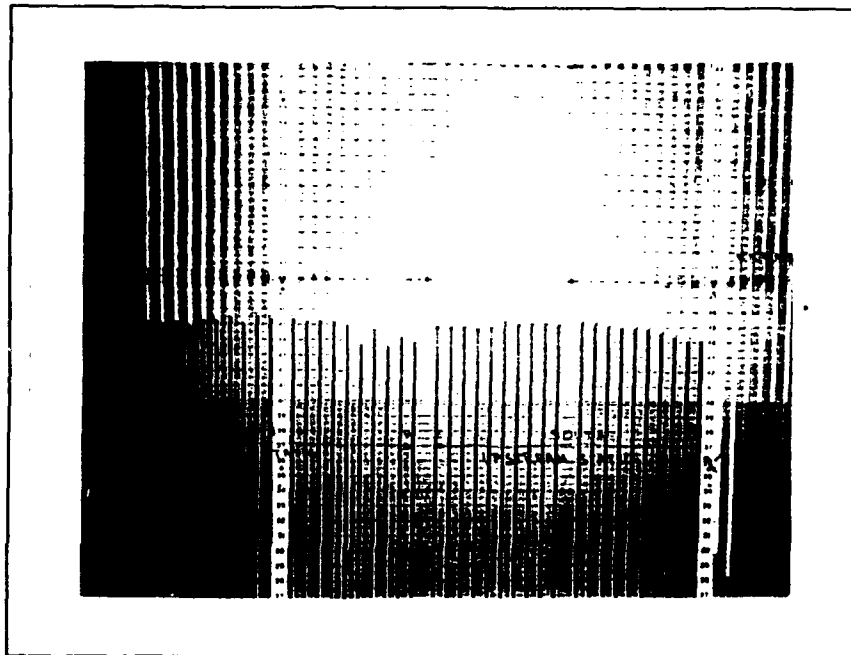


Figure A3. Manometer A for Medium Suction Tests

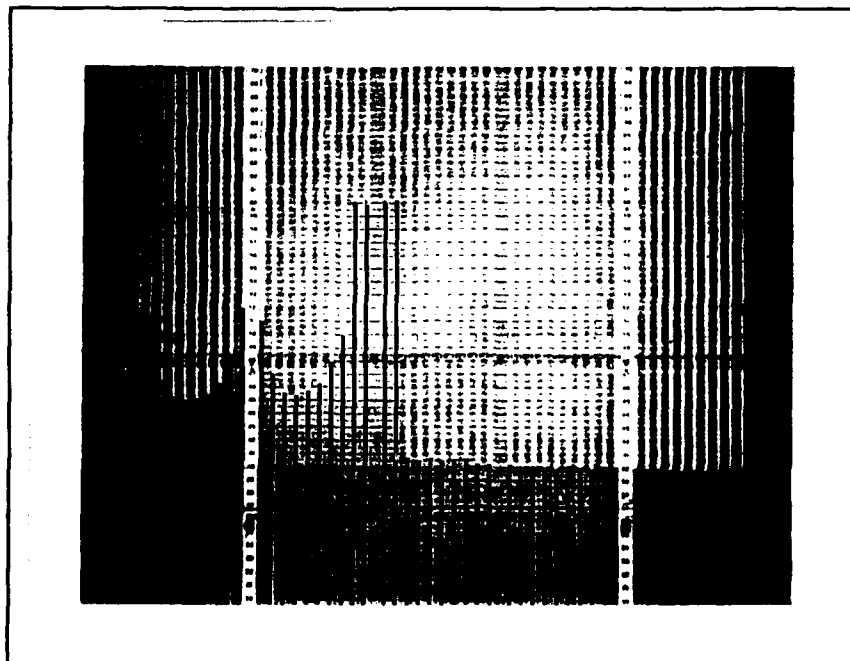


Figure A4. Manometer B for Medium Suction Tests

MANIFOLD AND ENDWALL STATIC PRESSURES  
HIGH SUCTION  
PRESSURES MEASURED IN INCHES OF WATER

PITCH (IN)	NORTH MANIFOLD	SOUTH MANIFOLD	NORTH ENDWALL	SOUTH ENDWALL
-7	-15.4	-16		
3	-14.3	-13.9	-5.1	-5
6			-5	-5
9			-5.1	-5
12			5	4.8
13	-7.2	-8		
15			-5.3	-4.95
18			-5.2	-5.4
21			-5.5	-5.2
23	-5.5	-6		
24			-5.05	-5.25
27			-5.7	-5.2
30			-5.5	-5.2
33	-5.3	-5.85	-5.6	-5.4
36			-5.4	-5.3
39			-5.4	-5.1
42			-5.35	-5
43	-5.55	-6.45		
45			-5.2	-4.9
48			-5	-4.85
51			-4.8	-4.5
53	-6.15	-7.6		
54			-5	-4.3
57			-4.3	-4
60			-4	-4.1
63	-7.8	-10.6		
73	-11.5	-14.2		

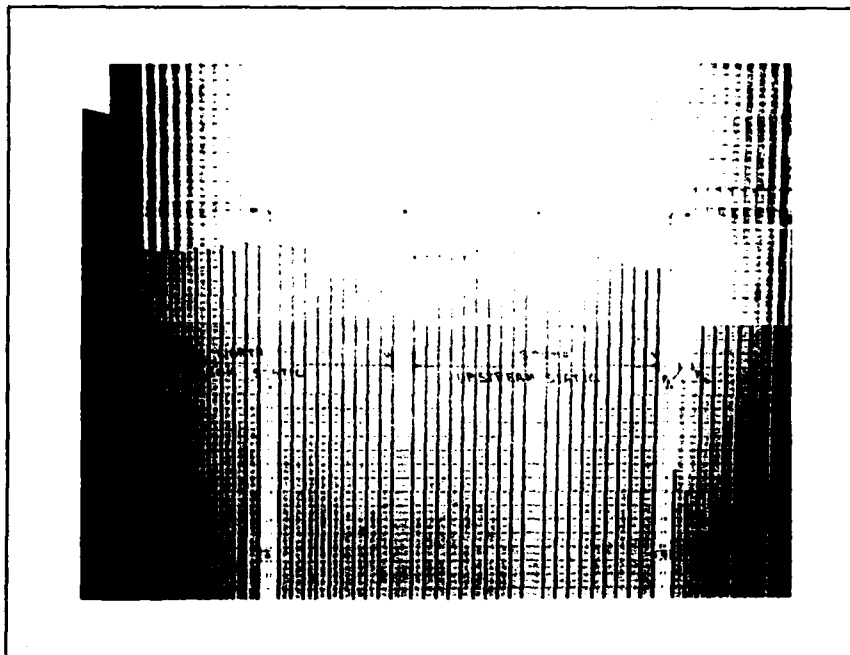


Figure A5. Manometer A for High Suction Tests

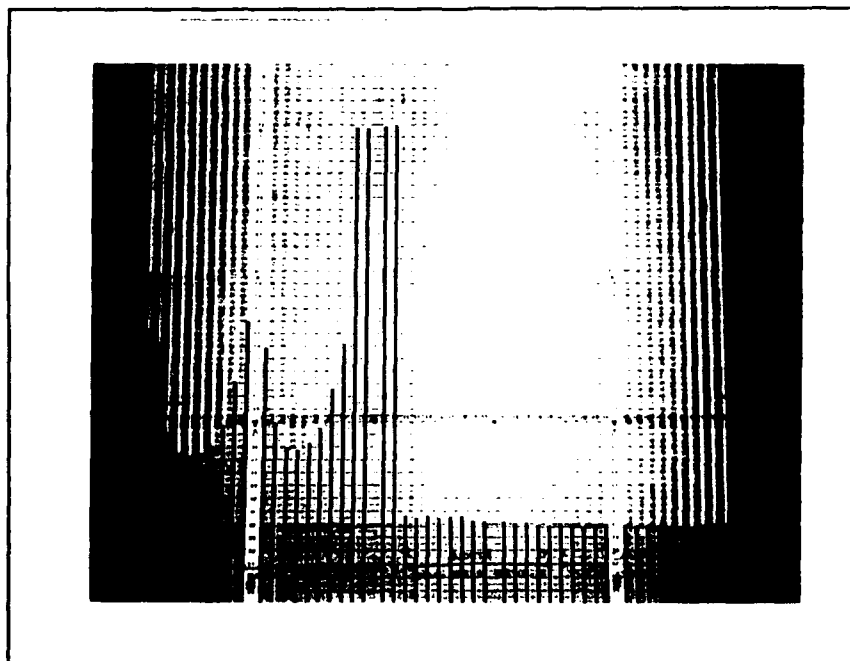


Figure A6. Manometer B for High Suction Tests

#### **A4. LDV DATA**

The LDV data is presented sequentially from stations 1 to 18. Graphs not seen in the main body are found after the corresponding data set.

FLOW ANGLE ADJUSTMENT

Velocity in M/SEC

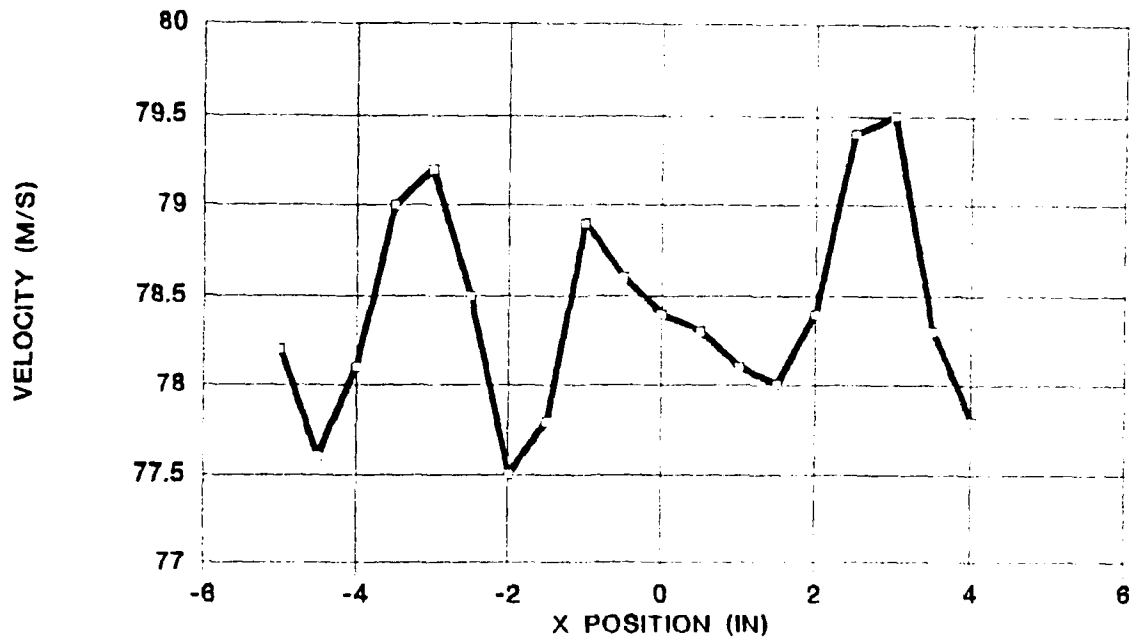
Data is from the following files:

A1101392,1.19

X(in)	Z(in)	U-Mean	U-Standard	V-Mean	V-Standard	UV-Angle Mean
		Velocity	Deviation	Velocity	Deviation	
4	-6.29	53.1	1.09	56.9	1.36	47
3.5	-6.29	53.1	1.25	57.5	1.67	47.2
3	-6.29	54.1	1.24	58.2	1.36	47.1
2.5	-6.29	54.4	1.09	57.8	1.48	46.7
2	-6.29	54.1	1.06	56.7	1.42	46.4
1.5	-6.29	54	1.31	56.9	1.34	46.1
1	-6.29	53.7	1.22	56.7	1.32	46.6
.5	-6.29	53.5	1.11	57.1	1.55	46.9
0	-6.29	53.7	1.08	57.2	1.27	46.8
-.5	-6.29	54.2	1.26	56.9	1.38	46.4
-1	-6.29	55	1.16	56.5	1.24	45.8
-1.5	-6.29	54.2	1.03	55.9	1.18	45.9
-2	-6.29	53.5	1.07	56.1	1.39	46.4
-2.5	-6.29	54.2	1.36	56.8	1.4	46.4
-3	-6.29	54.7	1.13	57.3	1.32	46.3
-3.5	-6.29	54.8	1.09	56.9	1.34	46.1
-4	-6.29	54.7	1.03	55.8	1.38	45.6
-4.5	-6.29	54	1.28	55.7	1.37	45.9
-5	-6.29	54.1	1.43	56.4	1.38	46.2

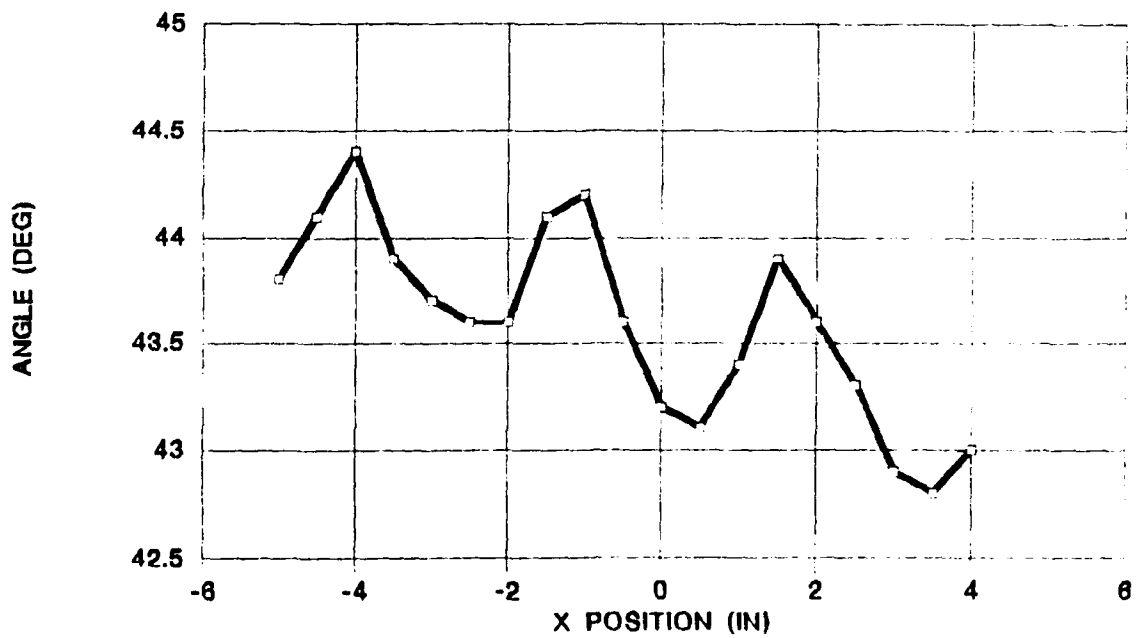
## STATION 1 FLOW ADJUST

TOTAL VELOCITY



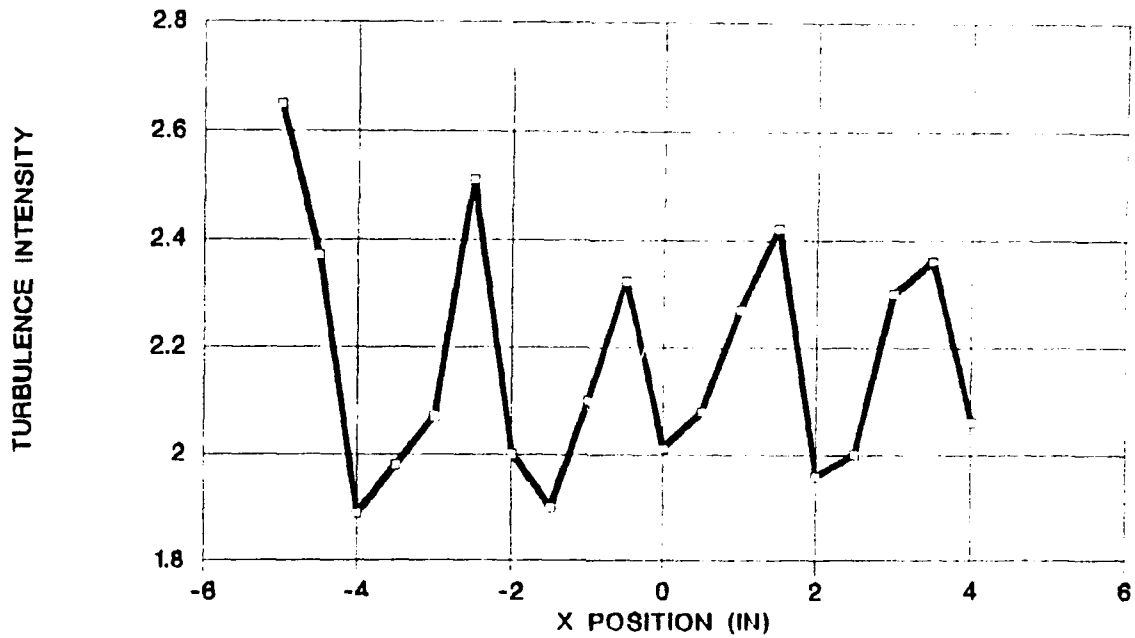
## STATION 1 FLOW ADJUST

FLOW ANGLE



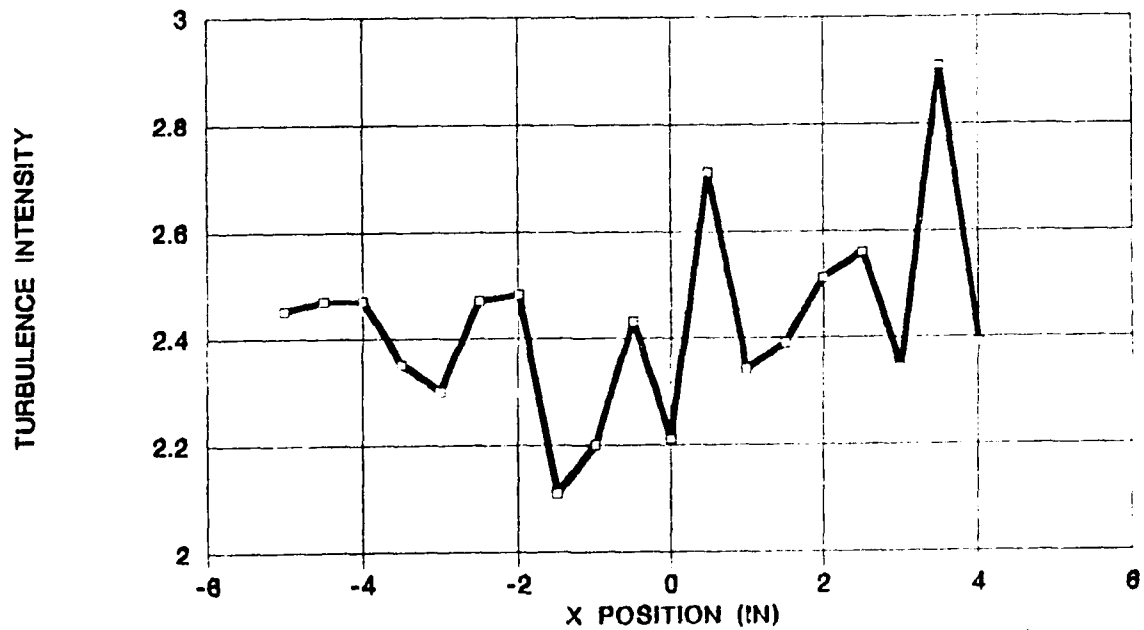
## STATION 1 FLOW ADJUST

U TURBULENCE INTENSITY



## STATION 1 FLOW ADJUST

V TURBULENCE INTENSITY





BASE LINE STATION 1  
Velocity in M/SEC  
Data is from the following files:  
ST1A1209,1.41

STATION 1						
X(in)	Z(in)	U-Mean VEL	U-Standard DEV	V-Mean Velocity	V Standard Deviation	UV-Angle Mean
-1	-6.29	0	0	0	0	0
-.875	-6.29	60.7	1.21	60.5	1.23	44.9
-.75	-6.29	60.7	1.24	60.6	1.28	45
-.625	-6.29	60.7	1.22	60.8	1.17	45
-.5	-6.29	60.6	1.22	60.9	1.23	45.2
-.375	-6.29	60.5	1.88	61	1.17	45.2
-.25	-6.29	60.3	1.28	61.1	1.19	45.4
-.125	-6.29	60	1.22	61	1.27	45.5
0	-6.29	59.7	1.25	61	1.2	45.6
.125	-6.29	59.7	1.2	61	1.16	45.6
.25	-6.29	59.6	1.2	60.9	1.2	45.6
.375	-6.29	59.3	1.48	60.9	1.28	45.7
.5	-6.29	59.1	1.48	61	1.26	45.9
.625	-6.29	59.2	1.12	61	1.38	45.8
.75	-6.29	59.3	1.27	60.9	1.22	45.8
.875	-6.29	59.4	1.19	60.8	1.25	45.7
1	-6.29	59.5	1.3	60.7	1.31	45.6
1.13	-6.29	59.8	1.22	60.7	1.28	45.5
1.25	-6.29	59.9	1.3	60.7	1.26	45.4
1.38	-6.29	60.1	1.34	60.6	1.3	45.2
1.5	-6.29	60.1	1.69	60.6	1.13	45.2
1.62	-6.29	60.3	1.4	60.6	1.21	45.1
1.75	-6.29	59.9	1.23	60.6	1.22	45.3
1.87	-6.29	60.1	1.24	60.6	1.28	45.3
2	-6.29	60	1.26	60.7	1.27	45.3
2.13	-6.29	60.3	1.14	61.1	1.21	45.4
2.25	-6.29	60.3	1.18	61.2	1.38	45.4
2.37	-6.29	60.4	1.12	61.4	1.25	45.5
2.5	-6.29	60.4	1.18	61.6	1.31	45.6
2.63	-6.29	60.6	1.12	61.2	1.22	45.6
2.75	-6.29	60.5	1.22	62	1.32	45.7
2.87	-6.29	60.7	1.21	62.3	1.3	45.8
3	-6.29	60.6	1.28	62.3	1.41	45.8
3.12	-6.29	60.5	1.32	62.2	2.22	45.8
3.25	-6.29	60.2	1.37	62.3	1.25	46
3.37	-6.29	60.1	1.43	62.3	1.25	46
3.5	-6.29	59.7	1.42	62	1.18	46.1
3.62	-6.29	59.4	1.44	61.8	1.21	46.1
3.75	-6.29	59.4	1.38	61.6	1.18	46
3.87	-6.29	59.4	1.32	61.4	1.17	46
4	-6.29	59.3	1.26	61.1	1.2	45.9

Velocity in M/SEC

Data is from the following files:

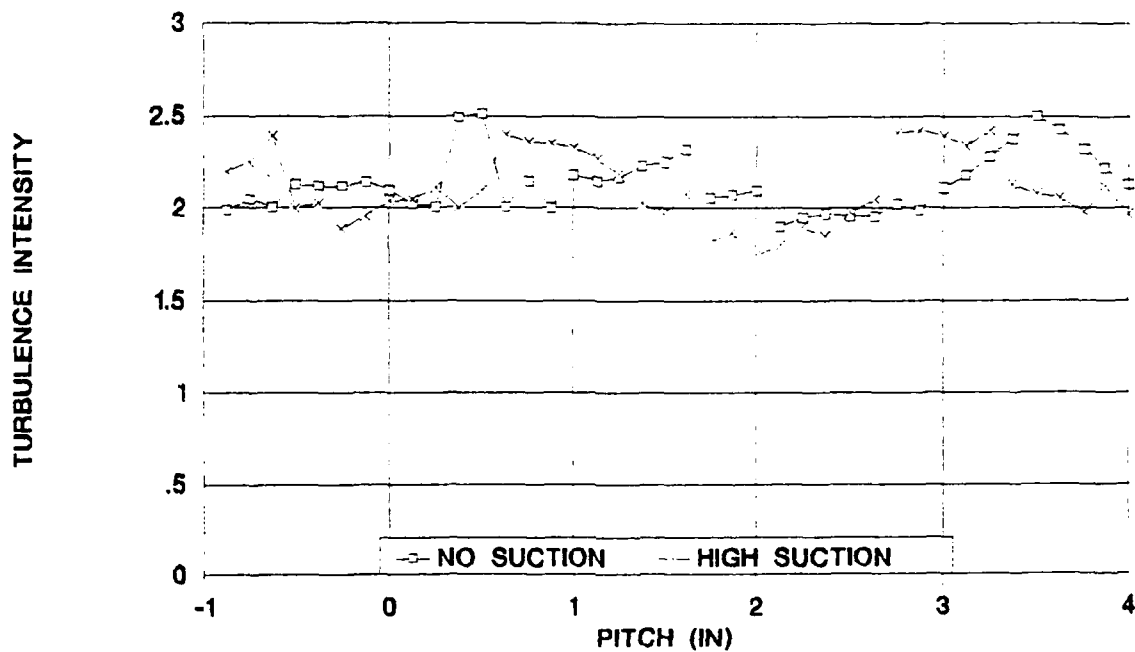
ST1SUCK.1,41

STATION 1 HIGH SUCTION

X(mm)	Z(mm)	U-Mean Velocity	U-Standard Deviation	V-Mean Velocity	V-Standard Deviation	UV-Angle Mean
-25.4	-160	58	1.35	58.6	1.89	45.3
-22.2	-160	58	1.28	58.5	2.1	45.3
-19	-160	58.2	1.31	58.8	2.01	45.3
-15.9	-160	58.1	1.38	58.9	2.03	45.4
-12.7	-160	57.9	1.16	59	1.73	45.5
-9.52	-160	57.7	1.17	58.9	2.03	45.6
-6.35	-160	57.6	1.09	59	1.94	45.7
-3.17	-160	57.7	1.13	59.2	2.35	45.7
0	-160	57.1	1.17	59.2	1.84	46.1
3.17	-160	57.1	1.17	59.2	2.91	46.1
6.35	-160	57	1.2	59.4	2.29	46.2
9.52	-160	57.2	1.14	59.5	2.38	46.1
12.7	-160	57.1	1.2	59.6	2.15	46.2
15.9	-160	56.7	1.36	59.5	2.05	46.4
19	-160	56.7	1.34	59.4	1.93	46.4
22.2	-160	56.6	1.33	59	2.2	46.2
25.4	-160	56.7	1.32	59	1.86	46.1
28.6	-160	56.6	1.29	58.7	1.93	46
31.7	-160	56.9	1.24	58.4	2.59	45.8
34.9	-160	56.8	1.16	58.4	2.45	45.8
38.1	-160	57.1	1.13	58.5	2.18	45.7
41.3	-160	57.2	1.19	58.5	2.12	45.6
44.5	-160	57.6	1.04	58.6	2.38	45.5
47.6	-160	57.7	1.08	58.9	2.29	45.6
50.8	-160	57.8	1.01	59.1	2.28	45.6
54	-160	58	1.04	59.3	2.47	45.6
57.2	-160	58.2	1.11	59.6	2.21	45.7
60.3	-160	58.1	1.08	59.9	1.61	45.9
63.5	-160	58.3	1.15	60	3.01	45.8
66.7	-160	57.9	1.19	60.1	2.57	46
69.8	-160	57.4	1.39	60.3	2.46	46.4
73	-160	57.4	1.39	60.3	2.08	46.5
76.2	-160	57.1	1.37	60.1	2.72	46.4
79.4	-160	56.8	1.33	60.2	2.18	46.6
82.6	-160	56.7	1.37	60	3.02	46.6
85.7	-160	56.6	1.2	59.8	2.75	46.6
88.9	-160	56.5	1.17	59.6	2.77	46.5
92.1	-160	56.5	1.16	59.6	1.99	46.5
95.2	-160	56.5	1.12	59.6	2.54	46.5
98.4	-160	56.8	1.2	59.4	2.55	46.3
102	-160	56.9	1.12	59.2	2.89	46.1

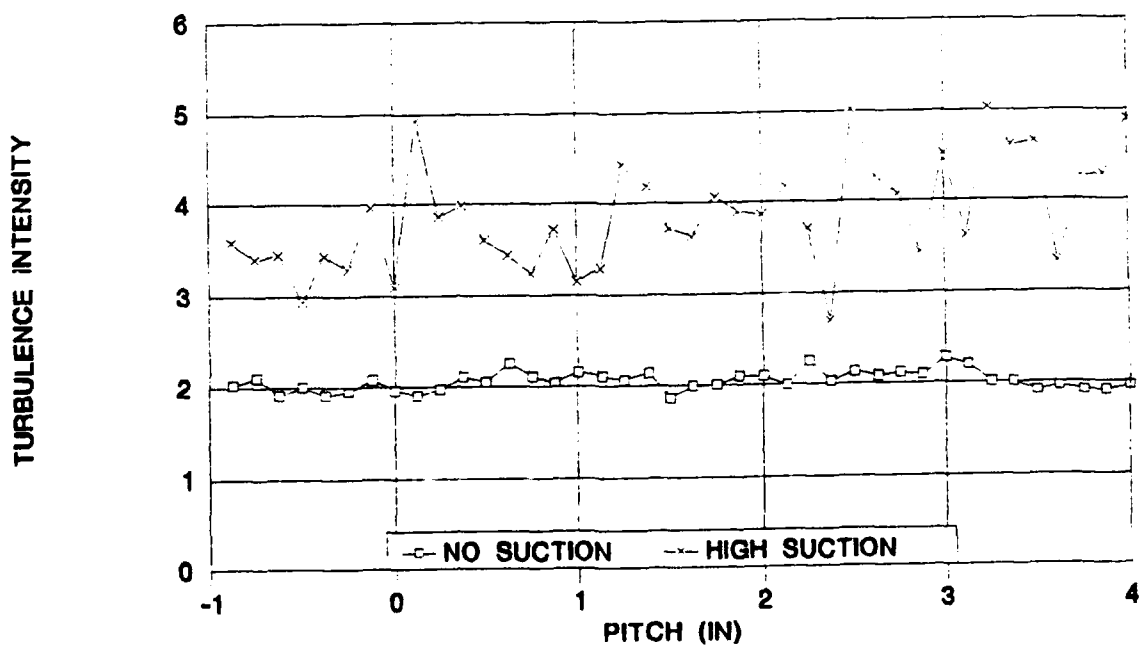
# STATION 1

## U TURBULENCE INTENSITY



# STATION 1

## V TURBULENCE INTENSITY



Velocity in M/SEC  
 Data is from the following files:  
 ST2A1118,1,33  
 STATION 2 SUCTION SIDE

X(in)	Z(in)	U-Mean Velocity	U-Standard Deviation	V-Mean Velocity	V-Standard Deviation	UV-Angle Mean
-1.08	-4.79	78.3	3.75	39.1	1.35	26.5
-1.02	-4.79	89.6	2.53	67.4	4.74	36.9
-1.01	-4.79	89	1.61	68	4.53	37.4
-1	-4.79	88.3	1.65	69.2	4	38.1
-.992	-4.79	87	1.75	69.8	3.76	38.7
-.979	-4.79	85.2	1.88	70	3.94	39.4
-.964	-4.79	83.5	1.8	70.9	3.49	40.3
-.948	-4.79	81.3	1.79	71.5	3.24	41.3
-.93	-4.79	79.1	1.89	72	3.05	42.3
-.911	-4.79	76.6	1.72	72.2	2.96	43.3
-.889	-4.79	75.3	1.78	72.7	2.59	44
-.866	-4.79	73.3	1.57	72.6	2.37	44.7
-.84	-4.79	71.2	1.54	72.5	2.42	45.5
-.811	-4.79	69.4	1.6	72.5	2.41	46.3
-.78	-4.79	67.5	1.52	72.2	1.87	46.9
-.745	-4.79	66	1.45	72.4	1.74	47.7
-.707	-4.79	64.2	1.34	71.8	1.66	48.2
-.665	-4.79	62.8	1.27	71.4	1.48	48.6
-.619	-4.79	61	1.25	70.7	1.61	49.2
-.569	-4.79	59.7	1.25	70.4	1.41	49.7
-.513	-4.79	58.4	1.23	69.8	1.32	50.1
-.452	-4.79	57.1	1.19	69.1	1.33	50.5
-.384	-4.79	55.9	1.14	68.9	1.33	50.9
-.31	-4.79	54.7	1.1	68.4	1.2	51.3
-.229	-4.79	53.5	1.07	67.7	1.16	51.7
-.14	-4.79	52.3	1.06	67.3	1.14	52.1
-.041	-4.79	51.2	1.07	66.7	1.17	52.5
.0672	-4.79	50.1	1.09	65.9	1.16	52.7
.186	-4.79	49	1.03	64.9	1.14	52.9
.317	-4.79	47.9	1.03	63.8	1.19	53.1
.462	-4.79	46.6	1.05	62.4	1.21	53.2
.62	-4.79	45.6	1.13	60.8	1.23	53.1
.795	-4.79	44.4	1.22	58.7	1.24	52.9

Velocity in M/SEC

Data is from the following files:

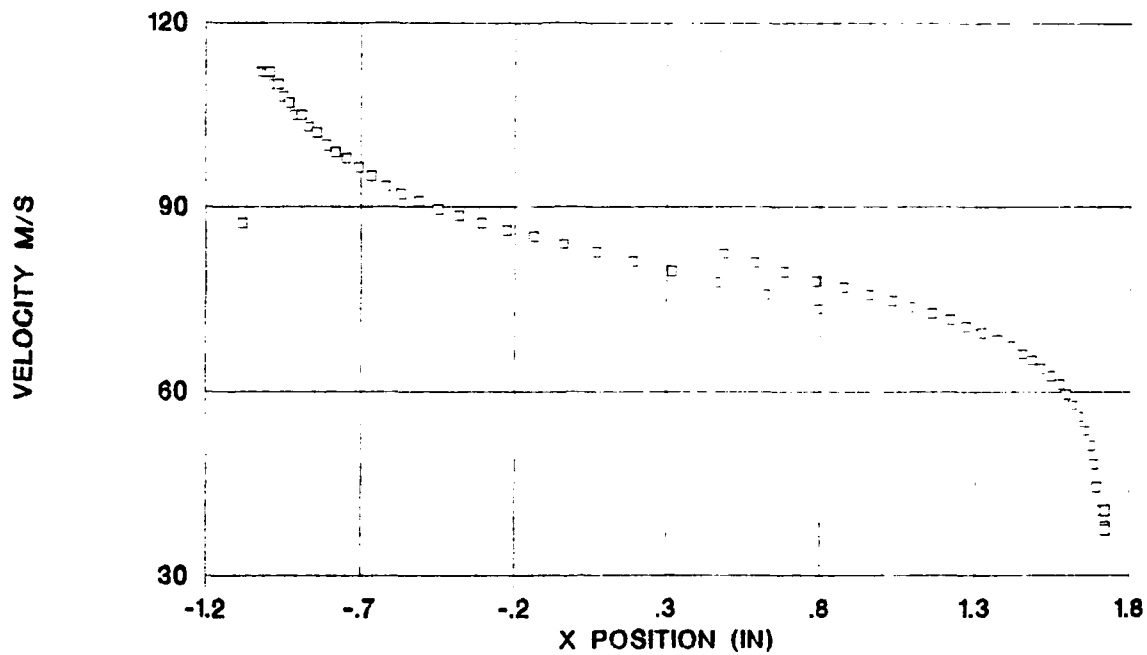
ST2B1209.1,30

STATION 2B PRESSURE SIDE

X(in)	Z(in)	U-Mean Velocity	U-Standard Deviation	V-Mean Velocity	V-Standard Deviation	UV-Angle Mean
1.72	-4.79	33.9	5.91	22.6	7.21	33.7
1.72	-4.79	31.6	4.22	21.5	3.68	34.3
1.72	-4.79	31.3	3.46	20.6	3.62	33.4
1.71	-4.79	34.9	1.95	20.8	2.27	30.7
1.69	-4.79	38	1.89	23.4	1.87	31.6
1.68	-4.79	40.5	1.74	26.2	1.73	32.9
1.67	-4.79	42.3	1.57	28.6	1.74	34.1
1.65	-4.79	43.9	1.42	30.7	1.48	35
1.63	-4.79	45.2	1.3	32.8	1.51	36
1.61	-4.79	46.1	1.17	34.9	1.56	37.1
1.59	-4.79	47	1.15	36.8	1.39	38
1.57	-4.79	47.6	1.05	38.5	1.38	39
1.54	-4.79	48.1	1.05	40.1	1.44	39.8
1.51	-4.79	48.5	1.05	41.6	1.24	40.6
1.48	-4.79	49	.969	43	1.25	41.2
1.45	-4.79	49.3	.98	44.2	1.35	41.9
1.41	-4.79	49.7	.978	45.6	1.21	42.5
1.37	-4.79	50.2	1	46.8	1.19	43
1.32	-4.79	50.5	.961	48	1.3	43.6
1.27	-4.79	50.7	.948	49.3	1.23	44.2
1.22	-4.79	51	.977	50.6	1.18	44.8
1.16	-4.79	51.4	1	51.8	1.19	45.3
1.09	-4.79	51.6	1.01	52.9	1.18	45.7
1.03	-4.79	51.9	1.07	54.1	1.29	46.1
.953	-4.79	52.3	1.13	55.2	1.23	46.5
.872	-4.79	52.7	1.32	56.3	1.17	46.9
.782	-4.79	53.2	1.22	57.4	1.25	47.2
.683	-4.79	53.9	1.54	58.7	1.1	47.5
.585	-4.79	54.5	1.69	60	1.2	47.8
.485	-4.79	56.2	2.35	60.5	.981	47.1

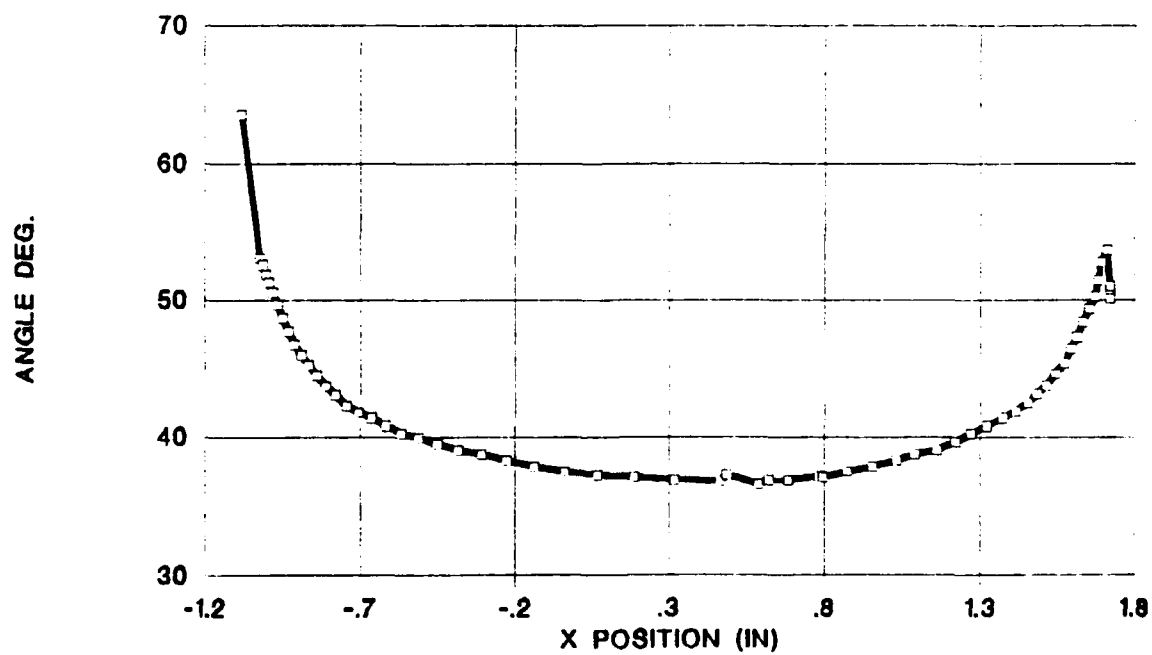
# STATION 2

## TOTAL VELOCITY



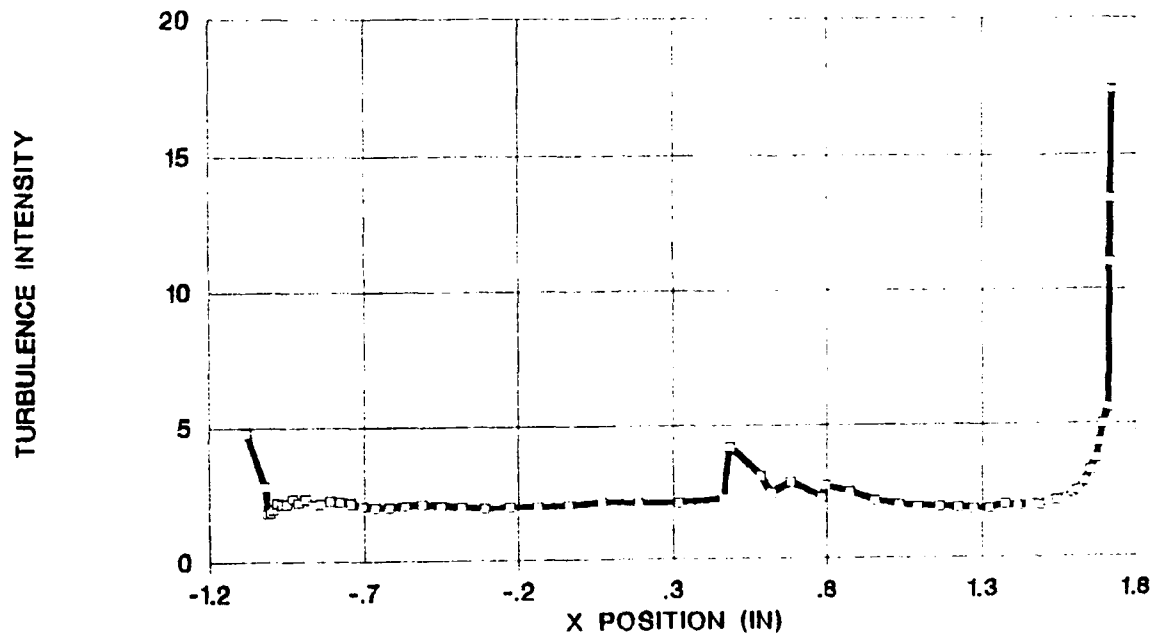
# STATION 2

## FLOW ANGLE



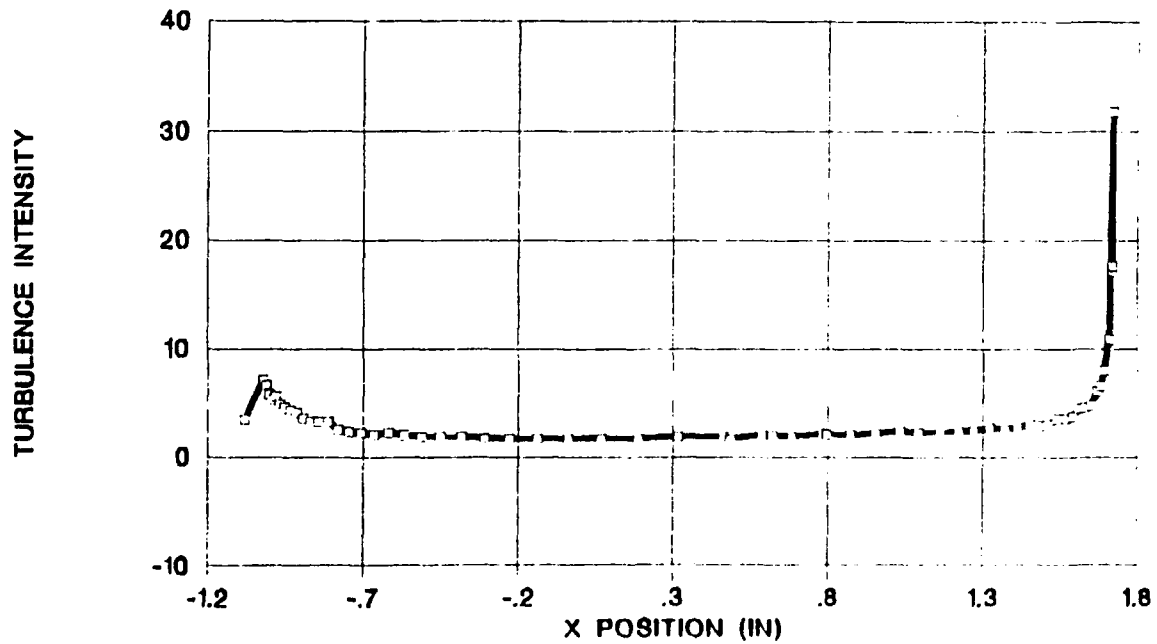
## STATION 2

### U TURBULENCE INTENSITY



## STATION 2

### V TURBULENCE INTENSITY



Velocity in M/SEC

Data is from the following files:

ST7A0126,1,31

STATION 7A SUCTION SIDE

X(in)	Z(in)	U-Mean Velocity	U-Standard Deviation	V-Mean Velocity	V-Standard Deviation	UV-Angle Mean
-.169	-3.29	41.8	10.6	58	9.3	54.2
-.164	-3.29	35.7	7.85	60.6	8.14	59.5
-.16	-3.29	43.6	9.57	60.5	8.35	54.2
-.15	-3.29	34.3	4.1	64	8.37	61.8
-.125	-3.29	36.1	4.07	67.7	8.8	61.9
-.101	-3.29	37.5	4.05	71.6	9.03	62.4
-.0837	-3.29	38.5	3.95	73.4	9.35	62.3
-.0642	-3.29	39.8	3.72	76.8	8.75	62.6
-.0428	-3.29	40.1	3.27	79.4	7.84	63.2
-.0192	-3.29	40.7	3.17	81.5	5.89	63.5
.0067	-3.29	40.9	2.47	82.7	4.1	63.7
.0352	-3.29	40.8	2.05	82.9	3.17	63.8
.0667	-3.29	40.5	1.71	82.3	2.66	63.8
.101	-3.29	40.2	1.45	81.9	1.93	63.9
.139	-3.29	39.8	1.39	81.1	2.16	63.8
.181	-3.29	39.3	1.4	80	2.45	63.9
.227	-3.29	38.9	1.35	79	2.97	63.8
.277	-3.29	38.4	1.24	78.2	1.61	63.9
.333	-3.29	38	1.25	77.2	1.39	63.8
.394	-3.29	37.6	1.54	76.2	1.54	63.7
.461	-3.29	37	1.36	75.2	1.66	63.8
.535	-3.29	36.4	1.36	73.9	1.65	63.8
.617	-3.29	35.9	1.16	72.9	1.74	63.8
.706	-3.29	35.2	1.26	71.7	1.44	63.8
.805	-3.29	34.6	1.33	70.4	1.3	63.8
.913	-3.29	33.9	1.08	69.2	1.1	63.9
1.03	-3.29	33.2	1.22	67.9	1.16	63.9
1.16	-3.29	32.4	1.22	66.3	1.2	64
1.24	-3.29	32	1.25	65.4	1.18	64
1.34	-3.29	31.4	1.75	64.3	1.21	64
1.44	-3.29	30.7	1.24	63.3	1.24	64.1



Velocity in M/SEC

Data is from the following files:

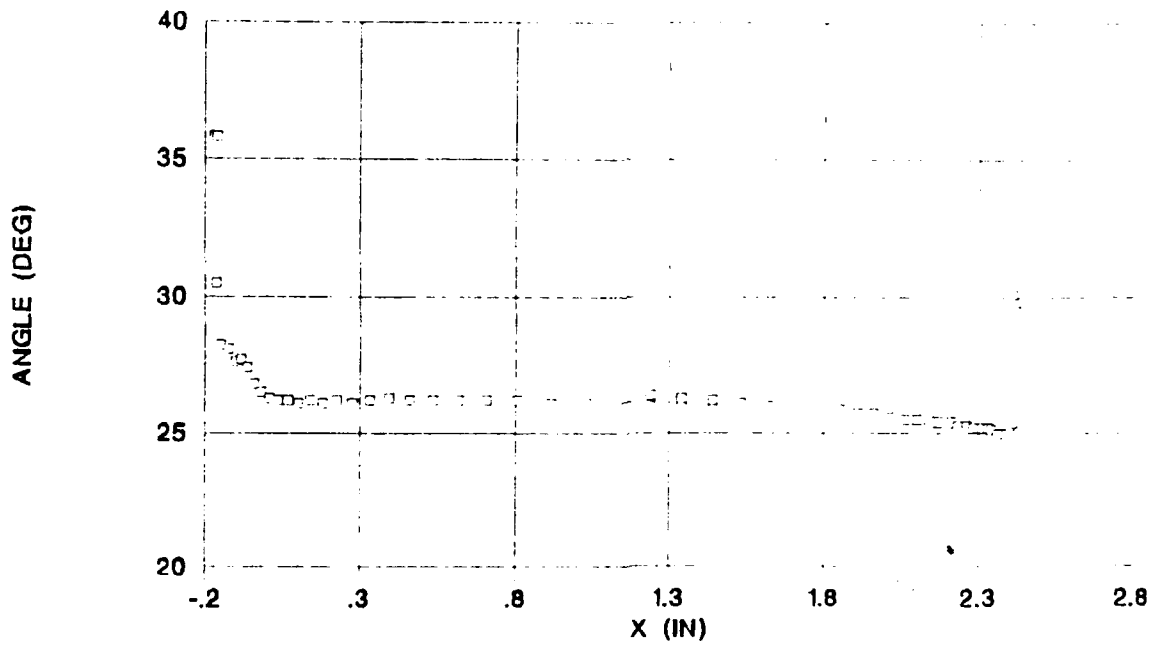
ST7B0121.1,30

STATION 7B PRESSURE SIDE

X(in)	Z(in)	U-Mean Velocity	U-Standard Deviation	V-Mean Velocity	V-Standard Deviation	UV-Angle Mean
2.42	-3.29	20.1	1.7	35.4	4.98	60.5
2.41	-3.29	23.6	2.68	40.7	8.56	59.9
2.41	-3.29	24.8	1.29	52.2	4.89	64.6
2.4	-3.29	25	.911	53.5	3.06	64.9
2.4	-3.29	25.2	.833	53.9	1.57	65
2.39	-3.29	25.2	.84	54.1	1.25	65
2.37	-3.29	25.3	.871	54.2	1.24	65
2.36	-3.29	25.4	.882	54.3	1.22	65
2.34	-3.29	25.5	.94	54.5	1.17	64.9
2.33	-3.29	25.7	.945	54.6	1.31	64.8
2.31	-3.29	25.8	.97	54.9	1.19	64.8
2.28	-3.29	25.9	.992	55.1	1.21	64.8
2.26	-3.29	26.1	1.04	55.4	1.16	64.7
2.23	-3.29	26.2	1.02	55.5	1.17	64.7
2.21	-3.29	26.4	1.01	55.8	1.22	64.6
2.17	-3.29	26.6	1.04	56	1.2	64.6
2.14	-3.29	26.8	1.08	56.4	1.14	64.6
2.1	-3.29	27	1.04	56.8	1.15	64.5
2.06	-3.29	27.3	1.07	57.1	1.21	64.5
2.01	-3.29	27.6	1.13	57.4	1.28	64.4
1.96	-3.29	27.9	1.13	58	1.2	64.3
1.91	-3.29	28.2	1.1	58.5	1.22	64.3
1.85	-3.29	28.6	1.16	59	1.19	64.1
1.78	-3.29	29	1.21	59.7	1.15	64.1
1.71	-3.29	29.3	1.17	60.3	1.31	64.1
1.62	-3.29	29.8	1.17	61.1	1.24	64
1.53	-3.29	30.4	1.28	62	1.27	63.9
1.44	-3.29	31	1.22	63	1.21	63.8
1.34	-3.29	31.7	1.26	64.1	1.25	63.7
1.24	-3.29	32.3	1.28	65.2	1.33	63.6

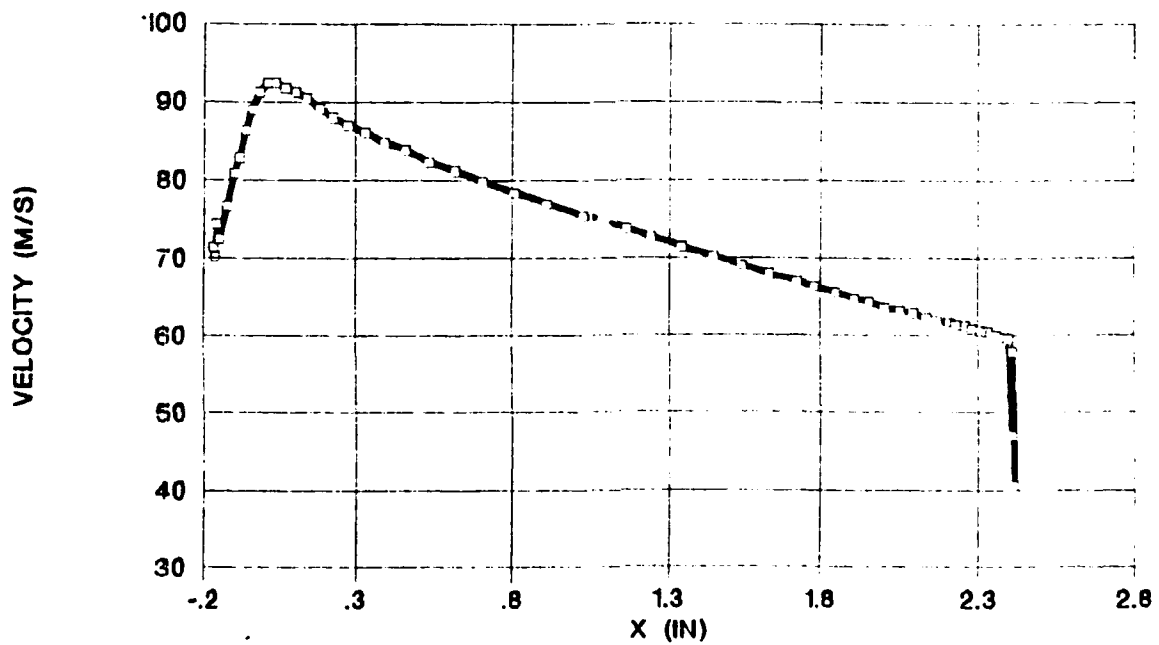
# STATION 7

## FLOW ANGLE



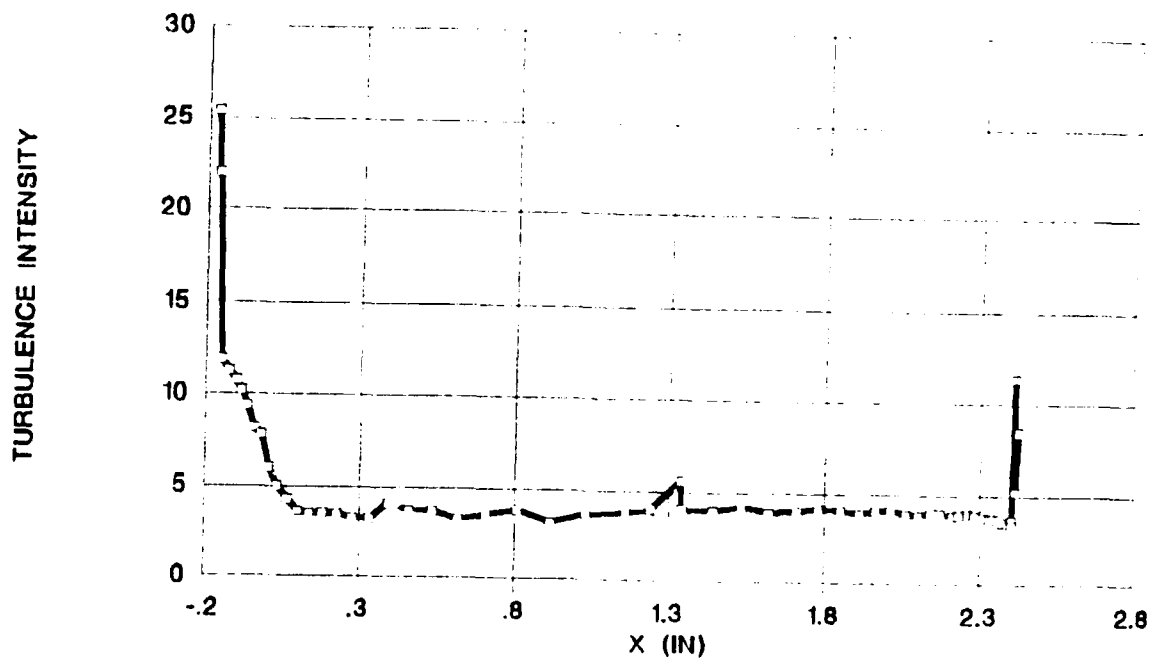
# STATION 7

## TOTAL VELOCITY



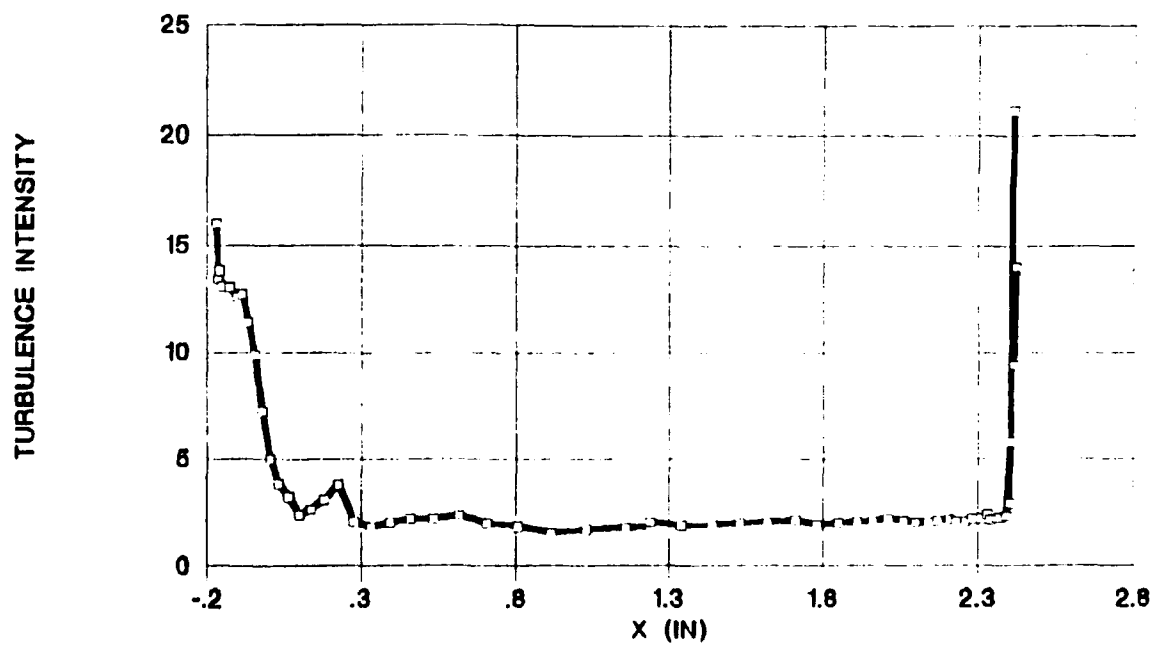
# STATION 7

## U TURBULENCE INTENSITY



# STATION 7

## V TURBULENCE INTENSITY



Velocity in M/SEC

Data is from the following files:

S15A1202,1,30

STATION 15 A SUCTION SIDE

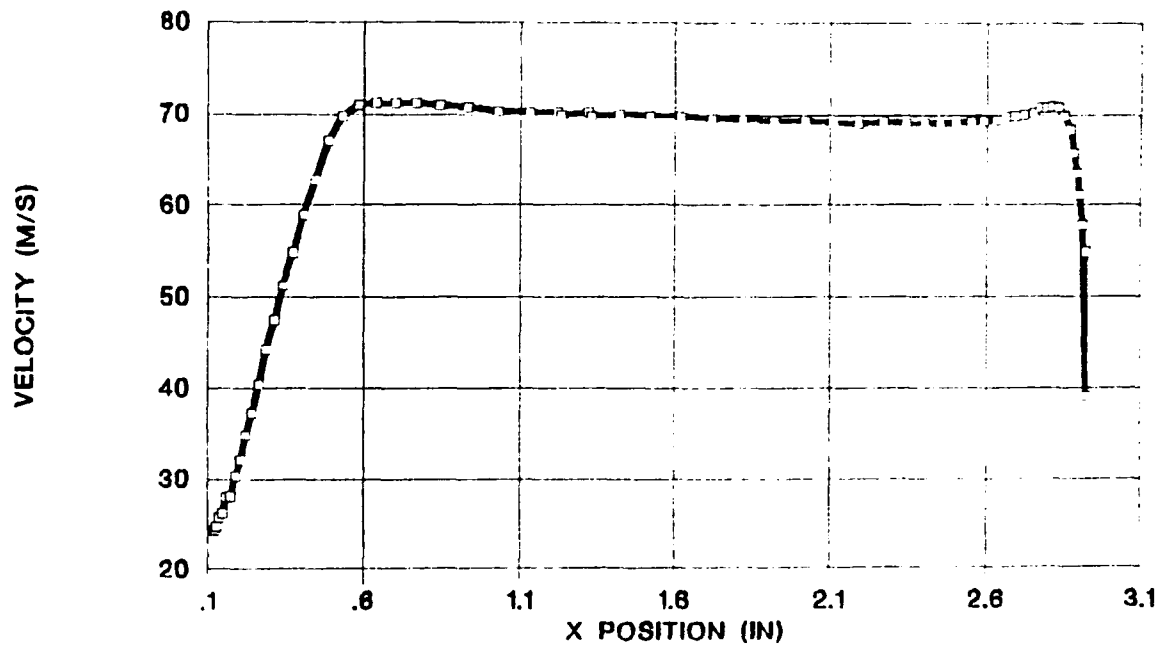
X(in)	Z(in)	U-Mean Velocity	U-Standard Deviation	V-Mean Velocity	V-Standard Deviation	UV-Angle Mean
.119	.036	-2.04	.904	24.1	2.66	94.8
.123	.036	-3.04	4	24.2	6.18	97.2
.129	.036	-3.08	4.15	24.6	6.51	97.1
.138	.036	-3.27	4.55	25.6	6.58	97.3
.149	.036	-3.48	4.74	26	6.7	97.6
.162	.036	-3.49	4.92	27.8	7.11	97.2
.175	.036	-3.19	4.96	27.9	6.97	96.5
.19	.036	-3.26	5.16	30.2	7.74	96.2
.206	.036	-3.48	5.21	32	7.97	96.2
.223	.036	-3.8	5.35	34.7	8.58	96.2
.243	.036	-3.67	5.25	37.1	8.9	95.7
.264	.036	-4.01	5.64	40.2	9.04	95.7
.288	.036	-4.13	5.25	44.1	8.68	95.3
.314	.036	-4.1	5.14	47.2	8.06	95
.342	.036	-4.3	5.09	51	7.79	94.8
.374	.036	-4.28	4.51	54.6	7.45	94.5
.408	.036	-4.32	4.59	58.7	7.21	94.2
.446	.036	-4.31	4.08	62.8	6.7	93.9
.488	.036	-4.38	3.49	66.9	4.97	93.7
.534	.036	-4.41	2.83	69.5	3.35	93.6
.584	.036	-4.22	2.19	70.9	2.05	93.4
.64	.036	-3.99	1.73	71.2	1.64	93.2
.701	.036	-3.68	1.36	71.2	1.48	93
.768	.036	-3.43	1.22	71.1	1.42	92.8
.842	.036	-3.17	1.18	70.9	1.4	92.6
.932	.036	-2.86	1.18	70.6	1.34	92.3
1.03	.036	-2.59	1.17	70.4	1.31	92.1
1.13	.036	-2.3	1.2	70.3	1.28	91.9
1.23	.036	-2.11	1.15	69.9	1.53	91.7
1.33	.036	-1.94	1.14	69.7	1.28	91.6

Velocity in M/SEC  
 Data is from the following files:  
 S15B1202.1,35  
 STATION 15B PRESSURE SIDE

X(in)	Z(in)	U-Mean Velocity	U-Standard Deviation	V-Mean Velocity	V-Standard Deviation	UV-Angle Mean
2.92	.0359	6.05	2.71	38.7	8.33	81.1
2.92	.036	7.54	2.91	54.3	7.26	82.1
2.91	.036	7.79	3.05	57.2	6.22	82.2
2.9	.036	8.32	3.13	60.7	5.81	82.2
2.89	.036	8.42	2.91	62.9	5.43	82.4
2.88	.036	8.51	2.66	65.3	4.74	82.6
2.87	.036	8.7	2.2	67.8	4.15	82.7
2.85	.036	8.73	1.8	69.3	2.95	82.8
2.83	.036	8.72	1.4	70.2	1.86	82.9
2.81	.036	8.67	1.16	70.3	1.61	83
2.79	.036	8.78	1.08	70.2	1.4	82.9
2.77	.036	8.77	1.06	70.1	1.23	82.9
2.75	.036	8.8	1.12	69.7	1.34	82.8
2.72	.036	8.79	1.09	69.5	1.19	82.8
2.69	.036	8.93	1.11	69.3	1.23	82.7
2.66	.036	9.02	1.19	69.1	1.37	82.6
2.63	.036	9.11	1.23	68.9	1.44	82.5
2.59	.0359	9.27	1.22	68.8	1.21	82.3
2.55	.036	9.41	1.19	68.5	1.41	82.2
2.5	.036	9.48	1.26	68.4	1.33	82.1
2.45	.0359	9.56	1.28	68.4	1.29	82
2.4	.036	9.8	1.26	68.4	1.24	81.8
2.34	.036	9.88	1.29	68.3	1.27	81.8
2.27	.036	10.1	1.26	68.4	1.32	81.6
2.19	.036	10.1	1.28	68.2	1.23	81.6
2.11	.036	10.2	1.27	68.3	1.26	81.5
2.02	.036	10.4	1.23	68.5	1.25	81.4
1.92	.036	10.4	1.21	68.7	1.35	81.4
1.82	.036	10.3	1.22	68.7	1.34	81.5
1.72	.036	10.4	1.18	68.9	1.19	81.4
1.62	.036	10.3	1.14	69	1.16	81.5
1.52	.036	10.2	1.09	69.1	1.15	81.6
1.42	.036	10.2	1.14	69.3	1.16	81.6
1.32	.036	10.1	1.1	69.5	1.2	81.7
1.22	.036	9.85	1.07	69.6	1.34	81.9

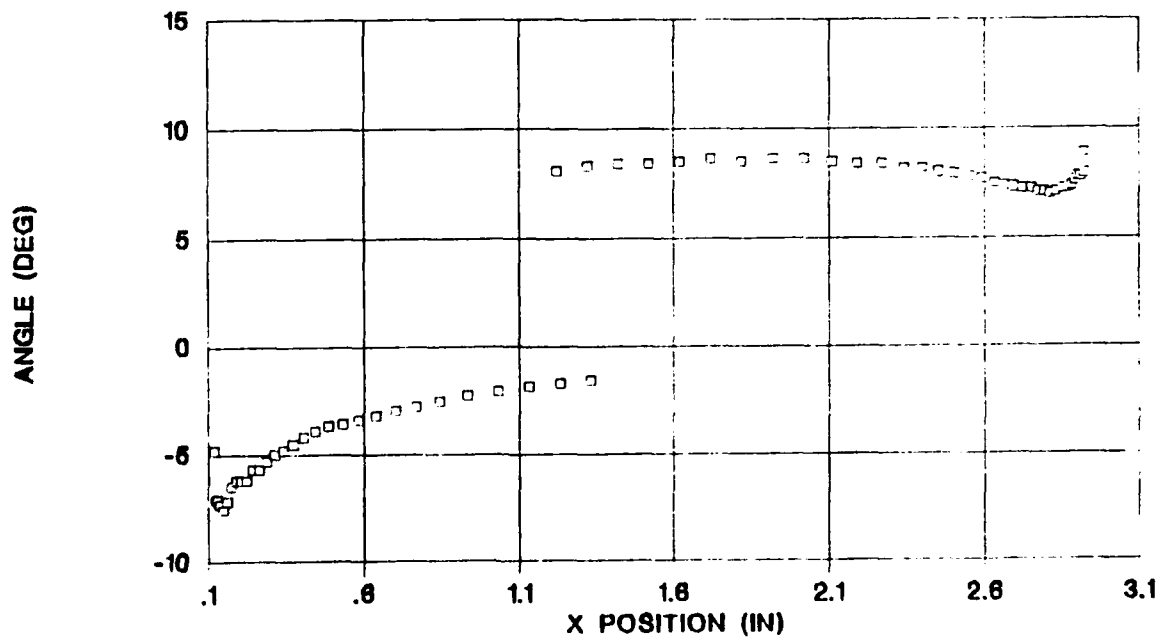
## STATION 15

### TOTAL VELOCITY



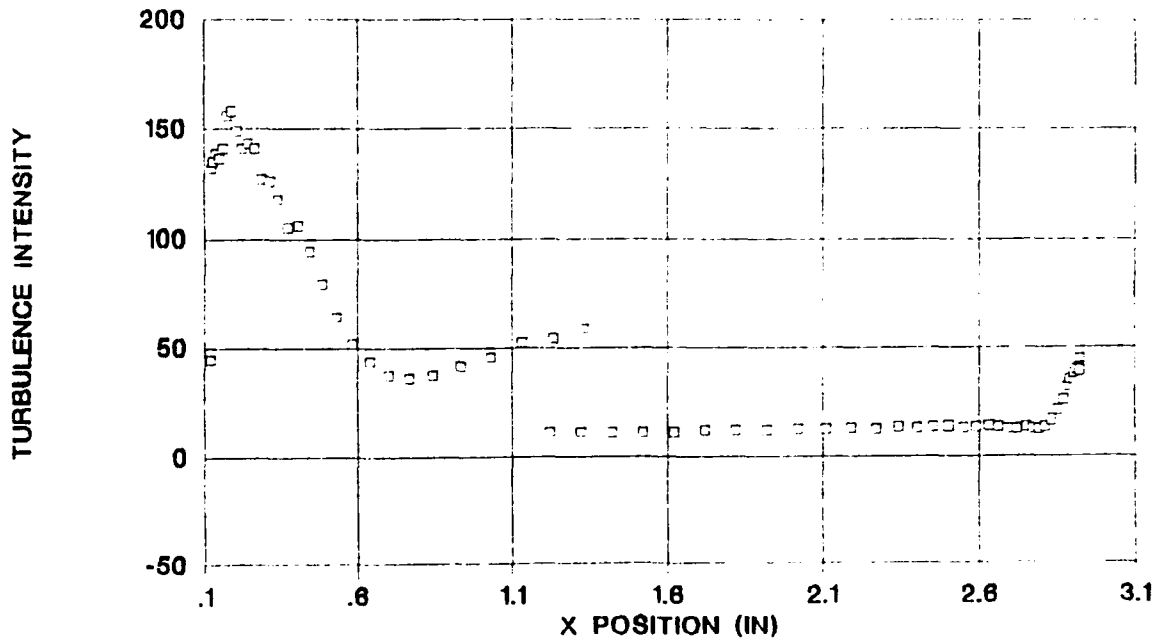
## STATION 15

### FLOW ANGLE



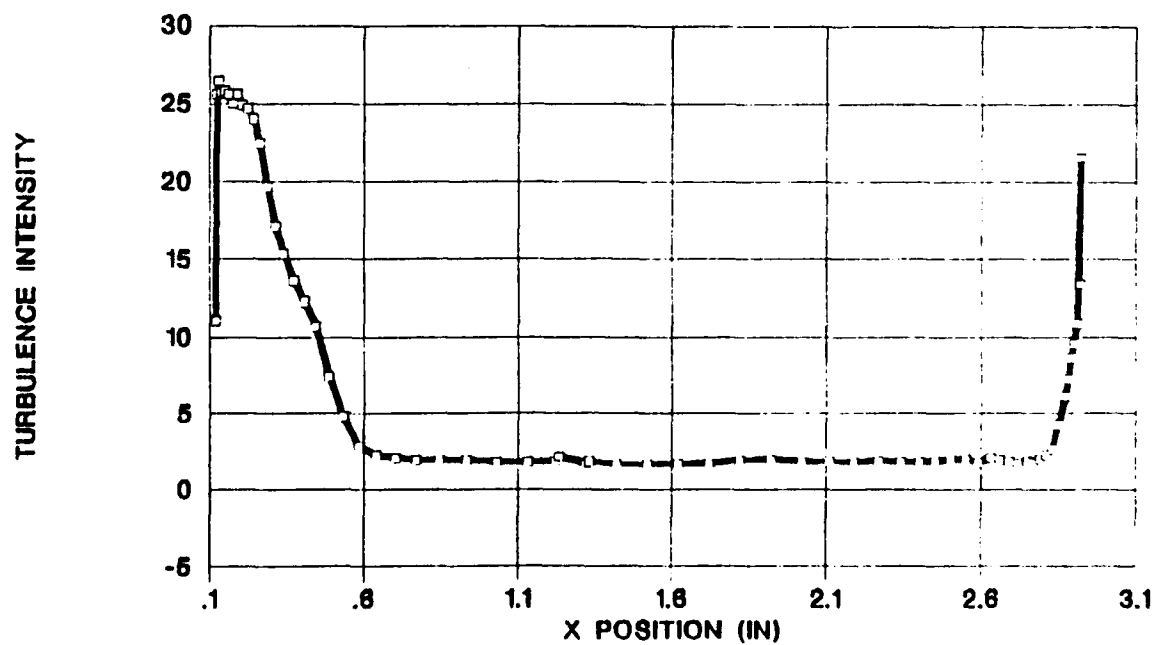
# STATION 15

## U TURBULENCE INTENSITY



# STATION 15

## V TURBULENCE INTENSITY



Velocity in M/SEC  
Data is from the following files:  
ST181207.1,41  
STATION 18

X(in)	Z(in)	U-Mean Velocity	U-Standard Deviation	V-Mean Velocity	V-Standard Deviation	UV-Angle Mean
-1	.678	3.64	1.17	67.5	1.31	86.9
-.875	.678	3.66	1.06	67.5	1.4	86.9
-.75	.678	3.78	1.09	67.6	1.37	86.8
-.625	.678	3.82	1.11	67.6	1.56	86.8
-.5	.678	4.01	1.16	67.7	1.43	86.6
-.375	.678	4.14	1.17	67.8	1.26	86.5
-.25	.678	4.28	1.25	67.9	1.36	86.4
-.125	.678	4.68	2.09	67.8	2.56	86.1
.0001	.678	4.64	7.69	38.1	10	83.1
.125	.678	-.398	5.45	24.6	6.5	90.9
.25	.678	-.0235	5.07	35.6	8.03	90
.375	.678	.95	4.37	50.4	7.23	88.9
.5	.678	1.68	3.3	63.4	5.3	88.5
.625	.678	1.87	1.76	68.1	2	88.4
.75	.678	2.09	1.16	68.3	1.77	88.3
.875	.678	2.38	1.09	68.2	1.38	88
1	.678	2.81	1.09	68	1.6	87.6
1.12	.678	3.03	1.08	67.9	1.5	87.4
1.25	.678	3.2	1.03	67.7	1.24	87.3
1.38	.678	3.38	1.02	67.6	1.39	87.1
1.5	.678	3.49	1.01	67.6	1.23	87
1.62	.678	3.68	.992	67.6	1.64	86.9
1.75	.678	3.88	.995	67.6	1.61	86.7
1.87	.678	4.1	1.03	67.6	1.38	86.5
2	.678	4.15	1.06	67.6	1.21	86.5
2.12	.678	3.84	1.17	67.6	1.58	86.7
2.25	.678	3.91	1.22	67.4	1.34	86.7
2.38	.678	4.16	1.19	67.5	1.45	86.5
2.5	.678	4.2	1.13	67.3	1.71	86.4
2.62	.678	4.43	1.18	67.6	1.77	86.3
2.75	.678	4.59	1.16	67.5	1.8	86.1
2.87	.678	5.34	2.57	66.8	3.17	85.4
3	.678	4.67	8.03	31.5	9.13	81.6
3.12	.678	-.476	5.47	24.1	6.5	91.1
3.25	.678	.436	5.05	38.8	9	89.4
3.38	.678	1.45	3.65	57.3	6.65	88.6
3.5	.678	1.64	1.96	67.4	2.28	88.6
3.62	.678	1.8	1	68.3	1.71	88.5
3.75	.678	2.23	.976	68.6	1.36	88.1
3.87	.678	2.54	1.03	68.6	1.52	87.9
4	.678	2.62	1.1	68.5	1.47	87.8



Velocity in M/SEC

Data is from the following files:

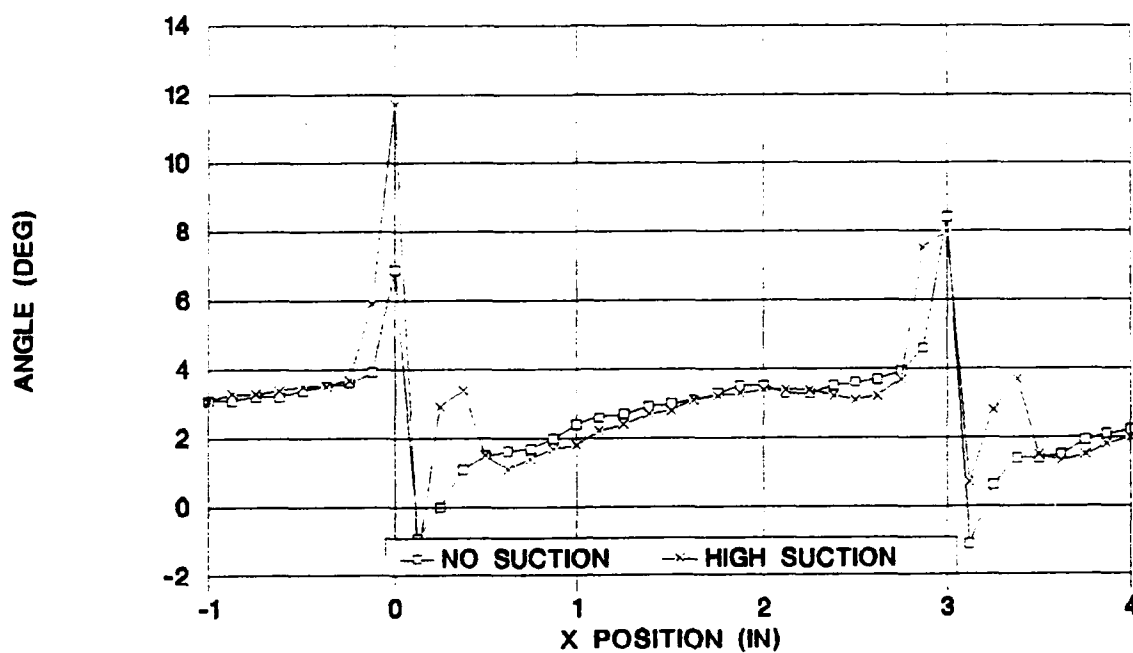
ST18SUCK,1,41

STATION 18 HIGH SUCTION

X(mm)	Z(mm)	U-Mean Velocity	U-Standard Deviation	V-Mean Velocity	V-Standard Deviation	UV-Angle Mean
-25.4	17.2	3.6	1.16	66.3	2.26	86.9
-22.2	17.2	3.87	1.19	67.7	2.62	86.7
-19	17.2	3.95	1.26	68	2.79	86.7
-15.9	17.2	4.08	1.34	68.1	2.82	86.6
-12.7	17.2	4.16	1.49	68	2.92	86.5
-9.52	17.2	4.15	1.47	68	3.17	86.5
-6.35	17.2	4.4	1.47	67.9	3.11	86.3
-3.17	17.2	6.53	3.06	62.7	7.15	84.1
0	17.2	4.69	9.07	22.7	7.24	78.3
3.17	17.2	1.586	4.81	21.4	6.35	91.6
6.35	17.2	1.95	4.38	39	9.06	87.1
9.52	17.2	3.39	3.64	56.5	7.16	86.6
12.7	17.2	1.78	1.92	67	3.35	88.5
15.9	17.2	1.3	1.26	67.8	2.57	88.9
19	17.2	1.61	.963	67.7	3.62	88.6
22.2	17.2	1.96	.941	68	2.81	88.3
25.4	17.2	2.19	.954	68	3.29	88.2
28.6	17.2	2.65	1.24	68.3	2.67	87.8
31.7	17.2	2.91	1.03	68.1	3.49	87.6
34.9	17.2	3.27	1.11	68.1	3.3	87.3
38.1	17.2	3.39	1.21	68.3	3.04	87.2
41.3	17.2	3.63	1.29	68.1	3.04	86.9
44.5	17.2	3.77	1.3	67.9	3.78	86.8
47.6	17.2	3.95	1.38	67.9	3.52	86.7
50.8	17.2	4.02	1.36	67.8	3.3	86.6
54	17.2	3.96	1.33	67.5	3.47	86.6
57.2	17.2	4.03	1.33	67.5	3.35	86.6
60.3	17.2	3.8	1.22	67	3.77	86.8
63.5	17.2	3.69	1.21	67.1	3.32	86.9
66.7	17.2	3.8	1.18	67.1	2.93	86.8
69.8	17.2	4.36	1.36	67.2	3.13	86.3
73	17.2	7.56	4.11	57.5	8.55	82.5
76.2	17.2	2.96	8.44	21	6.3	82
79.4	17.2	.362	4.56	28.4	7.98	89.3
82.6	17.2	2.3	3.81	46.3	7.31	87.2
85.7	17.2	3.91	3.33	59.9	6.65	86.3
88.9	17.2	1.76	1.89	67.9	3.36	88.5
92.1	17.2	1.58	1.1	68.2	3.7	88.7
95.2	17.2	1.8	1.16	68.1	3.62	88.5
98.4	17.2	2.19	1.19	68.2	3.58	88.2
102	17.2	2.43	1.14	68.2	2.8	88

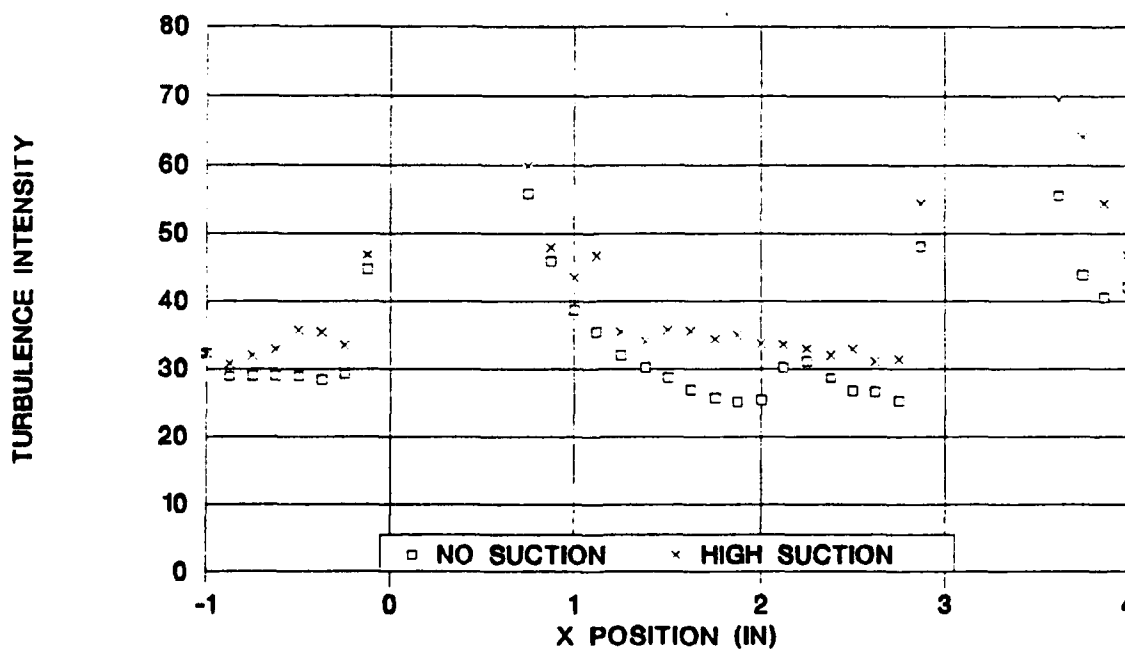
## STATION 18

FLOW ANGLE



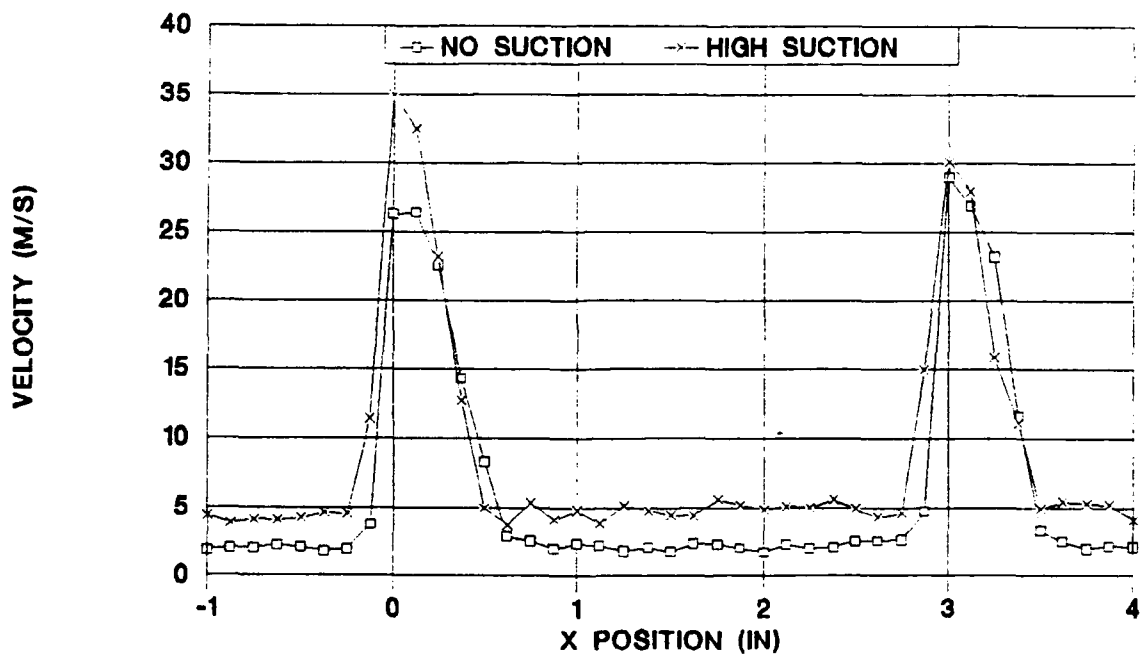
## STATION 18

U TURBULENCE INTENSITY



# STATION 18

## V TURBULENCE INTENSITY



## APPENDIX B. DATA REDUCTION

This Appendix contains the following:

1. All equations used for the calculations in this document.
2. The output of  $V_{ref}$  calculations for the LDV data and program.

### B1. LIST OF EQUATIONS

The following equations were used to reduce all data taken by the rake probe and the instrumented blade.

For all rake surveys:

$$C_p = \frac{P_{Ti}}{P_{Tmax}} \quad (1)$$

$$Q = P_{T12} - P_{S9} \quad (2)$$

$$\alpha = 0.0805 + 31.575 \frac{P_{10} - P_{11}}{Q} \quad (3)$$

$$P_{S \text{ corrected}} = P_S - (0.412 + 0.1817Q) \quad (4)$$

where numbered subscripts indicated probe number of the rake. The  $\alpha$  and corrected  $P_S$  derivations are described in Appendix C.

The computation of the displacement thickness comes from the definition:

$$\delta^* = \int \left( 1 - \frac{u}{U} \right) dy \quad (5)$$

For the rake probe, the integration interval was taken from the endwall to pressure port 8 for the north and 12 for the south, a total of three inches span for each. This resulted in the following equation for the north endwall:

$$\delta^* = \int \left[ 1 - \frac{\left[ \frac{C_{P \ 1}}{C_{P \ 8}} * P_{T \ max} \right] - P_s}{Q} \right] dy \quad (6)$$

The integration was performed using the trapezoidal method.

To compute the axial velocity ratio, the definition from Reference 1 was used:

$$AVR = \int_0^s \frac{c_{z \ 2}}{c_{z \ 1}} dx \quad (7)$$

where  $c_z$  is the axial velocity at the upstream, (1), and downstream, (2), locations and  $s$  is one blade space, (3 in.). Because of this definition and since the analysis was done for the incompressible case, the local velocity and flow angle measured by the rake probe was used and corrected for by the

Prandtl total pressure to compensate for the run-to-run variations. This yielded the following equation:

$$AVR = \int \frac{\sqrt{\left( (P_{12} - P_s) * \frac{2}{\rho} \right)_2}}{\sqrt{\left( (P_{12} - P_s) * \frac{2}{\rho} \right)_1}} * \sqrt{\frac{P_{T1}}{P_{T2}}} * \frac{\cos(\beta_2)}{\cos(\beta_1)} dx \quad (8)$$

where  $P_T$  is the corresponding Prandtl total pressure and  $P_s$  is from Equation (4). This integration was performed by a summation since the interval was held constant.  $\beta$  was measured with the following equation:

$$\beta = \beta_{rake} + \alpha \quad (9)$$

where  $\beta_{rake}$  is the angle the rake probe was mounted in the tunnel, which was  $46^\circ$  for upstream and  $90^\circ$  for downstream, and  $\alpha$  is from Equation (3).

The instrumented blade pressure coefficients were computed as follows:

$$C_p = \frac{P_i - P_\infty}{(P_T - P_s)_{Prandtl}} \quad (10)$$

where  $P_i$  was the instrumented blade pressure tap number.

## B2. $V_{\text{ref}}$ COMPUTATION AND PROGRAM

The method for the calculation of  $V_{\text{ref}}$  came from Elazar [Ref. 9]. The program was written by Garth V. Hobson.

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LEAST SQUARES STRAIGHT LINE CURVE FIT IS USED TO DETERMINE TUNNEL CHARACTERISTICS AT DIFFERENT SPEEDS

NEWTON'S METHOD IS USED TO DETERMINE THE PRESSURE VELOCITY FROM THE RECORDED AMBIENT PRESSURE AND THROAT PRESSURE AND TEMPERATURE

BEGIN DETERMINING TUNNEL CHARACTERISTICS  
FROM THE FOLLOWING MEASURED VALUES

AXIAL VEL. IN PER SEC.	TANGENTIAL VEL. IN PER SEC.	AMBIENT PRESS. INCHES MERCURY	PLENUM PRESS. INCHES WATER	PLENUM TEMP. DEG. C.
33.7320	33.5860	29.8700	4.3500	16.1111
40.8170	39.0550	29.8700	4.1000	16.6667
48.7490	46.5000	29.8700	3.9000	17.2222
56.0220	53.7310	29.8700	3.7000	17.7778
60.6650	59.0180	29.8700	3.5000	18.3333
66.0340	65.7400	29.8700	3.3000	18.8889

### CALCULATED VALUES FOR THE TUNNEL CONFIGURATION

[illegible]

CALLING LEAST SQUARES SUBROUTINE  
TO DETERMINE THE PRESSURE RATIO AS A FUNCTION OF EACH NO. PARAM  
PRESSURE RATIO = 01 \* OMUX + 00

### MAIRIX EQUATION

0.60E+01    0.14E+00 00    0.10E+00  
0.14E+00    0.36E-02 01    0.47E-02

01 =    0.12775874005E+01    00 =    0.25001140071E+02

REFERENCE CONDITIONS FOR EACH RUN	PLENUM PRESSURE	PLENUM TEMPERATURE	RUN NAME
AMBIENT PRESSURE INCHES MERCURY	INCHES WATER	DEGREES CELSIUS	
30.0500	12.1000	23.3300	SL101207.PRN
29.7600	11.9000	17.7778	SL101207.PRN

```

1
PRESSURE RATIO = 0.03004 MACH NUMBER PARAFINER = 0.3800E 01
RUN NAME = S1101209.PRN

```

BEGIN NEWTON ITERATION

```

ITERATION NUMBER 1  MACH NO.  PARAM.  -  0.106809  ERROR TERM  -  0.78737E-03
ITERATION NUMBER 2  MACH NO.  PARAM.  -  0.107077  ERROR TERM  -  0.49101E-03
ITERATION NUMBER 3  MACH NO.  PARAM.  -  0.107076  ERROR TERM  -  0.22309E-03
VREF = 02.63530826344

```

```

1 = 2
PRESSURE RATIO = 0.02984 MACH NUMBER PARAMETER = 0.3872E 01
RUN NAME = ST181207.PRN

```

BEGIN NEWION ITERATION

```

ITERATION NUMBER 1  MACH NO:  PARAN:  1  0.107896  ERROR TERM  0.3972E-03
ITERATION NUMBER 3  MACH NO:  PARAN:  1  0.106697  ERROR TERM  0.6517E-06
VREF = 01.57340572006

```

EXPERIMENT NUMBER	REFERENCE VELOCITY	NAME
1	87.6555	511A1207.PRN
	81.5934	511A1207.TRN

```

PROGRAM TO COMPUTE THE CALIBRATION CURVE FOR THE LHS LOW SPEED
CORRODE WIND TUNNEL.

A STRAIGHT LINE IS FITTED THROUGH THE REFERENCE CONDITIONS OF
THE TUNNEL AT DIFFERENT SPEEDS.
THE REFERENCE VELOCITY IS THEN OBTAINED, BY NEWTON'S METHOD,
DEPENDING ON THE TUNNEL PLENUM PRESSURE AND TEMPERATURE.

PROGRAM CALIBRATE
IMPLICIT REAL*8(A-H,O-Z)
PARAMETER (NP=6)
DIMENSION VA(NP), VT(NP), PA(NP), PP(NP), TP(NP)
DIMENSION VLOT(NP), X(NP), ANUX(NP), PR(NP)
DIMENSION PARR(100), APR(100), TPR(100), PTR(100), ANUXR(100)
DIMENSION VREF(100)
CHARACTER*14 NAME(100)

OPEN(UNIT=10, FILE='CALIB.DAT', STATUS='UNKNOWN')
OPEN(UNIT=11, FILE='REFER.DAT', STATUS='UNKNOWN')
OPEN(UNIT=12, FILE='CALIB.OUT', STATUS='UNKNOWN')

PRINT BANNER

      VA = AXIAL VELOCITY          MEASURED BY LDV (M/S)
      VT = TANGENTIAL VELOCITY    MEASURED BY LDV (M/S)
      PA = AMBIENT PRESSURE        (INCHES MERCURY)
      PP = PLENUM PRESSURE         (INCHES WATER)
      TP = PLENUM STAGNATION TEMPERATURE (DEG. C.)

DO 1 I=1, NP
  READ(10,100) VA(I), VT(I), PA(I), PP(I), TP(I)
  WRITE(11,110) VA(I), VT(I), PA(I), PP(I), TP(I)
  WRITE(12,110) VA(I), VT(I), PA(I), PP(I), TP(I)
100 CONTINUE
  FORMAT(1X,5F8.4)
110 FORMAT(3X,F8.4,7X,F8.4,8X,F8.4,7X,F8.4,7X,F8.4)
  CALC VLOT, X, ANUX AND PR
  WRITE(12,510)
510 FORMAT(5X, 'CALCULATED VALUES FOR THE TUNNEL CONFIGURATION', //
  #3X, 'TOTAL VELOCITY', 6X, 'MACH NUMBER', 7X, 'MACH NUMBER FUNCT.', 3X,
  #1X, 'PRESSURE RATIO //')
DO 2 I=1, NP
  VLOT(I) = DSQRT(VA(I)**2 + VT(I)**2)
  X(I) = VLOT(I) / DSQRT(2.00 * CPA * (TP(I) + 273.1500))
  ANUX(I) = (GAMMA/GM1) * (X(I)**2) * (1.00 - (X(I)**2)) * (1.00 / (GM1))
  PR(I) = 1.00 - (RHOHG * PA(I)) / (RHOH * PP(I) + RHOHG * PA(I))
  WRITE(12,101) VLOT(I), X(I), ANUX(I), PR(I)
  WRITE(12,101) VLOT(I), X(I), ANUX(I), PR(I)
2 CONTINUE
101 FORMAT(4D20.11)

  CALL THE LEAST SQUARES SUBROUTINE TO FIT A STRAIGHT LINE THROUGH THE
  DATA, X-AXIS = PR, Y-AXIS = ANUX (MACH NO. PARAMETER)
  WRITE(12,520)
  WRITE(12,520)
520 FORMAT(710X, 'CALLING LEAST SQUARES SUBROUTINE', //1X,
  #1X, 'TO DETERMINE THE PRESSURE RATIO AS A FUNCTION OF MACH NO. PARAM'
  #71X, 'PRESSURE RATIO = A1 * ANUX + A0', //)
  CALL LEASTSQUARE(NP, PR, ANUX, A0, A1)
  WRITE(12,530) A1, A0
  WRITE(12,530) A1, A0
530 FORMAT(710X, 'A1 = ', D20.11, ' A0 = ', D20.11 /)
  READ IN REFERENCE CONDITIONS
  WRITE(12,179)
  WRITE(12,179)
179 FORMAT(71X, 'REFERENCE CONDITIONS FOR EACH RUN', //
  #1X, 'AMBIENT PRESSURE', 'PLENUM PRESSURE', 'PLENUM TEMPERATURE',
  #1X, 'INCHES MERCURY', 'INCHES WATER', 'DEGREES CELSIUS', //)
  THE NUMBER OF EXPERIMENTS
  READ(11,178) NE
178 FORMAT(1X,13)
  THE REFERENCE CONDITIONS

```



```

777 CONTINUE
590 FORMAT(8X,12,16X,F8.4,10X,014)
STOP
END

```

```

-----
A LEAST SQUARES CURVE FIT FOR A STRAIGHT LINE THROUGH NOISY
DATA

```

```

ALGORITHM TAKEN FROM: NUMERICAL METHODS, ROBERT W. HORNBECK
PAGES 122 - 130, QUANTUM, 1975
-----

```

```

SUBROUTINE LEASTSQUARE(N,X,FX,A0,A1)
IMPLICIT REAL*8(A-H,O-Z)
DIMENSION COEFF(2,2),RHS(2),X(N),FX(N)

C
C SET UP COEFFICIENT MATRIX AND RIGHT HAND SIDE.
C
COEFF(1,1)=N
COEFF(1,2)=0.D0
COEFF(2,2)=0.D0
C
RHS(1)=0.D0
RHS(2)=0.D0
C
DO 2 I=1,N
  COEFF(1,2)=COEFF(1,2)+X(I)
  COEFF(2,2)=COEFF(2,2)+X(I)*X(I)
  RHS(1)=RHS(1)+FX(I)
  RHS(2)=RHS(2)+X(I)*FX(I)
2 CONTINUE
COEFF(2,1)=COEFF(1,2)

C
C PRINT MATRIX EQUATION
C
WRITE(*,110)
WRITE(12,110)
110 FORMAT(75X,'MATRIX EQUATION',/)
DO 3 I=1,2
  WRITE(12,102)(COEFF(I,J),J=1,2),1-1,RHS(1)
  WRITE(12,102)(COEFF(I,J),J=1,2),1-1,RHS(1)
3 CONTINUE
102 FORMAT(5X,2(2X,D10.2),', ',11,3X,D10.2)

C
C GAUSS ELIMINATION
C
TERM=COEFF(1,1)/COEFF(2,1)
COEFF(2,2)=COEFF(2,2)-COEFF(1,2)*TERM
RHS(2)=RHS(2)-RHS(1)*TERM
C
A1=RHS(2)/COEFF(2,2)
A0=(RHS(1)-COEFF(1,2)*A1)/COEFF(1,1)
C
RETURN
END

```

## APPENDIX C. PROBE CALIBRATION

This Appendix contains the following:

1. Yaw probe calibration and plot.
2. Static probe correction.

### C1. YAW PROBE CALIBRATION

The calibration of the yaw probe, pressure ports 10 and 11, located at the midspan of the rake probe was accomplished by mounting it in a free jet and subjecting it to three different dynamic pressures. The range of dynamic pressure was 12.2 in. H<sub>2</sub>O, 14.1 in. H<sub>2</sub>O and 16 in. H<sub>2</sub>O. The probe was yawed from -5° to +5° in one degree increments. A line was fitted to the data for each test and finally an average equation was fitted. It was noticed that the variation of ( $P_{10}-P_{11}$ ) was not the same for positive and negative incidence due to manufacture imperfection. The uncertainty in the probe calibration was calculated to be 0.5° . The average equation is found in Appendix B.

The data are presented on the following pages and followed by the plot of the data with each equation.

### C2. STATIC PROBE CORRECTION

The static probe displayed a deviation of 1.98 in.

H<sub>2</sub>O for the first calibration run and 2.34 in. H<sub>2</sub>O for the other two runs. Therefore, a line was fitted to correct the rake-measured static pressure for the tunnel velocity variations. The Reynolds number referenced to the probe diameter was between 6000 and 7000, therefore, Reynolds number effect was considered to be negligible. The correction equation is found in Appendix B.

# RAKE PROBE CALIBRATION 10/16/92

REFERENCE PA=49.9 IN H2O		PA= 29.99 IN Hg Q=12.2 IN H2O		Tt=112 F		
ALPHA	PT	PROBE (P9)	PROBE (P10)	PROBE (P11)	PROBE (P12)	ATMOS
-5	37.7	47.7	38.6	40.3	37.7	49.9
-4	37.7	47.75	38.7	40.15	37.7	49.9
-3	37.7	47.85	38.95	39.9	37.7	49.9
-2	37.7	47.9	39.05	39.7	37.7	49.9
-1	37.7	47.9	39.25	39.45	37.7	49.9
0	37.7	47.9	39.3	39.3	37.7	49.9
1	37.7	47.9	39.35	39.15	37.7	49.9
2	37.7	48.1	39.5	39	37.7	49.9
3	37.7	48.2	39.8	38.9	37.7	49.9
4	37.7	48.2	39.9	38.8	37.7	49.9
5	37.7	48.2	40.1	38.65	37.7	49.9

Q=14.1 IN H2O

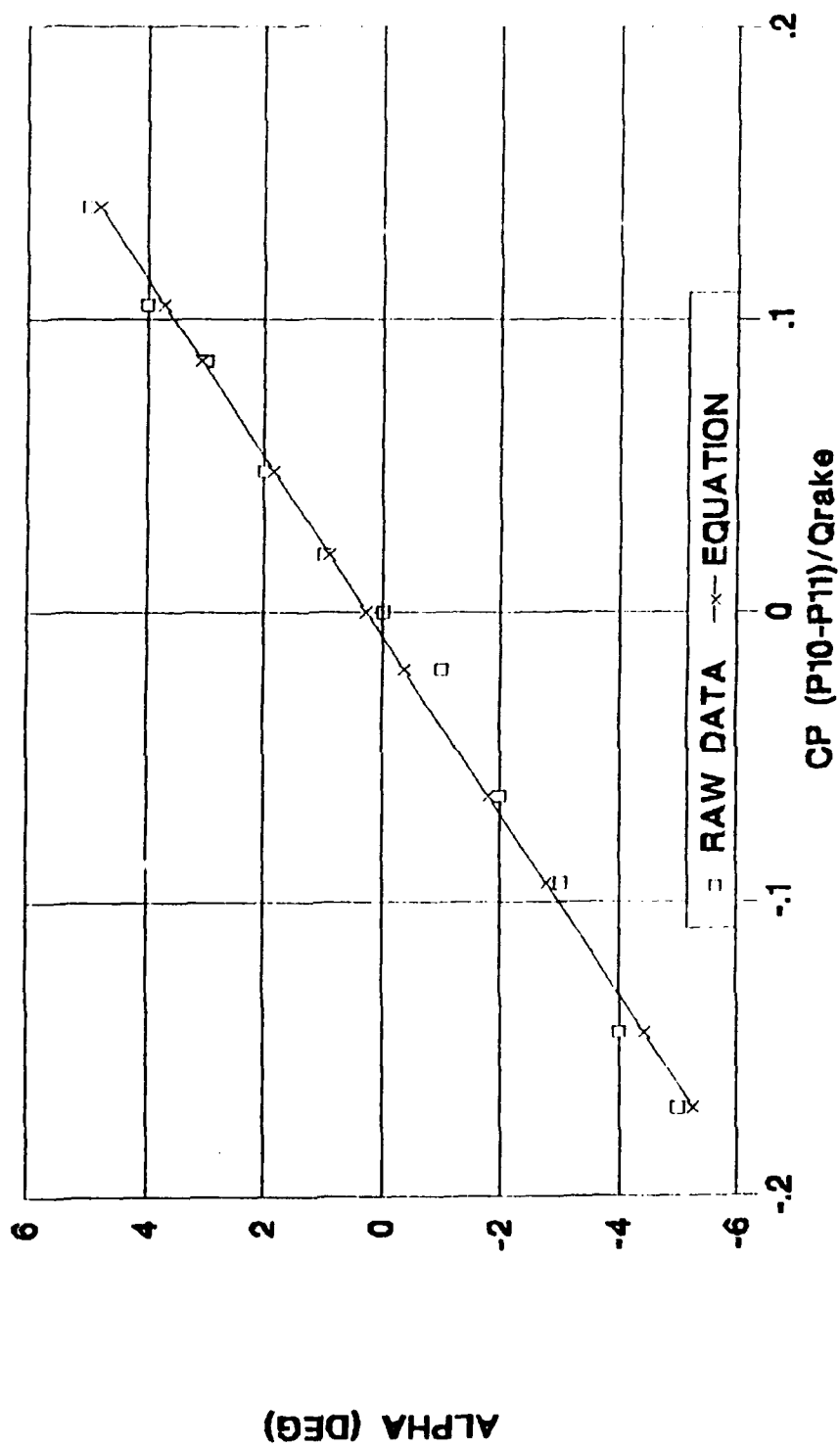
ALPHA	PT	PROBE (P9)	PROBE (P10)	PROBE (P11)	PROBE (P12)	ATMOS
-5	35.8	47.35	36.9	38.9	35.8	49.9
-4	35.8	47.4	37.05	38.65	35.8	49.9
-3	35.8	47.5	37.25	38.4	35.8	49.9
-2	35.8	47.55	37.45	38.15	35.8	49.9
-1	35.8	47.6	37.65	37.9	35.8	49.9
0	35.8	47.6	37.7	37.7	35.8	49.9
1	35.8	47.6	37.7	37.5	35.8	49.9
2	35.8	47.85	38.1	37.4	35.8	49.9
3	35.8	47.7	38.35	37.2	35.8	49.9
4	35.8	47.85	38.6	37.05	35.8	49.9
5	35.8	47.7	38.8	36.95	35.8	49.9

Q=16 IN H2O

ALPHA	PT	PROBE (P9)	PROBE (P10)	PROBE (P11)	PROBE (P12)	ATMOS
-5	33.9	47.5	35.2	37.35	33.9	49.9
-4	33.9	47.6	35.3	37	33.9	49.9
-3	33.9	47.7	35.55	36.75	33.9	49.9
-2	33.9	47.7	35.7	36.5	33.9	49.9
-1	33.9	47.7	35.9	36.15	33.9	49.9
0	33.9	47.7	36	36	33.9	49.9
1	33.9	47.7	36.2	35.7	33.9	49.9
2	33.9	47.6	36.45	35.55	33.9	49.9
3	33.9	47.6	36.8	35.4	33.9	49.9
4	33.9	47.55	37	35.2	33.9	49.9
5	33.9	47.5	37.3	35.2	33.9	49.9

# RAKE CALIBRATION

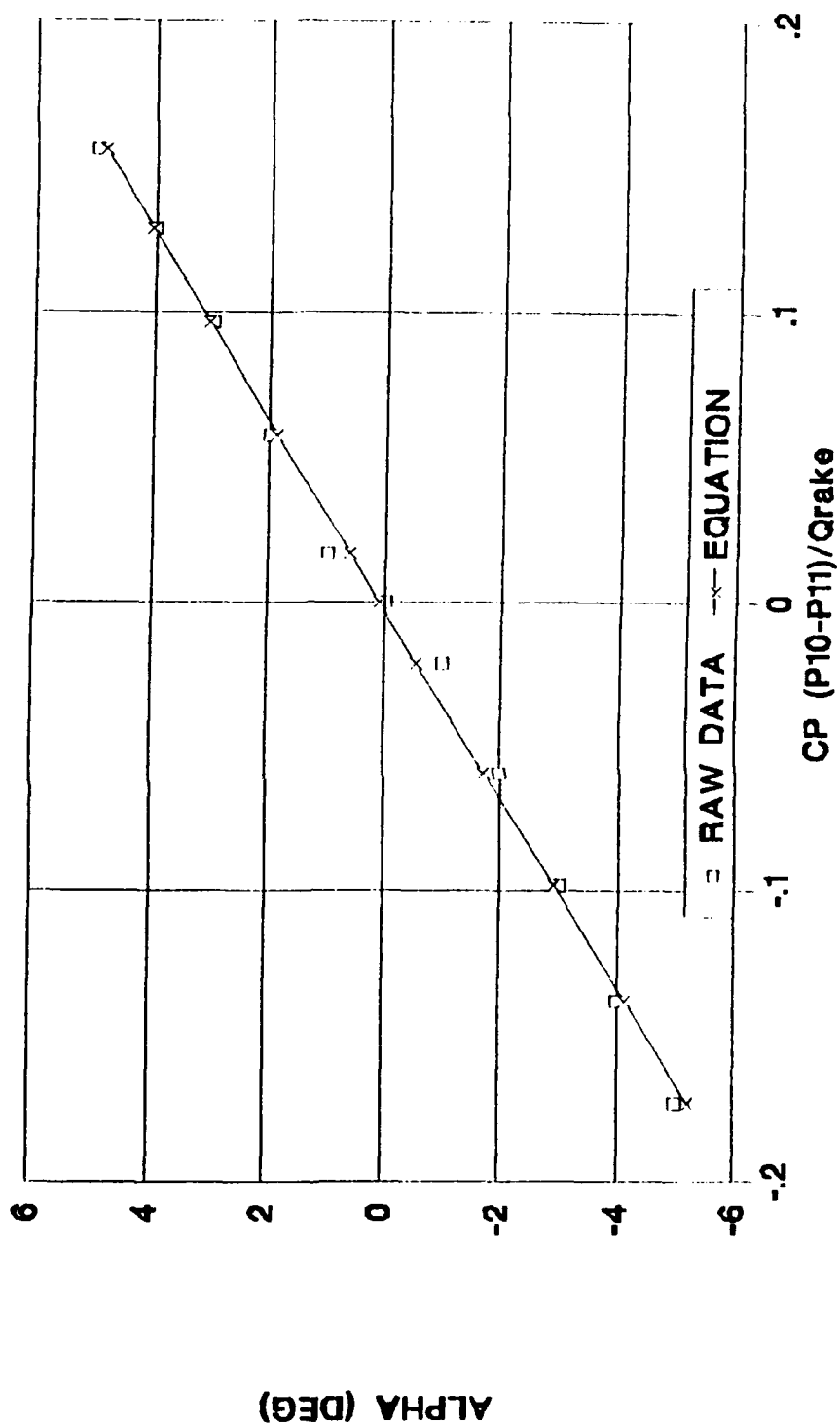
Q = 12.2 IN H2O





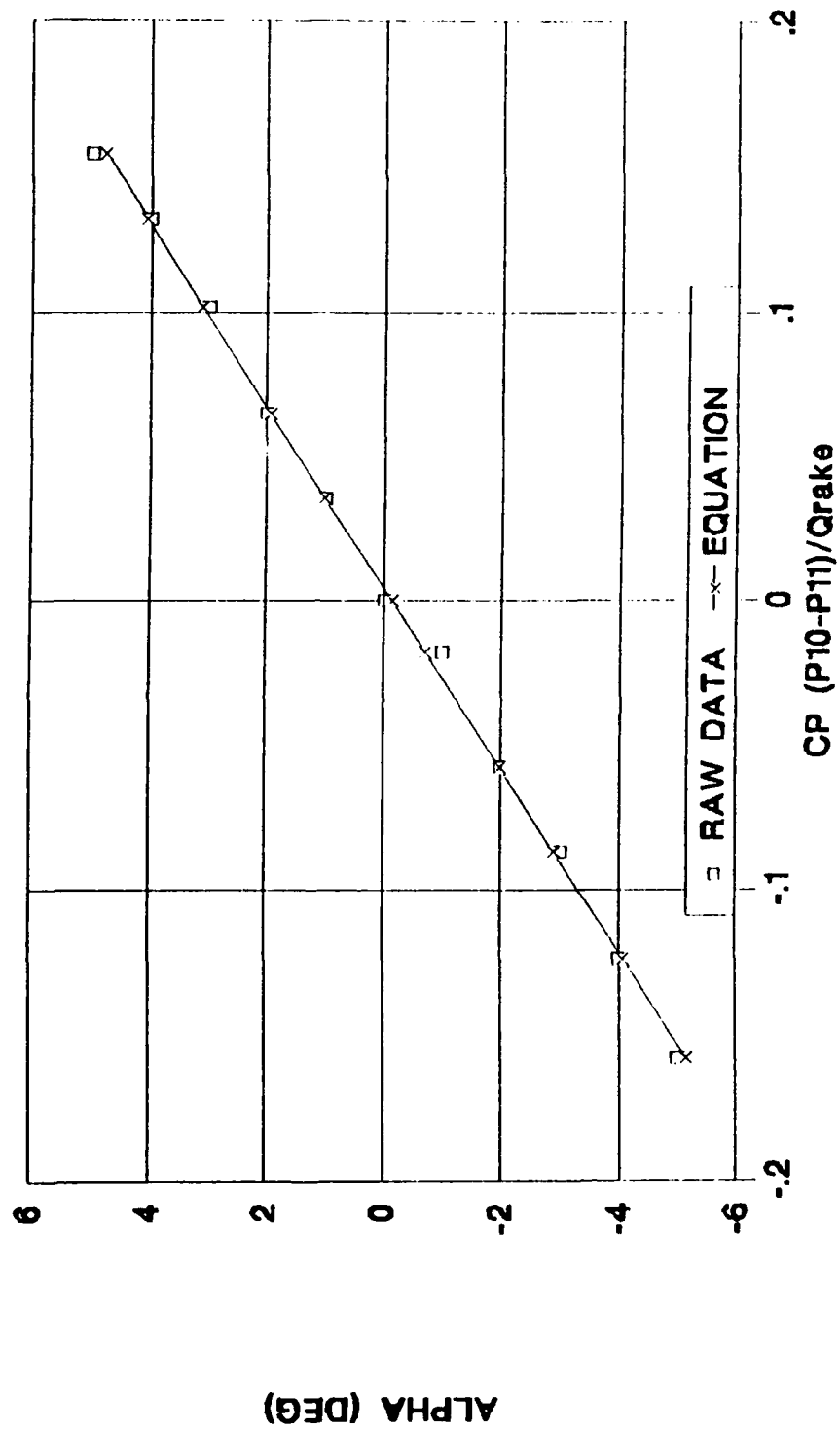
# RAKE CALIBRATION

Q = 14.1 IN H2O



# RAKE CALIBRATION

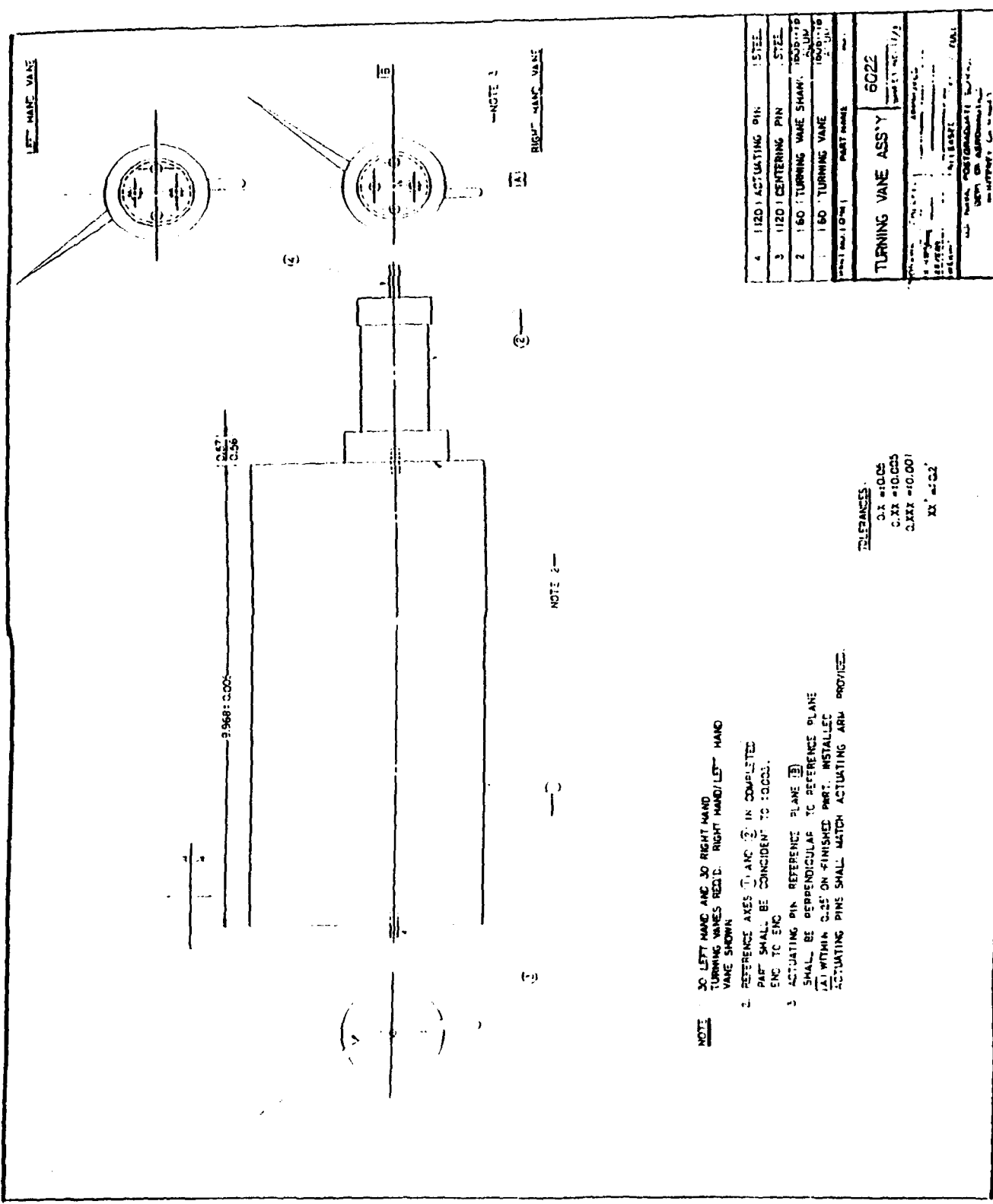
Q = 16.0 IN H2O



## APPENDIX D. ENGINEERING DRAWINGS

This Appendix contains all detailed drawings cited in the body of this document. They are displayed in the following order:

1. Inlet guide vanes
2. Suction slot
3. Suction slot adjustable mounting brackets
4. Porous baffle
5. Suction manifold cross section
6. Rake probe plan view
7. Rake probe spanwise dimensions



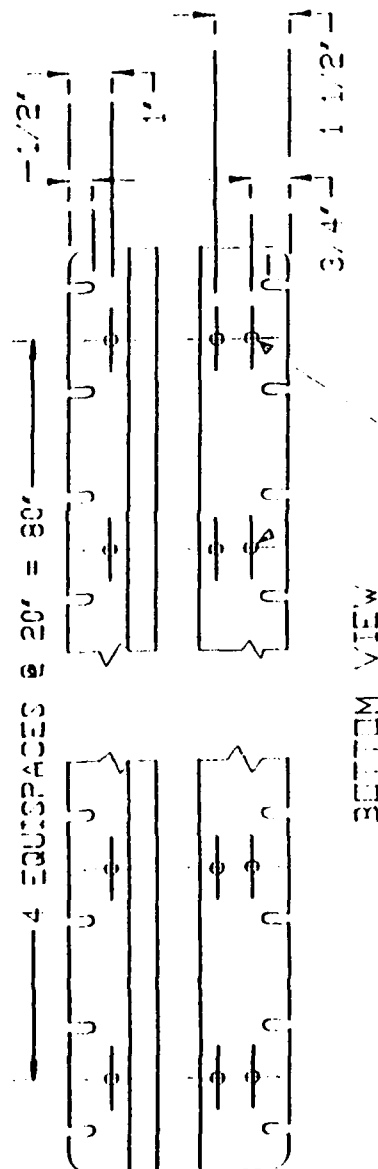
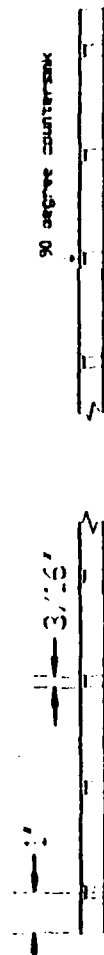
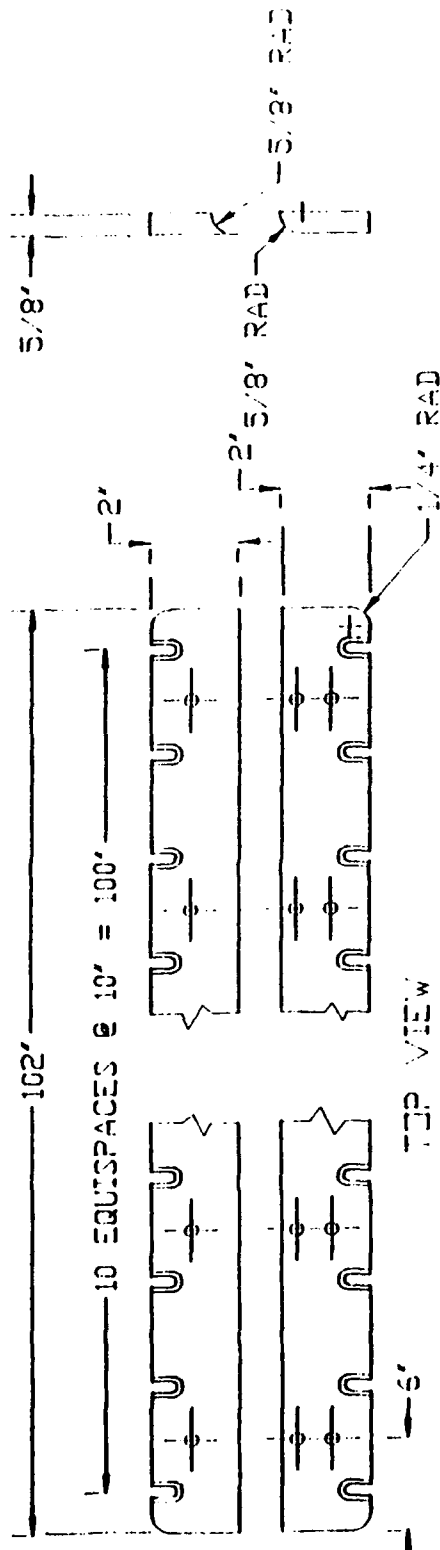
NOTE 1: 30 LEFT HAND AND 30 RIGHT HAND TURNING VANES REQ'D. RIGHT HAND/LEFT HAND VANE SHOWN

NOTE 2: REFERENCE AXES (1) AND (2) IN COMPLETED PART SHALL BE COINCIDENT TO 0.003. END TO END

NOTE 3: ACTUATING PIN REFERENCE PLANE (3) SHALL BE PERPENDICULAR TO REFERENCE PLANE (1) WITHIN 0.25' ON FINISHED PART. INSTALLED ACTUATING PINS SHALL MATCH ACTUATING ARN PROVIDED.

TOLERANCES:  
 .XX = 0.005  
 .XX = 0.003  
 .XXX = 0.001  
 XX = 0.02

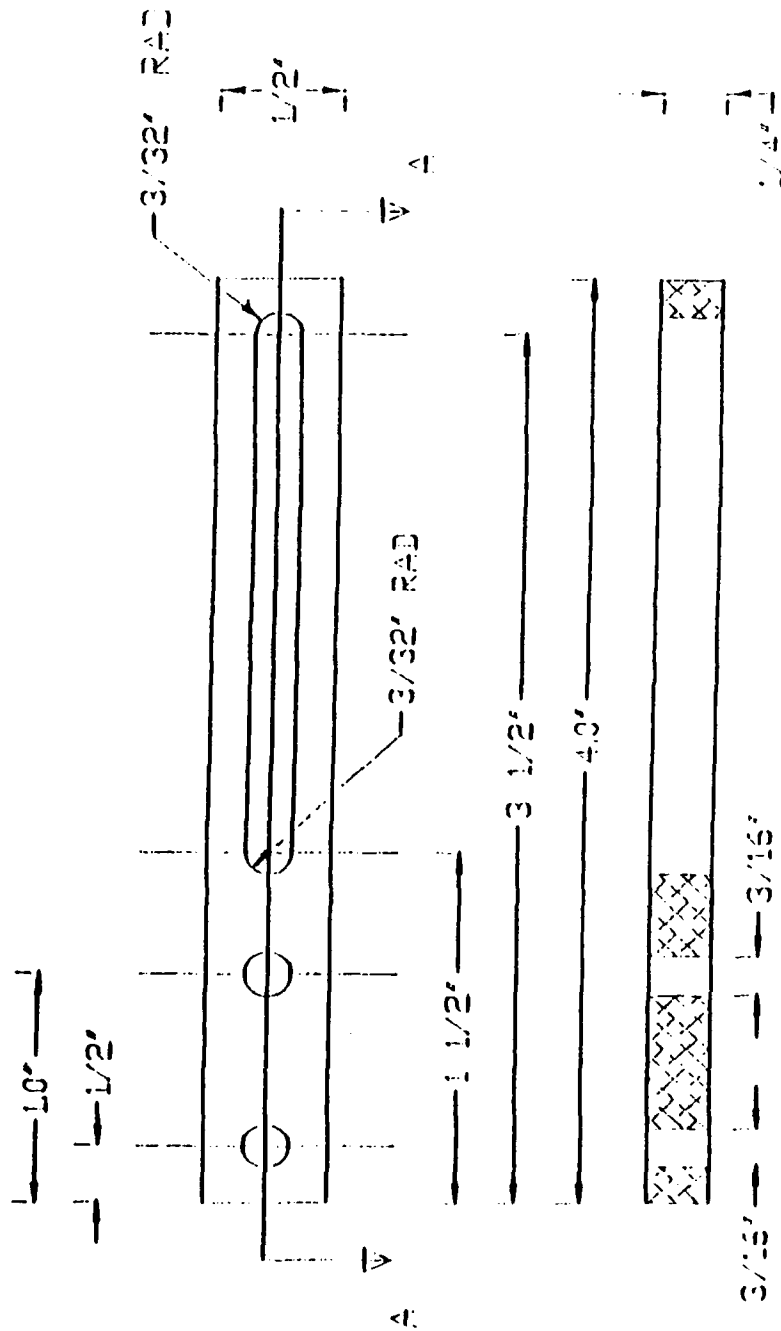
4	1120	ACTUATING PIN	STEEL
3	1120	CENTERING PIN	STEEL
2	160	TURNING VANE SHAFT	ROBUST ALUMINUM
1	160	TURNING VANE	ROBUST ALUMINUM
PART NAME		PART NAME	
TURNING VANE ASS'Y		6022	
DATE		11/1/73	
DESIGNED BY		J. L. BAKER	
CHECKED BY		J. L. BAKER	
APPROVED BY		J. L. BAKER	
ALL DATA POSTGRADUATE ENGINEERING		ALL DATA POSTGRADUATE ENGINEERING	
DEPT. OF AERONAUTICS		DEPT. OF AERONAUTICS	
UNIVERSITY OF CALIFORNIA		UNIVERSITY OF CALIFORNIA	



Material: Aluminum Alloy  
 6061-T6  
 Quantity: Two sets  
*gantt Hahn 09/20/01*

HOLE TAPPED FOR  
 3/16" CAP HEAD BOLTS

# CASCADE ENDWALL SLOTS



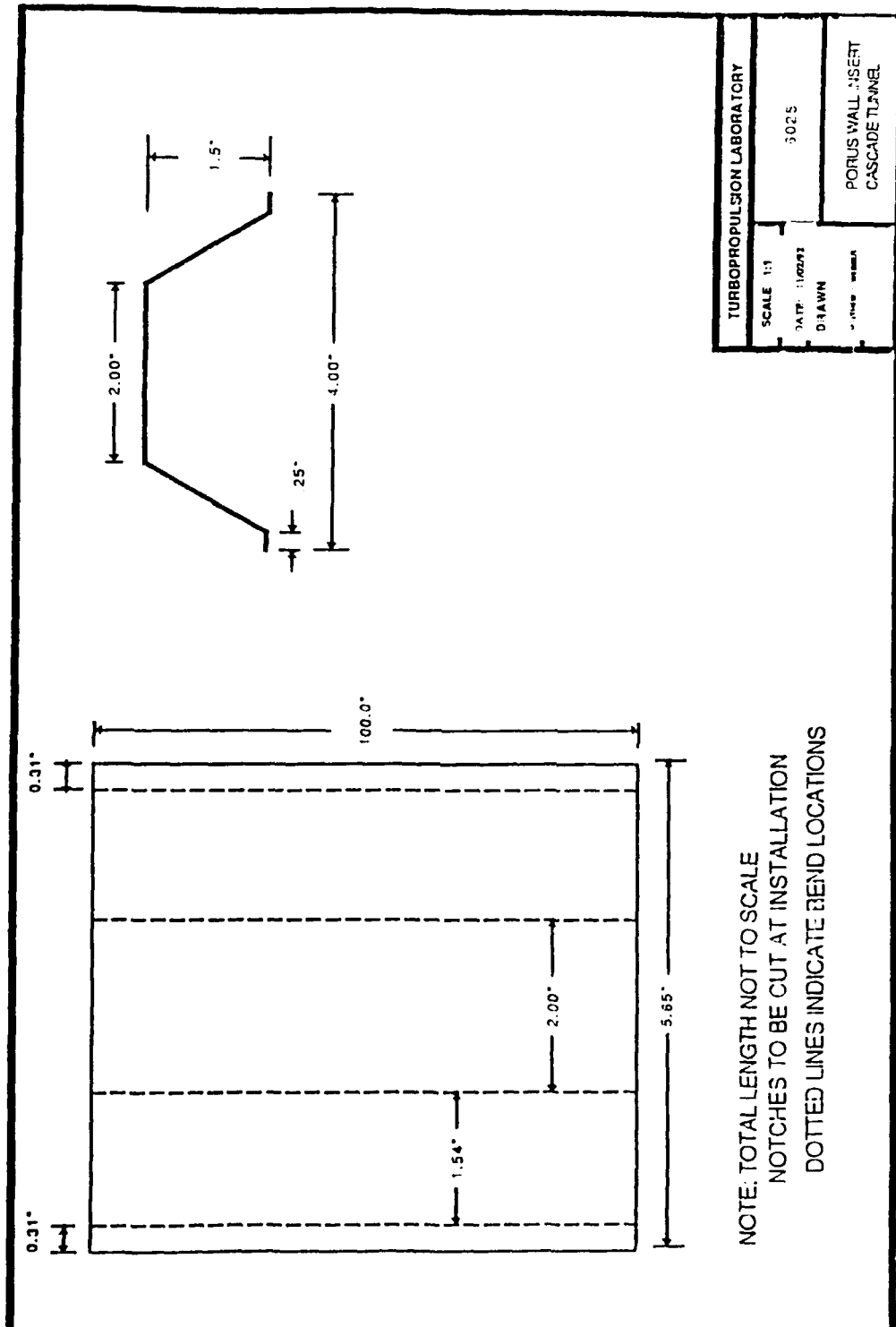
SECTION 2-2

GUIDES

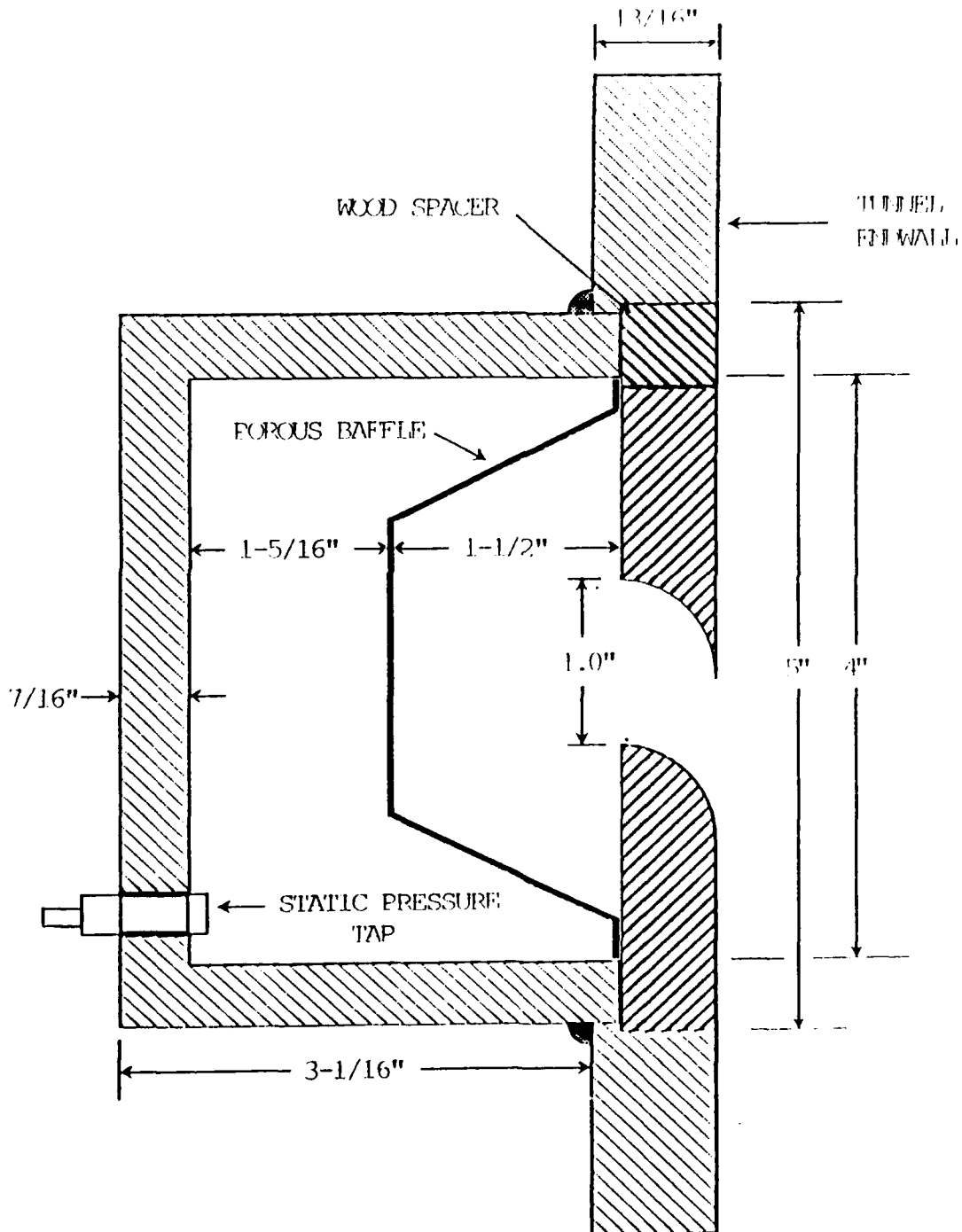
Material: Aluminum Alloy

Quantity: 10

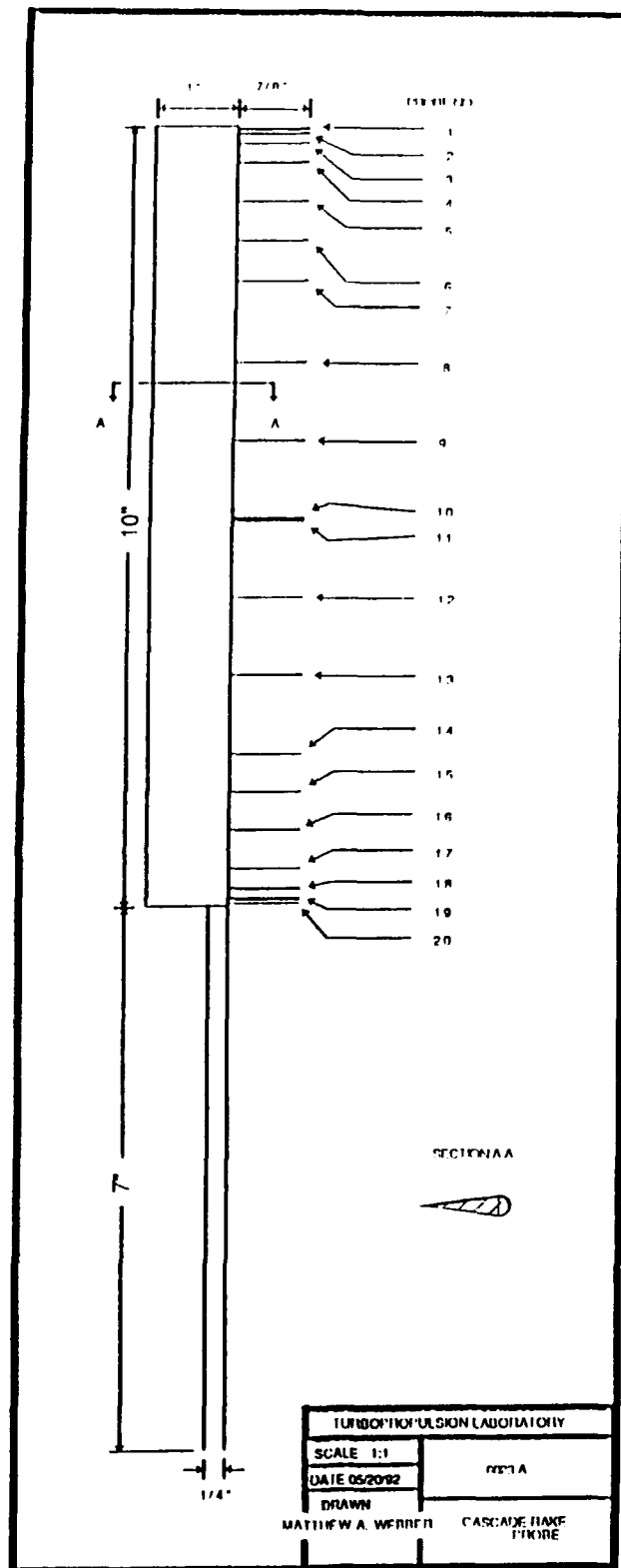
*gentle Hobbs 09/20/41*

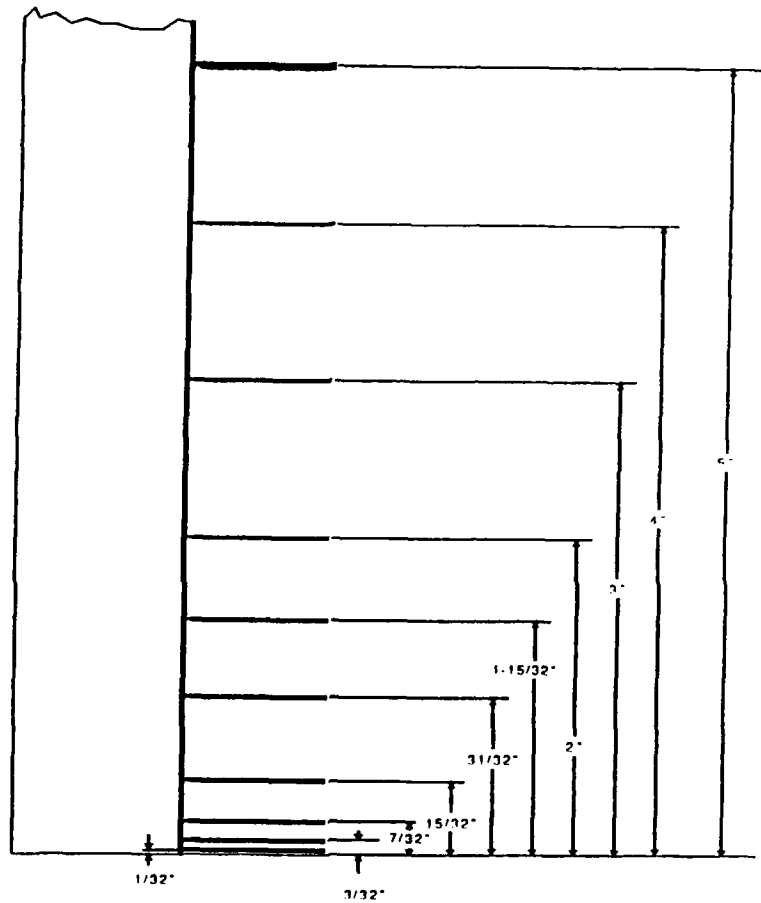


# CROSS-SECTION OF SUCTION MANIFOLD









TURBOPROPULSION LABORATORY	
SCALE 1:1	6023 D
DATE 05/20/92	
DRAWN	RAKE PROBE - PROBE DIMENSIONS
MATTHEW A. WEBBER	

## LIST OF REFERENCES

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